



## **A systematic investigation of the effects of process parameters on heat and fluid flow and metallurgical conditions during laser-based powder bed fusion of Ti6Al4V alloy**

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1 **A Systematic investigation of the effects of process parameters on heat and fluid flow and**  
2 **metallurgical conditions during laser-based powder bed fusion of Ti6Al4V alloy**

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6 **Abstract**

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Additive manufacturing (AM) of metals faces a growing number of applications in different industries e.g. aerospace, medical, automotive, etc. Although metal AM outweighs current conventional production methods in some certain areas, the exact effect of processing conditions on the final quality and microstructure of the parts is still not well understood. An efficient way of understanding the effect of these processing conditions on a part's quality is via a calibrated and validated numerical model. Hence, in the current work a finite element model for analyzing the heat and fluid flow along with metallurgical conditions during Laser-based Powder Bed Fusion (L-PBF) of a titanium alloy has been developed and implemented in the commercial software code COMSOL Multiphysics. The thermal effect of the laser is modelled via a novel conico-Gaussian moving heat source, based on the concept of modified optical penetration depth. Analytical expressions for the geometrical distribution of the heat source are derived to obtain the heat source's effective depth. The model has been both verified and validated through mesh sensitivity analysis and comparison with experimental results. Furthermore, a detailed description about the role of the various driving forces for fluid flow has been given based on a thorough analysis using relevant dimensionless numbers. A systematic procedure to study the influence of neglecting the fluid flow inside the melt pool on the thermal field has also been devised. Moreover, a parametric study has been carried out to understand the effect of varying beam size and laser travel speed on heat and fluid flow conditions along with the final microstructures. The results show that changing the beam size or travel speed highly influences the grain sizes, dendritic growth directions and also the grain morphologies. To study the metallurgical conditions of the process, a microstructural sub-model has been developed. It is shown that by choosing different process parameters, one can manipulate the direction of the dendritic growth and change the grain sizes. Specifically, it is found that the overall effect of changing beam size on grain morphology is less pronounced than changing the travelling speed.

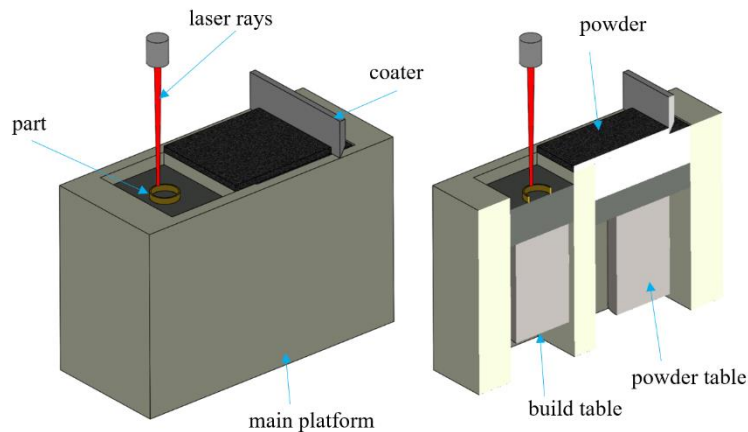
7 **Keywords:** Conical heat source, L-PBF process, heat and fluid flow, liquid metal, microstructure.

8 **1. Introduction**

9 In Metal Additive Manufacturing (MAM), parts are manufactured in a sequential layer by layer  
10 manner. This technology is widely used in various industries such as medical, aerospace,  
11 automotive, etc [1], largely due to its unique capability of producing complex parts within an  
12 acceptable timespan and with low material waste (unlike subtractive manufacturing processes). In  
13 L-PBF which belongs to the group of MAM technologies [2], a laser with a very tiny spot size  
14 (below 200  $\mu\text{m}$  diameter [3]) is used as the heating source to melt down the powder particles and  
15 subsequently fuse them together.

16 A schematic view of a typical L-PBF machine is shown in Figure 1. In the L-PBF process, first  
17 the coating mechanism spreads a layer of fine metallic spheres (whose sizes are in the range of 20-  
18 50  $\mu\text{m}$  typically [3]) on the building table. Then the laser starts to scan the predefined locations  
19 based on the input CAD file [4]–[6]. In this way the powder particles get fused together and at the  
20 end of the scanning step, a coherent layer of the part will be formed. Then after a cooling time of

21 1-2 seconds, the build table moves one increment down while the powder table moves an increment  
22 up (each increment is roughly equal to the thickness of a powder layer). Then the coating and  
23 scanning steps start sequentially and this cycle continues until the whole part is manufactured.



24  
25 Figure 1. Schematic 3D view of a typical L-PBF machine along with the components. On the right side see the cross-section.

26 The L-PBF process is very fast and typically the laser scans parts with sizes of 1-30 mm, with a  
27 significant speed of  $50\text{-}3000\text{ mm}\cdot\text{s}^{-1}$  [3]. However, there still exists a lot of uncertainties about the  
28 morphology and microstructural patterns of the produced parts, because of the unsteady nature of  
29 the process. Experimental investigation of a large number of samples made with different process  
30 parameters would typically be the straightforward and conventional way of understanding the  
31 effects of these parameters on the characteristics of the parts produced [7]–[11]. Melt pool size  
32 and its geometry, location of hotspot zones, grain morphologies and void positions, would  
33 typically be the characteristics of highest importance.

