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Inertial amplification of continuous structures: Large band gaps from small masses

Niels M. M. Frandsen, ^{1,a)} Osama R. Bilal, ^{2,b)} Jakob S. Jensen, ³ and Mahmoud I. Hussein ² Department of Mechanical Engineering, Section for Solid Mechanics, Technical University of Denmark, 2800 Kgs. Lyngby, Denmark

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We investigate wave motion in a continuous elastic rod with a periodically attached inertial-amplification mechanism. The mechanism has properties similar to an "inerter" typically used in vehicle suspensions, however here it is constructed and utilized in a manner that alters the intrinsic properties of a continuous structure. The elastodynamic band structure of the hybrid rod-mechanism structure yields band gaps that are exceedingly wide and deep when compared to what can be obtained using standard local resonators, while still being low in frequency. With this concept, a large band gap may be realized with as much as twenty times less added mass compared to what is needed in a standard local resonator configuration. The emerging inertially enhanced continuous structure also exhibits unique qualitative features in its dispersion curves. These include the existence of a characteristic double-peak in the attenuation constant profile within gaps and the possibility of coalescence of two neighbouring gaps creating a large contiguous gap. © 2016 AIP Publishing LLC. [http://dx.doi.org/10.1063/1.4944429]

I. INTRODUCTION

The band structure of a material represents the relation between wave number (or wave vector) and frequency; thus it relates the spatial and temporal characteristics of wave motion in the material. This relation is of paramount importance in numerous disciplines of science and engineering such as electronics, photonics, and phononics. It is well-known that periodic materials exhibit gaps in the band structure, referred to as *band gaps* or *stop bands*. In these gaps, waves are attenuated whereby propagating waves are effectively forbidden. Their defining properties are the frequency range, i.e., position and width, as well as the depth in the imaginary part of the wave number spectrum, which describes the level of attenuation.

In the realm of elastic wave propagation, the two primary physical phenomena utilized for band-gap creation are Bragg scattering or local resonance (LR). Bragg scattering occurs due to the periodicity of a material or structure, where waves scattered at the interfaces cause coherent destructive interference, effectively cancelling the propagating waves. Research on waves in periodic structures dates back to Newton's attempt to derive a formula for the speed of sound in air, see e.g., Chapter 1 in Ref. 2 for a historical review before the 1950s. Later review papers on wave propagation in periodic media include Refs. 3–5.

The concept of local resonance is based on the transfer of vibrational energy to a resonator, i.e., a part of the material/structure that vibrates at characteristic frequencies. Within structural dynamics, the concept dates back, at least, to Frahm's patent application.⁶ Since then, dynamic vibration absorbers and tuned mass dampers have been areas of extensive research within structural vibration suppression. In the field of elastic band gaps, the concept of local resonance is often considered within the framework of periodic structures, as presented in the seminal paper of Liu et al., where band gaps are created for acoustic waves using periodically distributed internal resonators. The periodic distribution of the resonators does not change the local resonance effects; however, it does introduce additional Bragg scattering at higher frequencies, as well as allow for a unit-cell wave based description of the medium. Local resonance has also been used in the context of attaching resonators to a continuous structure, such as a rod, beam, or a plate 10,11 in order to attenuate waves by creating band gaps in the low frequency range. A problem with this approach in general, which has limited proliferation to industrial applications, is that the resonators need to be rather heavy for a band gap to open up at low frequencies.

Another means for creating band gaps is by the concept of inertial amplification (IA) as proposed by Yilmaz and collaborators in Refs. 12–14. In this approach, which has received less attention in the literature, inertial forces are enhanced between two points in a structure consisting of a periodically repeated mechanism. This generates *anti-resonance* frequencies, where the enhanced inertia effectively cancels the elastic force; see e.g., Ref. 15 where two levered mass-spring systems are analysed for their performance in generating stop bands. While it is possible to enhance the inertia between two points by means of masses, springs, and levers, a specific mechanical element, *the inerter*, ¹⁶ was created as the ideal inertial equivalent of

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springs and dampers, providing a force proportional to the relative acceleration between two points. This concept, while primarily used in vehicle suspension systems, ¹⁷ has been utilized in Refs. 12–14 in the context of generating band gaps in lattice materials by inertial amplification where the same underlying physical phenomenon is used for generating the anti-resonance frequencies. The frequency responses to various harmonic loadings were obtained, numerically and experimentally, and low-frequency, wide and deep band gaps were indeed observed for these novel lattice structures. In Ref. 18, size and shape optimization is shown to increase the band gap width further, as illustrated by a frequency-domain investigation.

Until now, both inerters and inertial amplification mechanisms have been used as a backbone structural component in discrete or continuous systems. In this paper, we propose to use inertial amplification to generate band gaps in conventional continuous structures, by attaching light-weight mechanisms to a host structure, such as a rod, beam, plate, or membrane, without disrupting its continuous nature (therefore not obstructing its main structural integrity and functionality). With this approach, we envision the inertial amplification effect to be potentially realized in the form of a *surface coating*, to be used for sound and vibration control.

For proof of concept, we consider a simple onedimensional case by analyzing an elastic rod, with an inertial amplification mechanism periodically attached. The mechanism is inspired by that analyzed in Ref. 12; however, the application to a continuous structure increases the practicality and richness of the problem considerably and several intriguing effects are illustrated.

Our investigation focuses mainly on the unit-cell band-structure characteristics. However, we also compare our findings from the analysis of the material problem to transmissibility results for structures comprising a finite number of unit cells. The finite systems are modelled by the finite-element (FE) method.

II. MODEL

In order to utilize the concept of inertial amplification in a surface setting as proposed, the mechanisms should be much smaller than the host structure such that their distributed attachment does not change the main function of the structure, nor occupy a significant amount of space. Fulfilling this constraint requires a relatively large effect with only a modest increase in mass.

Considering the ideal mechanical element, the inerter, we know that the factor of proportionality, the inertance, can be much larger than the actual mass increase, as demonstrated experimentally in Refs. 16 and 19. We propose to utilize the same effect using a mechanism similar to the one considered in Refs. 12 and 14. Our two-dimensional interpretation of the system is seen in Fig. 1, illustrating the comparably small inertial amplification mechanisms distributed over the host-structure. In principle, the distributed effect of the mechanisms, in the long wave-limit, reduces to the notion of an inertially modified constitutive relation in the elastodynamic equations.

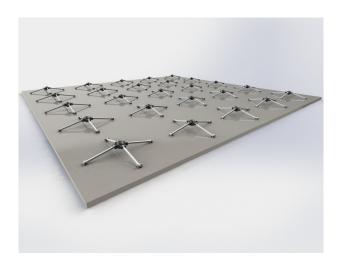


FIG. 1. General 2D realization of the proposed inertially amplified system.

A. Model reduction

In this study, we restrict ourselves to a one-dimensional structure with an inertial amplification mechanism attached, as illustrated in Fig. 2, where the mechanism is attached to the rod with bearings. A similar bearing is used at the top connection such that, ideally, no moment is transferred through the mechanism. This ensures that the connecting links do not deform, but move the amplification mass by rigid-body motion. The 1D-system is simplified further to a hybrid model consisting of a continuous, elastic bar and a discrete mechanism as seen in Fig. 3, as this allows for a rigorous analytical formulation for the underlying dynamics.