34 There is, however, an elegant alternative way to investigate the impact of the process parameters  
35 on the mentioned part characteristics, and that is via a calibrated numerical model [12]. Such  
36 calibrated and validated model, can be implemented as a cheap, reliable and powerful tool for  
37 studying the thermal behaviour, grain morphologies and fluid dynamics inside the melt pool during  
38 the L-PBF, laser welding or any other similar laser-based process [13]–[20]. In the recent years, a  
39 substantial amount of research has been carried out in the modelling of MAM including the L-PBF  
40 process, spanning from thermal models to microstructural models, mechanical models and  
41 complex computational fluid dynamics (CFD) models [21]–[30].

42 Conduction heat transfer models, mainly developed in the Finite Element (FE) framework (one  
43 case with finite volume method [31]), have been widely used so far for simulation of the thermal  
44 behavior of the L-PBF process [24]–[29]. Criales et al. [24] performed a comprehensive sensitivity  
45 analysis on the effects of both material properties and process parameters on the thermal conditions  
46 during the L-PBF process. They developed a 2D heat conduction model for this purpose and  
47 showed that the powder packing's density and its reflectivity have the highest influence on the  
48 peak temperatures formed during the process [24]. Huang et al. [25] also developed an FEM-based  
49 thermal model for the Ti6Al4V alloy and studied the effects of laser input power and its scanning  
50 speed on the shape and size of the melt pool for a single track L-PBF process. Liu et al. [30] studied  
51 the effects of thermal cycling during the course of a single-track multi-layer process and found

52 that the size of the melt pool increases with the number of layers, mainly due to accumulation of  
53 heat from previous layers. In a recent work, Loh et al. [26] included the volume shrinkage in their  
54 heat conduction model and showed that neglecting this effect will result in a small overestimation  
55 of peak temperatures. Also some work has been dedicated to coupled thermo-metallurgical models  
56 of MAM. Bontha et al. [32] studied the effect of varying laser input power and its scanning speed  
57 on the grain morphology of the samples produced by means of MAM. They implemented an  
58 analytical moving point source (based on the classical Rosenthal's thick plate solution) to model  
59 the thermal effects during the process. Raghavan et al. [33] and Nie et al. [34] have separately  
60 studied the effects of process parameters on the grain morphology of Nickel-based alloys in  
61 Electron Beam Melting (EBM) with each their conduction-based model.

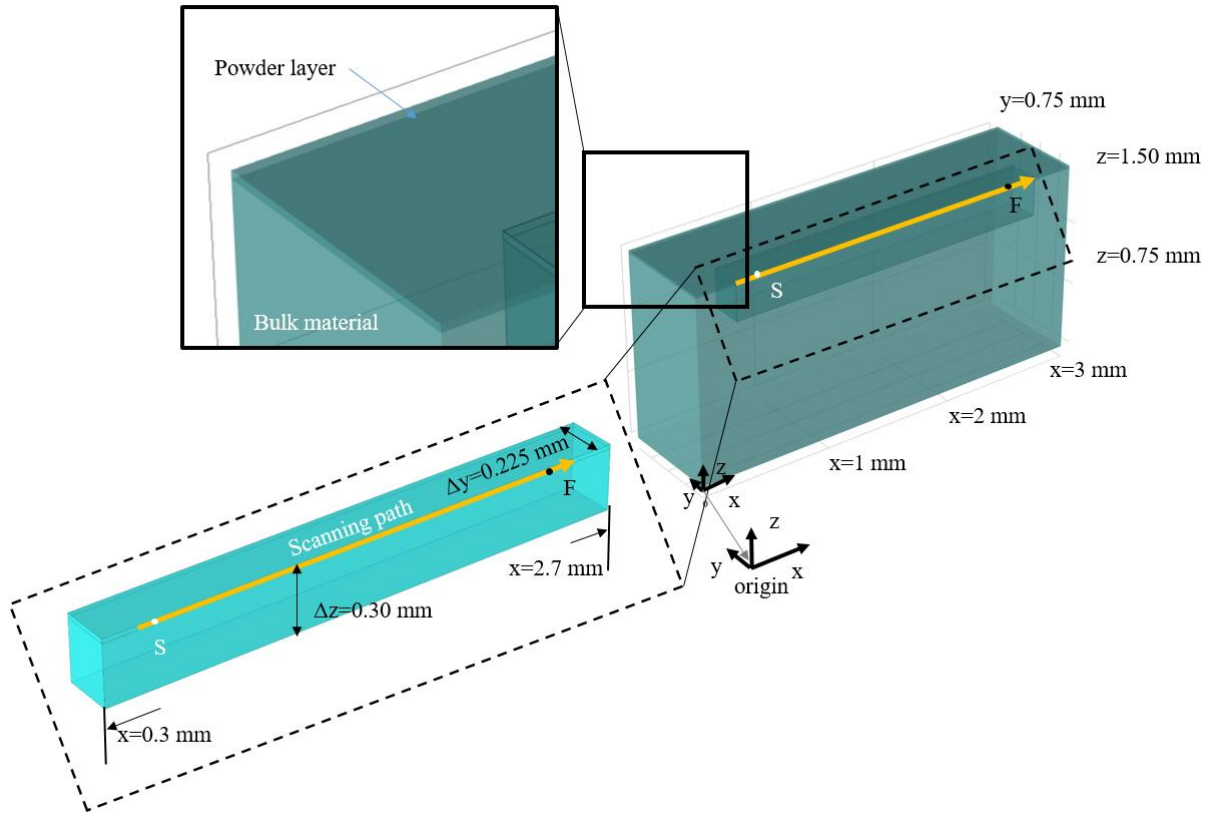
62 All of the aforementioned contributions use a thermal model based on heat conduction only and  
63 without any strong argument, apart from simplicity and computational efficiency, have excluded  
64 the fluid flow from their simulations [24]–[30], [32]–[34]. However, in the recent years a number  
65 of researchers have developed multiphysics models by taking the fluid motion into account [35]–  
66 [38] both for MAM and welding. For example, Leitz et al. [39] have developed a multiphysics  
67 numerical model based on the FEM and included the fluid motion in their calculations. In another  
68 recent work, Lee and Zhang [40] developed a multiphysics numerical model in the Finite Volume  
69 Method (FVM) framework and considered the deformation of the exposed surface of the metal as  
70 well. The mentioned thermo-fluid models mostly focused on just one single set of process  
71 parameters and have not as such been used for studying the metallurgical characteristics of the  
72 samples [22], [35]–[38], except for [40] and [41] where the authors showed that the grain  
73 morphology would remain columnar for their specific L-PBF process involving the Inconel 718  
74 alloy and Ti6Al4V, respectively.