The bar has Young's modulus E, cross-sectional area A, mass density ρ , and unit-cell length $l=l_1+l_2+l_3$, while the amplification mass is denoted m_a and θ is the amplification angle. In Fig. 3, heavy lines indicate rigid connections and the corners between vertical and inclined rigid connections are moment-free hinges; hence, the motion of the amplification mass quantified by $z_1(t)$ and $z_2(t)$ is governed by the motion at the attachment points $u(x_1,t), u(x_2,t)$ and the amplification angle θ , where $x_1=l_1$ and $x_2=l_1+l_2$. From a physical standpoint, any increase in static stiffness of the mechanism would arise from frictional stiffness in the bearings or at the top point; however, it is outside the scope of this work to include these residual stiffness effects, among other things, since they are assumed to be small.

The inertial amplification model in Fig. 3 assumes rigid connections between the rod and the mechanism. Should the connections be flexible as illustrated in Fig. 4, the unit cell may be tuned to respond as either a locally resonant or an inertially amplified medium, depending on the specific system parameters. With the flexible springs, the two additional

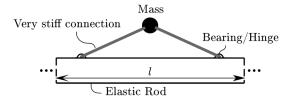


FIG. 2. 1D version of the proposed concept.

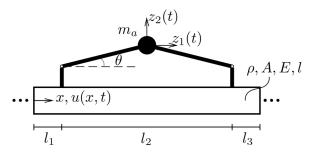


FIG. 3. Hybrid continuous-discrete model.

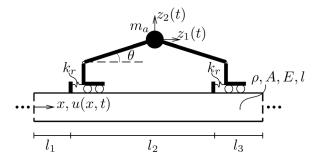


FIG. 4. Hybrid model including connection flexibility.

degrees of freedom introduced by the general mechanism create two local resonances which open up two LR-hybridization gaps. The IA mechanism on the other hand, being fixed to the host structure, does not add any additional degrees of freedom. Instead, it alters the intrinsic properties of the material and introduces anti-resonance frequencies throughout the entire frequency range. The LR type system recovers the ideal inertial amplification system for $k_r \to \infty$, and a standard local resonance system is realized when $l_2 = 0$.

The inertial amplification system in Fig. 3 is the main system investigated in this paper, while the system including the connection flexibility is used to investigate the transition between the fundamentally different types of gap generation possible for the general rod-mechanism configuration.

All the analytical formulations in this paper are based on the differential equation of a rod

$$\rho \ddot{u}(x,t) = \sigma'(x,t) + f(x,t), \tag{1}$$

where u(x, t) is the longitudinal displacement and $\sigma(x, t)$ is the normal stress while $() = \partial^2()/\partial t^2$ and $()' = \partial()/\partial x$. The body force f(x, t) will not be present in the material problem formulation considering infinite domains. The rod is considered to be homogeneous; however, a layered rod would pose no additional difficulty in terms of the transfer matrix method described in Section III B, since the method is applicable to layered materials. The rod is further assumed to be linear elastic with infinitesimal strains, which provides the constitutive relation

$$\sigma(x,t) = E\varepsilon(x,t) = Eu'(x,t),\tag{2}$$

where $\varepsilon(x,t)$ is the longitudinal strain in the rod.

B. Mechanism equations

Before considering the hybrid systems in Figs. 3 and 4, the mechanisms are considered with constraint forces

applied to account for the rod. These constraint forces are determined in terms of the mechanism parameters, whereby the effect of the mechanism on the rod is given in terms of these constraint forces.

1. Kinematics

Considering the isolated inertial amplification mechanism in Fig. 5, the motions z_1 and z_2 can be determined in terms of y_1 , y_2 , and θ . In the Appendix, the full nonlinear kinematic relations are derived. In this paper, we consider the linearized version, whereby z_1 and z_2 are determined as

$$z_1 = \frac{1}{2}(y_2 + y_1), \tag{3a}$$

$$z_2 = \frac{1}{2}\cot\theta(y_2 - y_1),$$
 (3b)

as seen in Ref. 12 as well.

2. Constraint forces

Considering the inertial amplification mechanism in Fig. 5, with the applied constraint forces P_1 and P_2 , the motions y_1 and y_2 correspond to longitudinal motion of the rod at the points x_1 and x_2 . Using Lagrange's equations, the governing equations for the mechanism are found

$$m_1 \ddot{y}_1 - m_2 \ddot{y}_2 = P_1, \tag{4a}$$

$$m_1\ddot{y}_2 - m_2\ddot{y}_1 = P_2,$$
 (4b)

where $m_1 = \frac{m_a}{4}(\cot^2\theta + 1)$ and $m_2 = \frac{m_a}{4}(\cot^2\theta - 1)$. Assuming harmonic motion, $y_j = Y_j e^{i\omega t}$, we obtain

$$k_1(\omega)Y_1 - k_2(\omega)Y_2 = \hat{P}_1(\omega),$$
 (5a)

$$k_1(\omega)Y_2 - k_2(\omega)Y_1 = \hat{P}_2(\omega),$$
 (5b)

where \hat{P}_j are the frequency domain representations of P_j and the dynamic stiffness parameters $k_i(\omega)$ are defined as

$$k_i(\omega) = -\omega^2 m_i, \quad i = 1, 2.$$
 (6)

Next, the constraint forces for the system with flexible connections in Fig. 4 are determined. Considering the isolated mechanism in Fig. 6, the constraint forces are applied at the constrained coordinates v_1 and v_2 , while the coordinates y_1 and y_2 are free.

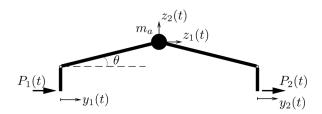


FIG. 5. IA mechanism with constraint forces.

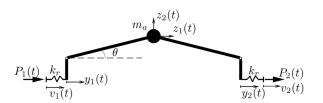


FIG. 6. LR mechanism with constraint forces.

The governing equations are found by Lagrange's equations

$$k_r(v_1 - y_1) = P_1,$$
 (7a)

$$m_1\ddot{y}_1 + k_r y_1 - m_2\ddot{y}_2 - k_r v_1 = 0,$$
 (7b)

$$m_1\ddot{y}_2 + k_r y_2 - m_2\ddot{y}_1 - k_r v_2 = 0,$$
 (7c)

$$k_r(v_2 - y_2) = P_2.$$
 (7d)

Assuming harmonic motion provides the frequency domain constraint forces, which in terms of the constraint coordinates V_1 and V_2 are

$$k_1^{lr}(\omega)V_1 - k_2^{lr}(\omega)V_2 = \tilde{P}_1(\omega),$$
 (8a)

$$k_1^{lr}(\omega)V_2 - k_2^{lr}(\omega)V_1 = \tilde{P}_2(\omega),$$
 (8b)

with the dynamic stiffness parameters defined as

$$k_1^{lr}(\omega) = \frac{\omega^2 \omega_{lr,1}^2 \left(\omega^2 (m_1 + m_2) - \omega_{lr,2}^2 m_1 \right)}{\left(\omega_{lr,1}^2 - \omega^2 \right) \left(\omega_{lr,2}^2 - \omega^2 \right)}, \quad (9a)$$

$$k_2^{lr}(\omega) = \frac{-\omega^2 \omega_{lr,1}^2 \omega_{lr,2}^2 m_2}{\left(\omega_{lr,1}^2 - \omega^2\right) \left(\omega_{lr,2}^2 - \omega^2\right)},\tag{9b}$$

where $\omega_{lr,1}$ and $\omega_{lr,2}$ are the local resonance frequencies defined by

$$\omega_{lr,1} = \sqrt{\frac{k_r}{m_1 + m_2}} = \sqrt{\frac{2k_r}{m_a \cot^2 \theta}},$$
 (10a)

$$\omega_{lr,2} = \sqrt{\frac{k_r}{m_1 - m_2}} = \sqrt{\frac{2k_r}{m_a}}.$$
 (10b)

Eq. (10a) corresponds to the out-of-phase mode, while Eq. (10b) corresponds to the in-phase mode of the mechanism. It is noted that the dynamic stiffness coefficients for the inertial amplification system are recovered from Eq. (9) when $k_r \to \infty$.