75 In this work, a thermo-fluid-metallurgical model based on the FEM framework has been developed  
76 in COMSOL Multiphysics for the single track L-PBF process of Ti6Al4V. The model includes all  
77 modes of heat transfer, namely conduction, convection and radiation. To model the fluid flow  
78 during the solidification, solidification drag forces have been inserted as volumetric forces into the  
79 Navier-Stokes equations. The thermo-capillary effect has been taken into account as well.  
80 Furthermore, a novel moving volumetric heat source based on the concept of optical penetration  
81 depth has been introduced and by just adjusting one parameter associated with the shape of the  
82 heat source, the model can be easily calibrated. The results of the current model have both been  
83 numerically verified and experimentally validated. A thorough analysis on the role of the different  
84 driving forces on the fluid flow and the mode of heat transfer on the temperature fields has been  
85 carried out by means of dimensionless numbers. Moreover, a parametric study has been performed  
86 to analyze the impact of varying the laser beam radius and its travelling speed on melt pool size  
87 and geometry, fluid dynamics, grain morphology, solidification patterns and dendritic growth  
88 directions. For this, a microstructural sub-model has been developed and coupled to the thermo-  
89 fluid model.

## 90 **2. Finite element model**

91 The developed numerical model is based on the FEM framework and has been implemented in  
92 COMSOL Multiphysics 5.3a. The domain is considered to be a rectangular parallelepiped with 3

93 mm length, 0.75 mm width and 1.5 mm height, see Figure 2. The model is meshed with tetrahedral  
 94 elements and as shown in Figure 2, the laser starts scanning the powder layer from the point “S”  
 95 all the way to the point “F”. The powder layer with thickness  $\chi$  (seen on figure 3) is situated right  
 96 above the bulk material and has its top boundary exposed to the surroundings. Since the area close  
 97 to the laser path experiences extremely large spatio-temporal changes in both temperature and fluid  
 98 flow, it is meshed with a much finer mesh.



99  
 100 Figure 2. Model geometry along with the part dimensions. The scanning line starts from point “S” all the way to point “F” along  
 101 the scanning path arrow shown in orange color. On the top of the bulk material the powder layer is set. The parallelepiped shown  
 102 in cyan blue is the zone which is meshed with very fine mesh elements. The symmetry plane is the  $y=0$  plane where the symmetry  
 103 boundary condition is set as well.

104 Due to the symmetrical nature of the problem shown in Figure 2, a symmetry boundary condition  
 105 is used on the  $y=0$  plane for both thermal and fluid dynamics models. The main assumptions made  
 106 are listed below

- 107 • The flow is assumed to be incompressible.
- 108 • The fluid is Newtonian and the flow is laminar regime.
- 109 • The powder layer is modelled as a continuum domain with effective thermo-physical  
 110 properties.
- 111 • The free surface of the fluid is assumed to be flat.
- 112 • Mass loss due to evaporation is ignored.

113 • Mass-averaged thermo-physical properties are used.