III. ANALYSIS

In order to characterize the effects of the inertial amplification mechanism, the band structure of an infinite array

of hybrid rod-mechanism systems is determined using the transfer-matrix method. The method has its origins within electrodynamics and optics, ²⁰ but has been widely used within elastic wave propagation. ^{21,22} The method is briefly described in Section III B, with a focus on the specific extension required for the particular unit cell considered here. Before describing the general transfer-matrix methodology, we consider a simplified unit cell in Section III A in order to shed light on the band-opening mechanism. We do this by a receptance approach, 23 where we determine the displacement at one end of a single unit cell when applying harmonic forcing at the other end. The anti-resonance frequencies can then be determined as those frequencies with zero receptance for any forcing magnitude (zeros). These anti-resonance frequencies are shown to be the points of maximum attenuation in the infinite system, and are thus relevant quantities for maximum attenuation design.

A. Receptance approach

Consider the simplified rod-mechanism system in Fig. 3 with free boundary conditions, $l_1 = l_3 = 0$, and harmonic forcing, $F = F_0 e^{i\omega t}$, at x = 0. Both the applied and constraint forces can be included via the boundary conditions to the rod differential equation, Eq. (1). With harmonic forcing, the linear response will be harmonic, $u(x,t) = \bar{u}(x)e^{i\omega t}$, whereby

$$\bar{u}''(x) + \kappa_b^2 \bar{u}(x) = 0, \quad \kappa_b = \frac{\omega}{c_0},$$
 (11)

with wave number κ_b and wave-speed $c_0 = \sqrt{E/\rho}$ in the homogeneous rod. The solution to (11) is

$$\bar{u}(x) = \alpha \sin \kappa_b x + \beta \cos \kappa_b x, \tag{12}$$

where the constants α and β are determined by the boundary conditions, given by force equilibria at both ends. Utilizing the constitutive relation $\bar{\sigma}(x) = E\bar{u}'(x)$, the force equilibria yield

$$x = 0$$
: $EA\bar{u}'(0) = F_0 - \hat{P}_1(\omega)$, (13a)

$$x = l: \quad EA\bar{u}'(l) = -\hat{P}_2(\omega), \tag{13b}$$

which, when inserting the constraint forces from Eq. (5) and the solution $\bar{u}(x)$ from Eq. (12), is expressed in matrix form

$$\begin{bmatrix} 1 - \hat{k}_2 \sin \kappa_b l & \hat{k}_1 - \hat{k}_2 \cos \kappa_b l \\ \cos \kappa_b l + \hat{k}_1 \sin \kappa_b l & \hat{k}_1 \cos \kappa_b - \sin \kappa_b l - \hat{k}_2 \end{bmatrix} \mathbf{x} = \mathbf{f},$$
(14)

with $\hat{k_i} = k_i(\omega)/(EA\kappa_b)$, $\mathbf{x} = [\alpha \beta]^T$, and $\mathbf{f} = [F_0/(EA\kappa_b)0]^T$. Solving for α and β provides the solution for $\bar{u}(x)$

 $\bar{u}(x) = \frac{\left(\left(\sin\kappa_b l - \hat{k}_1\cos\kappa_b l + \hat{k}_2\right)\sin\kappa_b x + \left(\cos\kappa_b l + \hat{k}_1\sin\kappa_b l\right)\cos\kappa_b x\right)}{\sin\kappa_b l\left(1 + \hat{k}_1^2 - \hat{k}_2^2\right)EA\kappa_b} F_0 = H_{0x}(\omega)F_0, \tag{15}$

where $H_{0x}(\omega)$ is the receptance function. The displacement at x = l is given by

$$\bar{u}(l) = \frac{1 + \hat{k}_2 \sin \kappa_b l}{\sin \kappa_b l \left(1 + \hat{k}_1^2 - \hat{k}_2^2\right) E A \kappa_b} F_0 = H_{0l}(\omega) F_0, \quad (16)$$

whereby the anti-resonance frequencies can be determined as the frequencies satisfying $H_{0l}(\omega) = 0$, i.e.,

$$k_2(\omega)\sin\kappa_b l + EA\kappa_b = 0. \tag{17}$$

This transcendental equation is solved numerically for any desired number of anti-resonance frequencies. The approximation for the first anti-resonance frequency is found in the long-wavelength limit where $\sin \kappa_b l \approx \kappa_b l$, i.e., in the sub-Bragg regime. The approximation $\tilde{\omega}_{a,1}$ is

$$-\omega^2 m_2 \kappa_b l + EA \kappa_b = 0 \Rightarrow$$

$$\tilde{\omega}_{a,1} = \sqrt{\frac{EA/l}{m_2}} = \sqrt{\frac{EA/l}{m_a (\cot^2 \theta - 1)/4}},$$
(18)

which is essentially the same as the anti-resonance frequency of a discrete system as presented in Ref. 12 with effective spring stiffness k = EA/l. Hence, discretizing the rod as a spring-mass system would provide the semi-infinite gap presented in the mentioned reference. The added complexity from the rod is illustrated by the higher roots of Eq. (17) and will be apparent from the band structures calculated in Section IV.

B. Transfer matrix method

The transfer matrix method is based on relating the state variables of a system across distances and interfaces, successively creating a matrix product from all the "sub" transfer matrices, forming the *cumulative transfer matrix*.

Consider the hybrid continuous-discrete, rod-mechanism system illustrated in Fig. 3, where the rod is modelled as a continuum and the mechanism is modelled by discrete elements. The transfer matrix for the unit cell is based on the host medium, i.e., the rod, representing the effects of the mechanisms by point force matrices at x_1 and x_2 . The state variables for the rod are the longitudinal displacement u(x, t) and the normal stress $\sigma(x, t)$. Dividing the system into three layers separated at x_1 and x_2 , the solution for the longitudinal displacement in layer j can be written as a sum of forward and backward travelling waves

$$u_j(x,t) = (B_j^{(+)}e^{i\kappa_b x} + B_j^{(-)}e^{-i\kappa_b x})e^{i\omega t},$$
 (19)

where $B_j^{(+)}$ and $B_j^{(-)}$ are the amplitudes of the forward and backward travelling waves. Using the linear elastic constitutive relation for the stress, the state variables are expressed as

$$\mathbf{z}_{j}(x) = \begin{bmatrix} u_{j}(x) \\ \sigma_{j}(x) \end{bmatrix} = \begin{bmatrix} 1 & 1 \\ Z & -Z \end{bmatrix} \begin{bmatrix} B_{j}^{(+)} e^{i\kappa_{b}x} \\ B_{j}^{(-)} e^{-i\kappa_{b}x} \end{bmatrix}$$
$$= \mathbf{H} \begin{bmatrix} B_{j}^{(+)} e^{i\kappa_{b}x} \\ B_{j}^{(-)} e^{-i\kappa_{b}x} \end{bmatrix}, \tag{20}$$