## 114 2.1. Heat transfer model

115 The transient temperature distribution over the computational domain can be found by solving the  
116 general energy equation accounting for both conduction and convection [42], [43].

$$\begin{aligned} \frac{\partial}{\partial t}(\rho C_p T) + \frac{\partial}{\partial x_j}(\rho C_p T u_j) \\ = \frac{\partial}{\partial x_j} \left[ k \left( \frac{\partial T}{\partial x_j} \right) \right] - \rho \Delta H_{fl} \left[ \frac{\partial}{\partial t} (f_l) + \frac{\partial}{\partial x_j} (f_l u_j) \right] + \dot{Q}_v''' \end{aligned} \quad (1)$$

117 The velocity vector is denoted  $u_i$  ( $\text{m.s}^{-1}$ ) in equation (1) while  $C_p$  ( $\text{J.kg}^{-1}.\text{K}^{-1}$ ) and  $\rho$  ( $\text{kg.m}^{-3}$ ) are  
118 specific heat capacity and density of the metal, respectively.  $k$  ( $\text{W.m}^{-1}.\text{K}^{-1}$ ) and  $\Delta H_{fl}$  ( $\text{J.kg}^{-1}$ ) are  
119 thermal conductivity and latent heat of fusion and  $\dot{Q}_v'''$  ( $\text{W.m}^{-3}$ ) is the volumetric heat source  
120 caused by the laser irradiation.  $f_l$  is the fraction of the liquid phase which for simplicity is assumed  
121 to be a linear function of temperature.

$$f_l = \begin{cases} 1 & T \geq T_l \\ (T - T_{sol}) / (T_{liq} - T_{sol}) & T_s < T < T_l \\ 0 & T \leq T_s \end{cases} \quad (2)$$

122 where  $T_s$  and  $T_l$  respectively stand for solidus and liquidus temperatures. The required thermo-  
123 physical properties of Ti6Al4V are given in Table 1.

124 Table 1. Thermo-physical properties of Ti6Al4V for the CFD calculations [25], [38], [32].

Property	Symbol	Value	Unit
Solidus temperature	$T_{sol}$	1893.2	(K)
Liquidus temperature	$T_{liq}$	1927.2	(K)
Solid specific heat capacity	$C_{ps}$	543	( $\text{J.kg}^{-1}.\text{K}^{-1}$ )
Liquid specific heat capacity	$C_{pl}$	750	( $\text{J.kg}^{-1}.\text{K}^{-1}$ )
Viscosity at melting point	$\mu$	0.005	(Pa.s)
Temperature dependency of surface tension	$\gamma$	-0.0002	( $\text{N.m}^{-1}.\text{K}^{-1}$ )
Solid thermal conductivity	$k_s$	13	( $\text{W.m}^{-1}.\text{K}^{-1}$ )
Liquid thermal conductivity	$k_l$	33	( $\text{W.m}^{-1}.\text{K}^{-1}$ )
Latent heat of fusion	$\Delta H_{fl}$	280000	( $\text{J.kg}^{-1}$ )
Laser absorption coefficient	$\alpha$	0.3	(-)

125 In this work mass-averaged material properties have been used for the calculations [21]. For the  
 126 bulk material, the effective values of density, specific heat capacity and thermal conductivity can  
 127 be determined with a simple mass-averaging between liquid and solid properties as shown in  
 128 equations (3)-(5):

$$\rho_{bulk} = f_l \rho_l + (1 - f_l) \rho_s, \quad (3)$$

$$C_{P,bulk} = \frac{1}{\rho_{bulk}} (f_l \rho_l C_{Pl} + (1 - f_l) \rho_s C_{Ps}), \quad (4)$$

$$k_{bulk} = f_l k_l + (1 - f_l) k_s. \quad (5)$$

129 Similarly the effective thermal properties of the powder layer can be found by mass averaging of  
 130 bulk metal and air properties, as well:

$$\rho_{powder} = \phi \rho_{air} + (1 - \phi) \rho_{bulk}, \quad (6)$$

$$C_{P,powder} = \frac{1}{\rho_{powder}} (\phi \rho_{air} C_{P,air} + (1 - \phi) \rho_{bulk} C_{P,bulk}), \quad (7)$$

$$k_{powder} = (1 - \phi)^2 k_{bulk}, \quad (8)$$

131 where  $\phi$  in equations (6)-(8) is the initial packing porosity of the powder layer and in this study is  
 132 assumed to be 0.4 [24], [25]. The subscript ( )<sub>air</sub> stands for air properties in the mentioned equations.

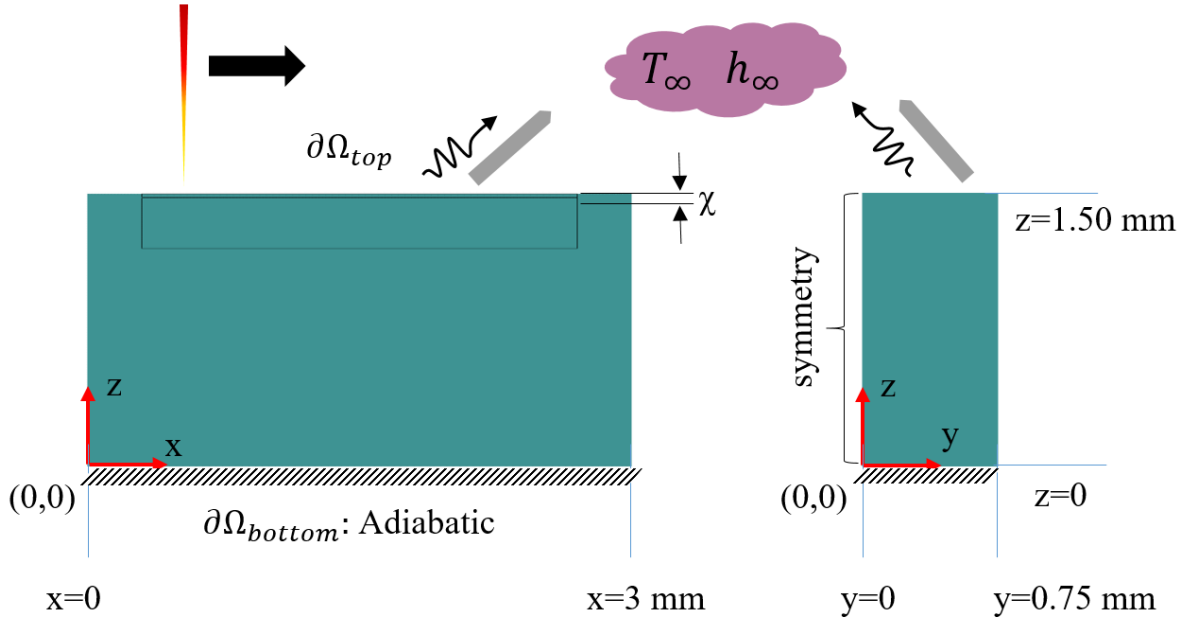
### 133 Thermal boundary conditions

134 The boundary conditions required for the thermal calculations are shown in Figure 3. According  
 135 to this figure, the top boundary is subjected to radiation and convection via the ambient, i.e.:

$$-k \frac{\partial T}{\partial z} = h_{\infty} (T - T_{\infty}) + \varepsilon \sigma (T^4 - T_{\infty}^4) \quad , \quad z = 1.5 \text{ mm or } \partial\Omega_{top} \quad (9)$$

136 The  $h_{\infty}$  (W.K<sup>-1</sup>.m<sup>-2</sup>) and  $T_{\infty}$  are the ambient convection heat transfer coefficient (found for the case  
 137 of natural convection from a hot lower surface [44]) and the surrounding temperature respectively.  
 138  $\varepsilon$  and  $\sigma$  (W.m<sup>-2</sup>.K<sup>-4</sup>) are surface emissivity and the Stephan-Boltzmann constant. As shown in  
 139 Figure 3, the bottom boundary condition is set to be adiabatic in order to represent the very low  
 140 thermal gradients at this distant boundary as compared to where the laser affects the material, i.e.:

$$-k \frac{\partial T}{\partial z} = 0 \quad , \quad z = 0.0 \text{ or } \partial\Omega_{bottom} \quad (10)$$



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Figure 3. Thermal boundary conditions for the FE model. The bottom boundary is adiabatic and the top boundary transfers heat via convection and radiation towards the ambient. The left plane at  $y=0$  on the  $yz$  plane shown on the right side of the figure is a symmetry boundary condition. The thickness of the powder layer on the top is denoted  $\chi$ .

### 145 Initial condition

146

147

The initial temperature of both the powder layer and the bulk material is set to be 300 K with a uniform distribution.

$$T(x, y, z, t) = 300 \quad , \quad t = 0.0 \quad \& \quad (x, y, z) \in \Omega_{\text{bulk}} \cup \Omega_{\text{powder}} \quad (11)$$

148

149

$\Omega$  stands for the computational domain whereas  $\partial\Omega$  denotes the corresponding boundaries of that domain.