thus defining the **H**-matrix, where $Z = iE\kappa_b$. Relating the state variables at either end of a homogeneous layer separated by the distance l_i yields

$$\mathbf{z}_{j}^{R} = \mathbf{H} \begin{bmatrix} e^{i\kappa_{b}l_{j}} & 0\\ 0 & e^{-i\kappa_{b}l_{j}} \end{bmatrix} \begin{bmatrix} B_{j}^{(+)}e^{i\kappa_{b}x^{j,L}}\\ B_{j}^{(-)}e^{-i\kappa_{b}x^{j,L}} \end{bmatrix}$$
$$= \mathbf{H}\mathbf{D}_{j} \begin{bmatrix} B_{j}^{(+)}e^{i\kappa_{b}x^{j,L}}\\ B_{j}^{(-)}e^{-i\kappa_{b}x^{j,L}} \end{bmatrix}, \tag{21}$$

defining the "phase-matrix" \mathbf{D}_j . The coordinate at the left end of layer j is denoted $x^{j,L}$. Solving Eq. (20) with $\mathbf{z}_j = \mathbf{z}_j^L$ for the vector of amplitudes and inserting into Eq. (21) define the transfer matrix for layer j

$$\mathbf{z}_{j}^{R} = \mathbf{H} \mathbf{D}_{j} \mathbf{H}^{-1} \mathbf{z}_{j}^{L} = \mathbf{T}_{j} \mathbf{z}_{j}^{L}$$

$$\mathbf{T}_{j} = \begin{bmatrix} \cos \kappa_{b} l_{j} & \frac{1}{E \kappa_{b}} \sin \kappa_{b} l_{j} \\ -E \kappa_{b} \sin \kappa_{b} l_{j} & \cos \kappa_{b} l_{j} \end{bmatrix}.$$
(22)

Having defined the transfer matrices T_j , j = 1, 2, 3, we turn to the constraint forces at the attachment points of the mechanism.

We base our derivation of the point force matrices on a frequency domain force equilibrium at the attachment points, considering x_1 first

$$A\sigma_2^L = A\sigma_1^R + \hat{P}_1 = A\sigma_1^R + k_1(\omega)u_1^R - k_2(\omega)u_2^R.$$
 (23)

It is noted that the force balance at point x_1 depends on the displacement at x_2 . Using the transfer matrix for layer 2, u_2^R is expressed as

$$u_2^R = u_2^L \cos \kappa_b l_2 + \frac{\sigma_2^L}{E_{K_b}} \sin \kappa_b l_2, \tag{24}$$

which, along with the continuity requirement $u_1^R = u_2^L$, yields the force equilibrium

$$\left(A + \frac{k_2(\omega)}{E\kappa_b}\sin\kappa_b l_2\right)\sigma_2^L = A\sigma_1^R + (k_1(\omega) - k_2(\omega)\cos\kappa_b l_2)u_1^R,$$
(25)

whereby the point force matrix, relating the state vector \mathbf{z}_2^L to \mathbf{z}_1^R , can be identified

$$\mathbf{z}_{2}^{L} = \begin{bmatrix} 1 & 0 \\ \frac{E\kappa_{b}(k_{1}(\omega) - k_{2}(\omega)\cos\kappa_{b}l_{2})}{AE\kappa_{b} + k_{2}(\omega)\sin\kappa_{b}l_{2}} & \frac{AE\kappa_{b}}{AE\kappa_{b} + k_{2}(\omega)\sin\kappa_{b}l_{2}} \end{bmatrix} \times \mathbf{z}_{1}^{R} = \hat{\mathbf{P}}_{1}\mathbf{z}_{1}^{R}.$$
(26)

Using a similar approach at point x_2 provides the point force matrix $\hat{\mathbf{P}}_2$ as

$$\hat{\mathbf{P}}_2 = \begin{bmatrix} 1 & 0 \\ \frac{k_1(\omega) - k_2(\omega)\cos\kappa_b l_2}{A} & 1 + \frac{k_2(\omega)}{AE\kappa_b}\sin\kappa_b l_2 \end{bmatrix}, (27)$$

which allows for relating the state vector at the right end of the unit cell to the state vector at the left end through the cumulative transfer matrix **T**

$$\mathbf{z}_3^R = \mathbf{T}_3 \hat{\mathbf{P}}_2 \mathbf{T}_2 \hat{\mathbf{P}}_1 \mathbf{T}_1 \mathbf{z}_1^L = \mathbf{T} \mathbf{z}_1^L. \tag{28}$$

The present framework is fully compatible with the local-resonator-type system described in Section II by changing the dynamic stiffness parameters in the point-force matrices to those of Eq. (9), rather than those defined by Eq. (5). Finally, it is noted that when the internal distance l_2 approaches zero, the point force matrices for the inertial amplification system approach that of an attached point mass, while the local-resonance point force matrices approach that of an attached local resonance with resonance frequency $\omega_{lr,2} = \sqrt{2k_r/m_a}$, recovering the expected limits.

With the cumulative transfer matrix for a unit cell determined, the Floquet-Bloch theorem for periodic structures²⁴ is used to relate the state vector at either end through a phase multiplier

$$\mathbf{z}_{3}^{R} = \mathbf{z}_{1}^{L} e^{i\kappa l}, \tag{29}$$

where $\kappa = \kappa(\omega)$ is the wave number for the periodic material and l is the unit-cell length. Combining Eqs. (28) and (29) yields a frequency-dependent eigenvalue problem in $e^{i\kappa l}$

$$(\mathbf{T} - e^{i\kappa l}\mathbf{I})\mathbf{z}_1^L = 0, (30)$$

whereby the band structure of our periodic material system is determined within a $\kappa(\omega)$ -formulation.

IV. BAND STRUCTURE

In this section, the band structure is calculated for the systems in Figs. 3 and 4 as well as a standard local resonator configuration, with primary focus on the inertial amplification system. The mechanism is attached to an aluminum rod with Young's modulus E, density ρ , width b, height h, and length l. These parameters are given in Table I along with the equivalent mass and stiffness parameters m_b and k_b and the first natural frequency ω_b .

The effect of the primary parameters of the system is investigated, with special attention devoted to gap width and depth. We begin by considering the case where the internal length of the mechanism is equal to the unit-cell length, i.e., $l_1 = l_3 = 0$, illustrating the effect of the added mass m_a , after which we consider the effect of reducing the mechanism size compared to the unit-cell length. Next, we consider the system with flexible connections and the transition from local resonance to ideal inertial amplification, after which we compare the performance of the IA system to that of a

TABLE I. Parameters of the host rod.

E (GPa)	$\rho~(\text{kg/m}^3)$	$b \times h(\mathbf{m}) \times (\mathbf{m})$	l(m)	ω_b (rad/s)	m_b (kg)	k_b (N/m)
69.8	2700	0.025×0.025	0.55	$\frac{\pi}{l}\sqrt{\frac{E}{\rho}}$	$\rho A l$	$\frac{EA}{l}$

TABLE II. Reference mechanism parameters.

m_a	θ	l
$m_b/10$	$\pi/18$	l

standard local resonator system where the resonator is attached to the rod at a point within the unit cell.

A. Band structure and anti-resonance frequencies

We begin by considering a reference case, with the rod parameters in Table I and the mechanism parameters given in Table II.