### 150 2.2. Conical equivalent heat source

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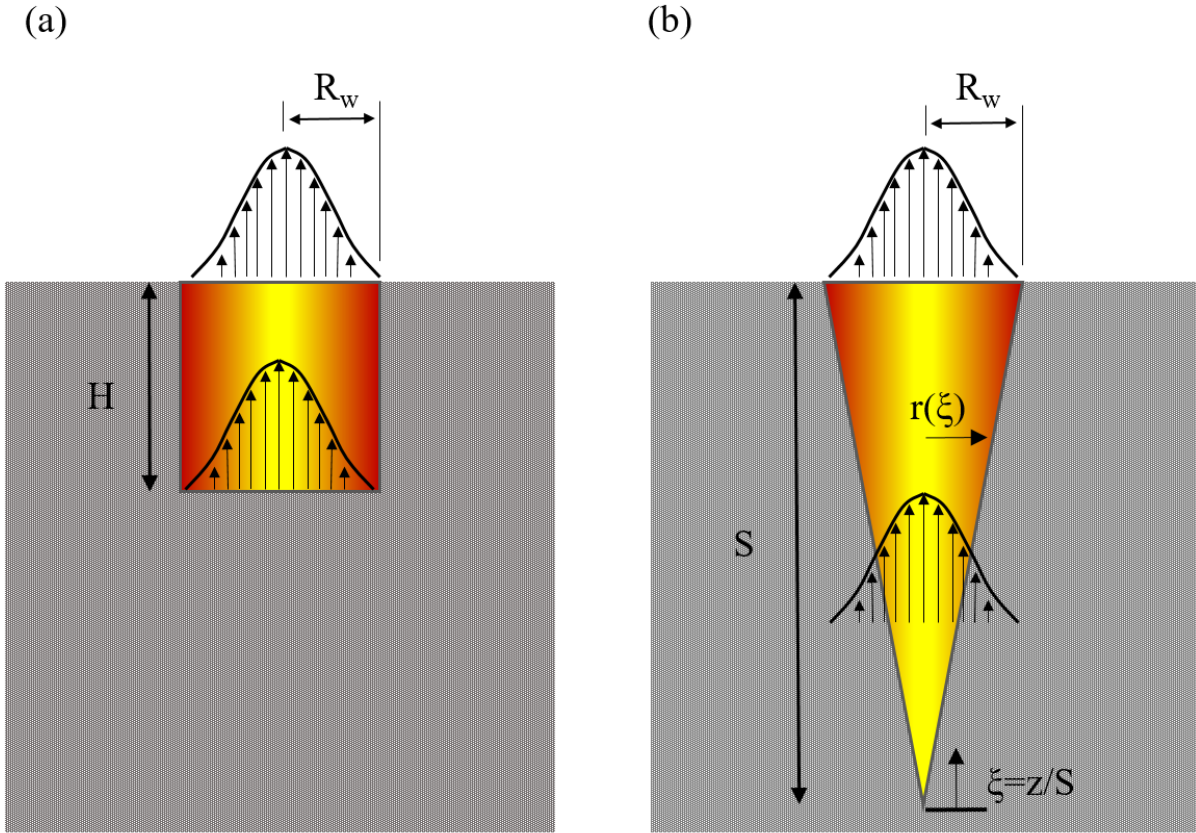
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163

In order to model the laser effect on the material, several options exist for MAM and welding simulations, such as moving point sources [32], moving surface heat flux [45], moving volumetric heat source [46], the ray-tracing method [40], etc. In this work a volumetric heat source with a conico-Gaussian distribution is introduced which is based on the concept of Optical Penetration Depth (OPD) [47]. The OPD concept is based on the fact that the laser rays in laser-based MAM can infiltrate to a certain depth into the powder layer, this way resulting in a heat generation in the domain up to a certain depth from the top surface [47]. The original OPD method was introduced for pure conduction problems and assumes a Gaussian spatial (in  $x$ - $y$  plane) distribution with a uniform vertical (along  $z$ ) distribution for a finite depth which is defined as the OPD. A schematic view of the aforementioned cylindrical-OPD heat source is shown in Figure 4 (a). Although the cylindrical OPD might give satisfactory results for pure conduction problems, in CFD models with inclusion of fluid flow, it will lead to large and unrealistic width to depth ratios for the melt pools [48]–[50]. Hence, in the present work, a modified version of the OPD heat source is developed



164 which has a Gaussian planar (x-y) and a linear vertical distribution (along the depth) as shown  
 165 schematically in Figure 4 (b).



166  
 167 Figure 4. The schematic view of (a) cylindrical OPD and (b) conical OPD. Note that the planar distribution for both cases remains  
 168 Gaussian and for the case (b) the heat is just generated until the fringes of the heat source which is denoted by  $r(\xi)$ . Both (a) and  
 169 (b) release the same amount of heat as the input energy  $\alpha P_w$ .

170 The necessary requirement for this volumetric heat source is that it should release the same amount  
 171 of energy as the laser input energy, which is  $\alpha P_w$ , i.e.:

$$\iiint_{V_h} \dot{Q}_v''' dV = \alpha P_w, \quad (12)$$

172 and the Gaussian heat flux is given by

$$\dot{q}_{laser}'' = \frac{2\alpha P_w}{\pi R_w^2} e^{-\frac{2(x^2+y^2)}{R_w^2}}, \quad (13)$$

173 where  $q_{laser}$  ( $\text{W}\cdot\text{m}^{-2}$ ) in equation (13) is the Gaussian heat flux and  $R_w$  (m) is the distance from the  
 174 center of the heat source of which the heat flux reaches  $e^{-2}$  of its peak value. Now, the cylindrical-  
 175 OPD heat source is simply found by dividing the Gaussian heat flux by the OPD depth which  
 176 according to Figure 4 (a) is denoted  $H$  (m).

$$\dot{Q}_{cyl}''' = \frac{\dot{q}_{laser}''}{H} = \frac{2\alpha P_w}{H\pi R_w^2} e^{-\frac{2(x^2+y^2)}{R_w^2}}. \quad (14)$$