For the reference case, the anti-resonance frequencies are predicted by numerically solving Eq. (17), while the band structure is calculated using the transfer-matrix method described in Sec. III B. Figure 7 shows the band structure with the anti-resonance frequencies indicated by red crosses, in the normalized frequency range $\bar{\omega} = \omega/\omega_b$.

It is clear that the solution of Eq. (17) predicts the peak values of the imaginary part of the wave number. A feature unique to inertial amplification seen in the band structure in Fig. 7 is the appearance of two peaks within the odd-numbered gaps that are noted to have merged with the odd-numbered Bragg gaps with upper limits at $\bar{\omega} = 1$, 3, 5, This double-peak behaviour is previously observed in Refs. 14 and 15, however, in the frequency response functions of finite structures rather than the band structure of the system. The reason that the inertial amplification gaps and Bragg gaps merge is the fact that the mechanism is attached to the ends of the unit cell; hence, the wavelengths most significant for activation of the inertial amplification mechanism are the same as those relevant for Bragg scattering. The number of attenuation peaks within a band gap is determined by the number of anti-resonance frequencies separating the two wave modes on either side of the gap. Hence when two resonance frequencies (poles) are separated by more than one anti-resonance frequency (zero), multiple attenuation peaks will occur. The low-pass filters shown in Ref. 15 are designed such that all

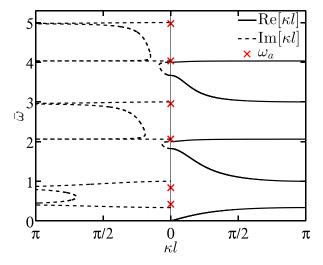


FIG. 7. Band-structure for reference case.

the anti-resonance frequencies are larger than the largest resonance frequency, whereby the multiple peaks are all found above the filter frequency. It is further noted that the even gaps are of Bragg-type; however, they exhibit an asymmetry, being distorted towards the double-peak gaps which is also common when Bragg gaps are close to the peak of local-resonance gaps.^{25,26}

When considering the band structure for increasing frequency, it appears that for higher band gaps, the anti-resonance frequencies get further separated, increasing the gap width at the cost of decreasing the gap depth. Furthermore, as the gap widens, the anti-resonance frequencies approach the gap limits. This indicates that the anti-resonance frequencies can be used as a design-parameter for gap width. This is investigated further in the Sec. IV C, where it is shown that this is only strictly true for "full-length" mechanisms, i.e., when $l_1 = l_3 = 0$.

B. Mass variation

Consider the model shown in Fig. 3 with $l_1 = l_3 = 0$. We investigate the effect of the attached mass m_a for the general rod parameters in Table I and the mechanism parameters in Table II. It is expected that increasing the attached mass will increase the gap width as well as decrease the frequency range. These effects are illustrated for distinct values of the mass ratio $\mu = m_a/m_b$ in Fig. 8(a) as well as by the attenuation profile in Fig. 8(b) which in this paper refers to a contour plot of the natural logarithm of the imaginary part of the normalized wave number. The four band structures in Fig. 8(a) are for the μ -values indicated by the four vertical dashed lines in the attenuation profile in Fig. 8(b). It is noted that for the mass ratio $\mu = 5\%$, the first gap is purely of Bragg-type since the inertial forces generated by the mechanism are not sufficient to generate anti-resonance frequencies, i.e., to fulfil Eq. (17). As μ is increased, the inertial forces are increased (by a factor of m_a), and $\mu = 7\%$ is seen to be the limiting value for generating true anti-resonance frequencies in the first gap. At this value, we see a single anti-resonance frequency which branches into two distinct anti-resonance frequencies as μ is further increased.

Aside from this double-peak behaviour, the width of the first gap is rather wide for relatively low mass-ratios, e.g., for a mass ratio of 10% we have a normalized first gap width of

$$\Delta \bar{\omega}_1 = \frac{\bar{\omega}_1^u - \bar{\omega}_1^l}{\bar{\omega}_1^c} = 99\%, \tag{31}$$

which, compared to gaps obtained by standard LR effects, is rather wide, as will be shown in Sec. IV E. The attenuation profile in Fig. 8(b) illustrates the continuous progression of the band gaps as the mass varies. Regions with color indicate band gaps and the color bar represents the magnitude of attenuation as a function of μ and $\bar{\omega}$. The logarithmic scale illustrates the order of magnitude of the gap depth. Between the peak values at the anti-resonances, the attenuation is seen to stay at a relatively high level, a level that decreases with gap-width as observed for increasing gap number in Section IV A.

C. Effect of attachment points

Investigating the effect of the relative mechanism length $\bar{l}=l_2/l$, it is expected that the performance metrics, i.e., the gap width and depth, of the mechanism will decrease with a smaller internal length due to the decreased lever-arm for the mass. This, however, is not a trivial investigation because decreasing the mechanism size can provide new ways for gaps to hybridize, e.g., the even-numbered wave-modes will be affected by the mechanism. Figure 9 illustrates the effect of varying the relative attachment length, $\bar{l}=l_2/l$, with four band diagrams in Fig. 9(a) and the attenuation profile in Fig. 9(b), for the general rod parameters in Table I and the mechanism parameters from Table II. The four band structures in Fig. 9(a) correspond to the vertical dashed lines in the attenuation profile in Fig. 9(b). Considering the attenuation profile first, the first striking feature is the multitude of coalescence

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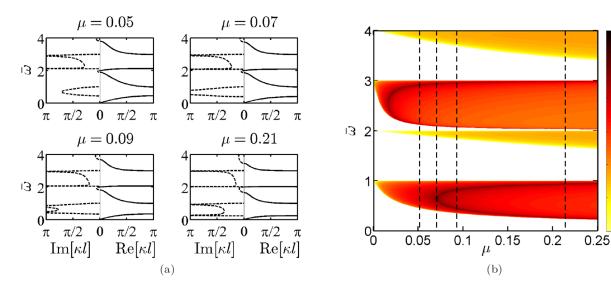


FIG. 8. Effect of amplification mass for full length mechanism. (a) Band structures. (b) Attenuation profile showing the normalized imaginary wave number in logarithmic scale.

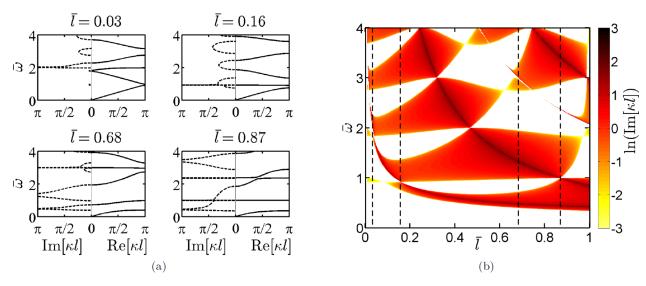


FIG. 9. Effect of decreasing $\bar{l} = l_2/l$. $\theta = \pi/18$, $\mu = 0.1$. (a) Band structures. (b) Attenuation profile.

points between the gaps, where the anti-resonance frequencies jump from one gap to another. At these coalescence points, the gap width is effectively increased to the sum of the two coalescent gaps, e.g., at l = 87%, we get an effective, normalized gap width of $\Delta \bar{\omega} = 136\%$. The corresponding band-structure illustrates that the two gaps do not coalesce completely, rather there is a standing-wave mode separating the two gaps. Hence, the second anti-resonance frequency is in fact larger than the second resonance frequency, whereby the first gap only contains one anti-resonance frequency. This indicates the presence of a critical region between the gap limits, especially for practical designs where design inaccuracies will make an exact realization of, for example, $\overline{l} = 87\%$ difficult. The three remaining band structures, at $\bar{l} = 3\%$, 16%, and 68%, respectively, illustrate various effects. The first of these illustrates the limiting behaviour towards the pure Bragg gaps from periodically attached masses when $\bar{l} \rightarrow 0$. It is noted that this transition occurs earlier for lower gaps, due to the shorter wavelengths of the higher gaps. The l = 16% band structure illustrates the first coalescence point between the first and second gap, the effect of which is not quite as powerful as for the $\bar{l}=87\%$ case; however, this can be explained by the smaller lever ratio, thus providing smaller inertial forces. The bandstructure for $\bar{l} = 68\%$ illustrates the first two distinct gaps, as well as a Bragg-type gap which is virtually unaffected by the anti-resonance frequency very close to it.

D. Transition from local resonance

In this section, we illustrate how the behaviour of the general local-resonator-type system illustrated in Fig. 4 tends toward that of the pure inertial amplification mechanism as the connection stiffness increases, $k_r \to \infty$. The local resonance frequencies seen in Eq. (10) are for the two modes supported by the mechanism, i.e., where the mechanism ends move out-of-phase and in-phase, respectively. Figure 10 shows the attenuation profile for increasing the relative connection stiffness $\bar{k}_r = k_r/k_b$, both in a low range to illustrate the local-resonance effect and a larger range to illustrate the limiting behaviour. The dash-dot lines are the resonance frequencies

for the local resonance-system predicted by Eq. (10), noted to follow gap limits, rather than peak attenuation points, as is the case for a standard local resonator attached at a point. This has a rather simple physical explanation when considering the working principle of the two types of systems. At their specific resonance frequencies, both resonators are maximally activated, but the fundamental difference lies in the number of attachment points. While the standard local resonator will "take out" some of the mechanical energy of the travelling wave, the two-terminal system will work as a path of travel for the wave whereby the mechanical energy is not "deposited" as in the standard local resonator case. Hence, the peak attenuation frequency for the system including connection flexibility should be predicted from the anti-resonance equation, Eq. (17), but with the dynamic stiffness coefficient given by Eq. (9b), $k_2 = k_2^{lr}$. The black horizontal lines in Fig. 10(b) are the anti-resonance frequencies predicted by Eq. (17) using the IA dynamic stiffness coefficient, $k_2 = -\omega^2 m_2$, in which the peak attenuation lines can be seen to converge towards, starting at the lower frequencies. Hence, as the connection stiffness increases, the lower gaps become dominated by inertial amplification effects, followed by the higher gaps. This is illustrated further by considering the band structures corresponding to the vertical dashed lines in Figs. 10(a) and 10(b). In the low-stiffness limit in Fig. 10(c), we see a behaviour that is similar to what is observed for classical local resonance; however, we know that the local resonance frequency actually represents the upper limit of the gap. The peak attenuation frequency however is the anti-resonance frequency, where energy cannot be transferred through the mechanism. In the high-stiffness limit in Fig. 10(d), we observe the ideal inertial amplification behaviour (where we have two anti-resonance frequencies separating the first and second wave modes leading to the appearance of the characteristic double peak).

E. Comparison to a standard local resonator

The performance metrics are now investigated for a rod with periodic, point-wise attached local resonators, in order

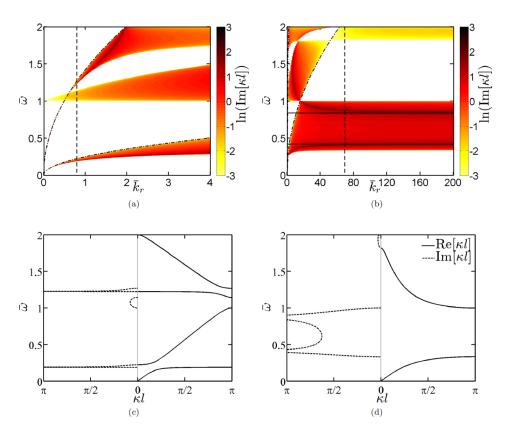


FIG. 10. Effect of connection stiffness $\bar{k}_r = k_r/k_b$. $\theta = \pi/18$, $\mu = 0.1$ $\bar{l} = 1$. (a) Attenuation profile in a low range. (b) Attenuation profile in a wide range. (c) Band structure for $\bar{k}_r = 0.8$ (local resonance). (d) Band structure for $\bar{k}_r = 69.3$ (inertial amplification).

to provide a direct performance comparison with the proposed inertial amplification system. The transfer matrix of such a rod is known and given in, e.g., Ref. 27. One strength of the classical local resonator configuration is the ability to create *any* anti-resonance frequency, given complete freedom in the design variables. This is equally possible for the inertial amplification configuration proposed here, which is evident from the anti-resonance equation, Eq. (17). Hence, the local resonator offers no design flexibility advantage over the inertial amplification mechanism in this regard. A primary parameter for both the inertial-amplification and local-resonator systems is the added mass m_a . The effect of the added mass is illustrated for the local-resonator system

for two tuning cases of local resonator stiffness, k_{eq} . One is where the local-resonance frequency is tuned to the approximate first anti-resonance frequency for the inertial amplification system, $\tilde{\omega}_{1,a}$ from Eq. (18), in order to attain the same peak attenuation frequency for the two systems. The second case is where the local resonator system is tuned to allow the local-resonance gap to coalesce with the first Bragg gap-since; it is known that this is where local resonance gaps achieve maximal width. ^{25,27,28} Fig. 11 illustrates the effect of increasing the mass ratio μ for the two cases. The attenuation profile seen in Fig. 11(a) illustrates how the local-resonance gap decreases in frequency range with increasing mass, while the Bragg gaps stay centred around the same

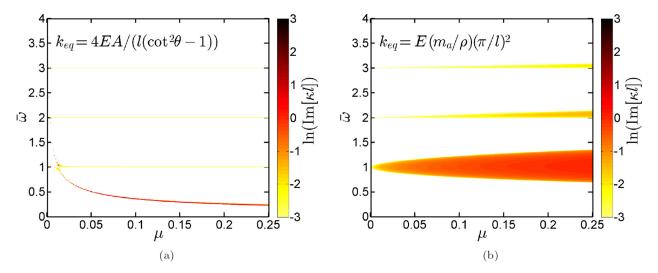


FIG. 11. Attenuation profiles illustrating the effect of the mass ratio μ in a standard local resonance configuration. (a) Stiffness tuning for similar first peak attenuation frequency. (b) Stiffness tuning for local-resonance-Bragg coalescence.

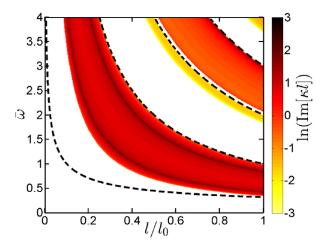


FIG. 12. Attenuation profile illustrating the dependence on unit-cell size. Local resonator: dashed lines, Inertial amplifier: colored regions.

frequency with mass increase. Comparing to the inertial amplification system for $l_1 = l_3 = 0$, the inertial amplification system is seen to perform better in terms of both gap width and depth, see Fig. 8. The attenuation profile for local-resonance-Bragg coalescence in Fig. 11(b) shows superior gap width than seen for the local-resonance attenuation profile in Fig. 11(a); however, the lowest and widest gap is centred around the first Bragg-gap frequency, thus not providing any low-frequency attenuation. Furthermore, the inertial amplification configuration still offers a wider and deeper gap, while being lower in frequency.