177 The subscript ( )<sub>cyl</sub> stands for cylindrical in equation (14). The conical-OPD heat source is defined  
 178 by multiplying the cylindrical-OPD heat source introduced in equation (14) by a dimensionless  
 179 vertical distribution function  $\xi$  (-) in which the  $\xi$  is a dimensionless coordinate from the bottom of  
 180 the conical heat source towards the top plane, as shown in Figure 4 (b).

$$\xi = z/S \quad (15)$$

$$\dot{Q}_{cone}''' = \frac{2\alpha P_w}{H\pi R_w^2} e^{-\frac{2(x^2+y^2)}{R_w^2}} \cdot \xi \quad (16)$$

181  $z$  (m) in equation (15) is assumed to be zero at the bottom of the heat source and  $S$  at its top and is  
 182 considered as a relative coordinate, which will be used later on for integration. Now in order to  
 183 find the relation between the cylindrical and conical OPD depths, respectively denoted by  $H$  and  
 184  $S$ , one must set the volume integral of equation (16) equal  $\alpha P_w$ .

$$\iiint_{V_{cone}} \dot{Q}_{cone}''' dV = \alpha P_w, \quad (17)$$

185 where  $V_{cone}$  stands for the domain inside the conical OPD. Now we set the bounds of the integral  
 186 in equation (17), i.e.:

$$\iiint_{V_{cone}} \dot{Q}_{cone}''' dV = \int_0^{2\pi r(\xi)} \int_0^S \int_0^S \dot{Q}_{cone}''' dz \cdot r dr \cdot d\theta, \quad (18)$$

187  $r(\xi)$  is schematically shown in Figure 4 (b) and varies from 0 to  $R_w$  for  $z$  in the interval of 0 to  $S$ .  
 188 By introducing  $r(\xi)$  into equation (18) the bounds of the integral are defined

$$\iiint_{V_{cone}} \dot{Q}_{cone}''' dV = \int_0^{2\pi \frac{z \cdot R_w}{S}} \int_0^S \int_0^S \dot{Q}_{cone}''' dz \cdot r dr \cdot d\theta. \quad (19)$$

189 And now by integration over  $\theta$ ,  $r$  and  $z$ , the net power produced with the conical-OPD heat source,  
 190 which will be a function of  $H$  and  $S$ , can be determined

$$2\pi \int_0^{\frac{z \cdot R_w}{S}} \int_0^S \left[ \frac{2\alpha P_w}{H\pi R_w^2} e^{-\frac{2(r^2)}{R_w^2}} \cdot \xi \right] dz \cdot r dr = \alpha P_w. \quad (20)$$

191 By integrating equation (20) and substituting the bounds of the integral, we obtain

$$\frac{\alpha \cdot S \cdot P_w}{H} \left[ \frac{\xi^2}{2} + \frac{1}{4} e^{-2\xi^2} \right]_0^1 = \alpha P_w, \quad (21)$$

192 and by inserting the upper and lower bounds, the following expression is obtained

$$0.283834 \frac{\alpha \cdot S \cdot P_w}{H} = \alpha \cdot P_w, \quad (22)$$

193 from which the following relation between  $S$  and  $H$  is determined:

$$S \cong 3.52 H. \quad (23)$$

194 Having obtained this relationship between  $S$  and  $H$ , it is sufficient to find one of them in order to  
 195 adjust the shape of the predicted melt pool profile to that of the experiments. In this study, the  $S$   
 196 value is changed and used as an independent variable for finding and calibrating the shape of the  
 197 heat source.

### 198 2.3. Computational Fluid Dynamics (CFD)

199 As the laser heats up the powder layer and subsequently the underneath bulk material beyond their  
 200 melting points, the material becomes liquid and hence highly deformable. In the presence of the  
 201 concentrated heat source, extremely high temperature gradients will also form which will  
 202 consequently lead to thermally-induced shear stresses that cause the liquid to flow and circulate  
 203 within the melt pool.

204 To find the velocity field inside the melt pool it is necessary to solve the continuity and momentum  
 205 equations, respectively, see e.g. [51].

$$\frac{\partial(\rho u_i)}{\partial x_i} = 0 \quad (24)$$

$$\begin{aligned} \frac{\partial}{\partial t}(\rho u_i) + \frac{\partial}{\partial x_j}(\rho u_i u_j) \\ = -\frac{\partial p}{\partial x_i} + \frac{\partial}{\partial x_j} \left[ \mu \left( \frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} - \frac{2}{3} \delta_{ij} \frac{\partial u_k}{\partial x_k} \right) \right] - \frac{K_C(1-f_l)^2}{C_K + f_l^3} \cdot u_i \\ - \rho g_i \beta (T - T_l) \end{aligned} \quad (25)$$

206 The divergence of the velocity field on the right hand side of equation (25) vanishes because of  
 207 the incompressibility of the flow shown in equation (24). The derivative on the left hand side of  
 208 the linear momentum balance equation is the total derivative in a Eulerian framework and the  
 209 velocity field components are described by the vector  $u_i$  (m.s<sup>-1</sup>).