F. Effect of unit-cell size

Now the attenuation profile for the inertial amplification system is compared to that of a local resonance system when the unit-cell size is decreased. This is done to examine the practical limitation of the inertial amplification system, given that an internal length is needed to generate the enhanced inertia to cancel the elastic forces in the rod. This limitation is similar to the one investigated in Sec. IV C concerning the internal length between attachment points, \bar{l} ; however, now we look into the unit cell size l rather than relative internal length. The standard local resonance system does not need any internal length and, as such, can operate unaltered as the unit-cell size goes towards zero. Figure 12 compares the attenuation profile for the two systems, when decreasing the unit cell size l/l_0 for the parameters seen in Table III, where both μ and \bar{k}_{eq} are based on the original length l_0 . The dashed lines represent the gap frequencies for the localresonator system. The first line represents the local resonance frequency $\omega_{eq} = \sqrt{k_{eq}/m_a}$ where k_{eq} is kept constant, while m_a decreases with unit cell size as μ is kept constant. The subsequent lines represent the Bragg-gap limits ω_{Bragg} $= j\omega_b, j = 1, 2, 3, \dots$ For this particular mass ratio

TABLE III. Parameters for unit-cell size investigation.

l_0 (m)	μ	θ (rad)	$ar{k}_{eq}$	$l_1 = l_3$
0.55	0.1	$\pi/18$	0.1	0

 $\mu = 10\%$, the actual gaps are very narrow. Thus, they are covered entirely by the line representation in Fig. 12.

The conclusion in this section is that for realistic and comparable parameters, the local-resonance system can provide wave attenuation at lower frequencies than the inertial-amplification system, as the unit-cell size decreases. This is a clear advantage of the classical locally resonant systems from the point of view of constraints on unit-cell size.

V. FINITE STRUCTURES

In this section, the band-structure results from Section IV are compared to the transmissibility for finite systems comprising a certain number of unit cells, illustrating that the material results are representative in a structural setting as well. The transmissibilities are calculated from a finite-element model of the continuous-discrete system in Fig. 3, using standard linear elements to discretize the continuous rod, with stiffness and mass matrix contributions from the mechanism at appropriate nodes. The simple FE model is compared to an FE model created in the commercial software ABAQUS, using 3D beam elements.

A. Transmissibility gaps

Using the FE implementation to model the 1D finite array illustrated in Fig. 13 with the number of unit cells denoted n, the transmissibility is expressed as the natural logarithm to the ratio between the output and input displacement divided by the number of unit cells. Hence, the transmissibility expresses the wave propagation/decay per unit cell, and is thus comparable to the dispersion curves.

The transmissibilities are calculated for the same aluminum rod as considered in Section IV, using the same rod parameters from Table I. We consider the case where the relative mechanism length is equal to the full unit cell length, i.e., $\bar{l} = 1$. Figure 14 compares the band structures from the infinite systems to the transmissibilities calculated using n=5 for two values of the mass-ratio μ , illustrating the branching point at $\mu = 7\%$ and the double-dip at $\mu = 21\%$ corresponding to the single- and double-peak in the attenuation profiles, respectively. The grey areas correspond to band gaps predicted by the infinite model. It is noted that some boundary effects exist, i.e., the finite systems have resonances within band-gaps, which is due to the symmetry breaking of the system. ^{29,30} In spite of the boundary effects, the comparison shows that the infinite-system properties carry over to the finite case, and perhaps equally important that within the band gaps, the curves for the imaginary part of the wavenumber and the transmissibility have similar shapes and magnitudes. This is expected from the exponential decaying behaviour of waves within the band gaps, but it does illustrate the design potential for finite structures by just considering the shape of the band-structure within the gaps calculated for infinite structures.

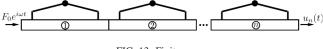


FIG. 13. Finite array.

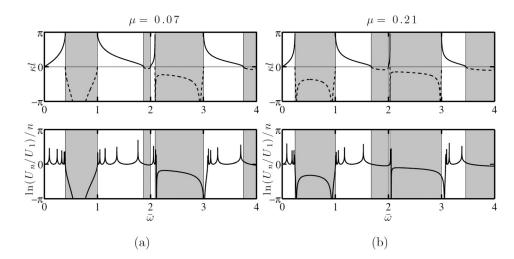


FIG. 14. Comparison of band structures to transmissibilities for $\bar{l}=1$. (a) Single peak behaviour. (b) Double peak behaviour.

B. ABAQUS verification

The FE implementation of the hybrid rod-mechanism system is tested against an implementation of the finite system in the commercial FE-software ABAQUS. The ABAQUS model is created as a 3D deformable "wire" model, using three-dimensional beam elements for both rod and mechanism. The mechanisms are distributed above and below the main structure to have equal but opposite transverse force components from the mechanisms. This is necessary to avoid bending phenomena, and may easily be implemented in an experimental setting as well. The rigid connecting links are modelled by assigning very large Young's modulus and very low density to the elements. The ideal connections are modelled using translatory constraints to connect the rigid connectors to the bar. Hence, the ABAQUS model is used to illustrate the phenomena in a finite setting without obstructing

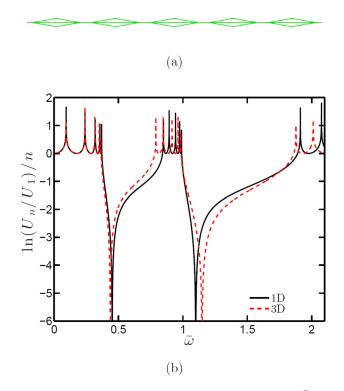


FIG. 15. (a) ABAQUS Model. (b) Comparison of 3D to 1D model. $\bar{l}=0.8,$ $\mu=0.1.$

the results with, for the present purpose, unnecessary complexities. Indeed, it is a subject of a future research paper to investigate more realistic models of the physical configuration in Fig. 2 both numerically and experimentally. The ABAQUS model is created with the general rod parameters seen in Table I and the general mechanism parameters $\theta = \pi/18$ and $\bar{l} = 0.8$. An illustration of the created ABAQUS model is seen in Fig. 15(a).

Figure 15(b) shows the comparison between the transmissibilities calculated by the 1D FE implementation of the rod-mechanism system and the 3D FE implementation in ABAQUS, respectively, for the mass ratio $\mu = 10\%$.

Comparing both maximum attenuation frequencies and gap limits, the transmissibility predicted by ABAQUS matches rather well, especially for the first gap. Concerning the deviations in the transmissibilities, it is worth considering the case of gap-coalescence which, as seen in Fig. 9, is a rather "singular" phenomena. As expected, using the "coalescence-parameters" predicted by the analytical model does not cause the gaps to coalesce in the 3D model. Figure 16 shows the transmissibility comparison for the analytically predicted coalescence-parameters and the ones found by inverse analysis in ABAQUS. This pass band could be detrimental for design if not taken into account, since the resonances are so closely spaced. Hence, designing for gap-coalescence should be done with care, as mentioned in Sec. IV C.

VI. CONCLUSIONS

We have investigated the wave characteristics of a continuous rod with a periodically attached inertial amplification mechanism. The inertial amplification mechanism, which is based on the same physical principles as the classical inerter, creates band gaps within the dispersion curves of the underlying continuous rod. The gap-opening mechanism is based on an enhanced inertial force generated between two points in the continuum, proportional to the relative acceleration between these two points. An inertial amplification mechanism has been used previously as a core building block for the generation of a lattice medium, rather than serve as a light attachment to a continuous structure as done here. Several prominent effects are featured in the emerging band structure of the hybrid rod-mechanism configuration. The anti-

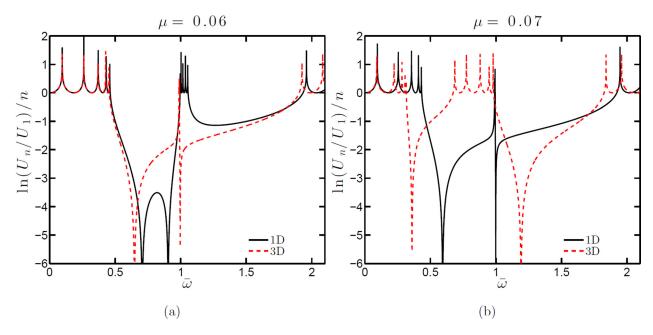


FIG. 16. Transmissibility comparison for coalescence in 1D and 3D models for (a) $\mu = 0.06$ and (b) $\mu = 0.07$.

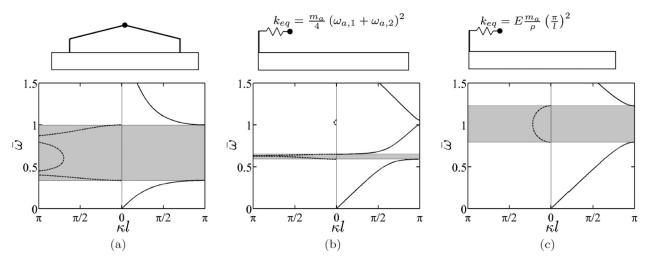


FIG. 17. Various band-gap opening mechanisms for constant mass ratio $\mu=10\%$ $\bar{l}=1$. (a) Inertial amplification. (b) Local resonance. (c) Bragg scattering.

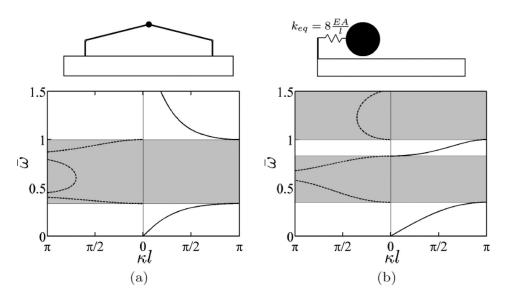


FIG. 18. Band structure for (a) proposed inertially amplified rod with $\mu=10\%,\ \bar{l}=1$ and (b) standard locally resonant rod with $\mu=204\%.$ The systems have almost equal-sized gaps despite a factor of 20 difference in added mass size.

resonance frequencies are governed by both mechanism- and rod-parameters. Hence, rather than a single anti-resonance frequency, we see an infinite number (by virtue of the continuous nature of the underlying rod), all of which can be predicted for a simple choice of unit-cell parameters. For an example choice of parameters, we illustrate the presence of multiple attenuation peaks within the same gap. Furthermore, when generalizing the parameters of the unit cell, we observe that the anti-resonance frequencies can jump between gaps whereby double-peak behaviour cannot be guaranteed for all parameters. At the specific values of the anti-resonance jump, band-gap coalescence emerges providing a very wide and deep contiguous gap. This gap, however, is rather sensitive to design and modelling inaccuracies.

In addition to these intriguing effects, we demonstrate how attaching an inertial amplification mechanism to a continuous structure may be practically superior to attaching a classical local resonator in that the former produces much larger gaps for the same amount of added mass. Figure 17 compares the band structure of the proposed inertial amplification system to those of a classical local resonator configuration for two different tunings of the local resonator stiffness k_{eq} . Figures 17(b) and 17(c) represent two cases of stiffness tuning that provide a locally resonant band gap (with equal central frequency) and a Bragg coalescence gap, respectively. The central frequency is determined by solving Eq. (17) numerically for the first two roots, $\omega_{a,1}$ and $\omega_{a,2}$. The comparison illustrates that when the same mass is used, the proposed concept achieves a first gap that is much wider than what is obtainable by the classical local resonator configuration, irrespective of the stiffness tuning for the local resonator. In order to obtain comparable performance in terms of band-gap width for the classical local-resonator system, the added mass m_a should be increased significantly. Figure 18 compares similar gap widths for an inertial amplification system and a local resonance system. From the figure, we see that the inertial amplification system is superior in terms of the magnitude of added mass, as the local resonance system requires an approximately twenty times heavier mass to obtain a comparable band-gap width (a mass that is more than two times as heavy as the rod it is attached to). The classical local resonator configuration, on the other hand, faces less constraints on unit-cell size as demonstrated in Fig. 12.

The presented concept of an inertially amplified continuous structure opens a new promising avenue of band-gap design. Potentially, it could be extended to surfaces of more complex structures such as plates, shells, and membranes leading to a general surface-coating design paradigm for wave attenuation in structures. Steps toward achieving this goal include a generalization of the formulation to admit transverse vibrations, incorporation of frictional stiffness and damping in the bearings of the mechanism, and generalization to two dimensions.

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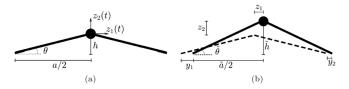


FIG. 19. Top triangle kinematics. (a) Undeformed triangle. (b) Deformed triangle.

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APPENDIX: KINEMATIC DERIVATION

Considering the top part of the deformed mechanism as illustrated in Fig. 19, the motions z_1 and z_2 can be determined in terms of y_1 , y_2 , and θ . The horizontal motion z_1 is by geometric consideration determined as $z_1 = (y_1 + y_2)/2$. The vertical motion is given by the difference $z_2 = \tilde{h} - h$. It turns out to be convenient to consider the difference of the squared triangle heights

$$\tilde{h}^{2} - h^{2} = \left(l^{2} - \left(\frac{\tilde{a}}{2}\right)^{2}\right) - \left(l^{2} - \left(\frac{a}{2}\right)^{2}\right)$$

$$= \frac{a^{2}}{4} - \frac{(a + y_{2} - y_{1})^{2}}{4}$$

$$= \frac{1}{4}\left(2a(y_{2} - y_{1}) - (y_{2} - y_{1})^{2}\right). \tag{A1}$$

Expressing the difference in squared heights as $\tilde{h}^2 - h^2 = (\tilde{h} + h)(\tilde{h} - h) = (2h + z_2)z_2$ and using a geometric relation for h lead to

$$(z_2 + a \tan \theta)z_2 = \frac{1}{4} \left(2a(y_2 - y_1) - (y_2 - y_1)^2 \right),$$
 (A2)

which provides a quadratic equation in z_2 . The equation is made explicit by assuming small displacements such that $(y_2 - y_1)^2 \ll 2a(y_2 - y_1)$ and $z_2 \ll a \tan \theta = h$, whereby the linearized kinematics of the mechanism has been determined as

$$z_1 = \frac{1}{2}(y_2 + y_1),$$
 (A3a)

$$z_2 = \frac{1}{2} \cot \theta (y_2 - y_1).$$
 (A3b)

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