210 The third term in equation (25) is the solidification drag force where the terms  $c$  (kg.m<sup>-3</sup>.s<sup>-1</sup>) and  $B$   
 211 are Carman-Kozeny constants, which are numerically big and very small, respectively [52]–[55].  
 212 Based on equation (25), when the liquid fraction goes to zero and the material solidifies, the drag  
 213 force will become a significant number. On the other hand, when the liquid fraction is one, the

214 drag force will vanish. In the current study  $c$  and  $B$  are set to  $4e5$  ( $\text{kg}\cdot\text{m}^{-3}\cdot\text{s}^{-1}$ ) and  $1e-4$  respectively.  
 215 Furthermore, the last term in equation (25) expresses the force caused by the buoyancy effect  
 216 which in this work is modelled based on the Boussinesq approximation and  $\beta$  ( $\text{K}^{-1}$ ) is the thermal  
 217 expansion coefficient in equation (25).

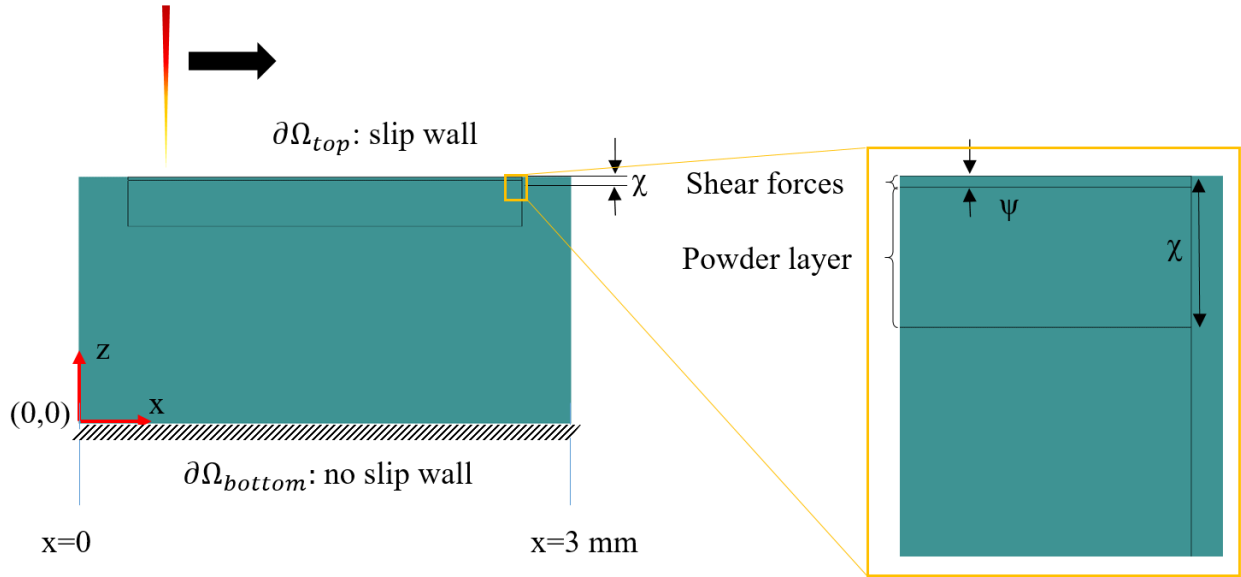
218 **CFD boundary conditions**

219 All of the boundaries are assumed to be no-slip walls except for the top boundary which is set to  
 220 a slip-wall condition, i.e.:

$$u_i = 0 \quad , \quad \partial\Omega_{sides} \cup \partial\Omega_{bottom} \quad (26)$$

$$u_i \cdot n_i \partial\Omega = 0 \quad , \quad \partial\Omega_{top} \quad (27)$$

221 Based on equation (27), only the  $z$ -direction of the velocity will become zero and the CFD  
 222 boundary conditions are shown in Figure 5.



223  
 224 Figure 5. Boundary conditions for the CFD model. The top surface is assumed to be a slip-wall while all other faces are no slip  
 225 walls. The domain in which the thermally-induced shear stresses are introduced has the depth  $\psi$  and is set on the top of the whole  
 226 domain.

227 The thermally-induced shear stresses are active as surface tractions on the top boundary and more  
 228 generally, wherever large thermal gradients prevail in the liquid, i.e.:

$$\sigma_{xz} = -\gamma \frac{\partial T}{\partial x} \quad , \quad \partial\Omega_{top} \quad (28)$$

$$\sigma_{yz} = -\gamma \frac{\partial T}{\partial y} \quad , \quad \partial\Omega_{top} \quad (29)$$

229  $\sigma_{xz}$  and  $\sigma_{yz}$  (Pa) are top surface shear stresses in the  $x$  and  $y$  directions.  $\gamma$  ( $\text{N}\cdot\text{m}^{-1}\cdot\text{K}^{-1}$ ) is the linear  
 230 dependency of the surface tension on the temperature. From a numerical point of view, imposing

231 both a slip wall condition along with shear stresses on the same boundary is unphysical and in this  
 232 regard an additional volumetric force has been introduced on a very thin subdomain (with  
 233 thickness  $\psi$ ) beneath the top boundary, in such a way that it will produce an equal amount of shear  
 234 force close to that boundary, as shown in Figure 5.

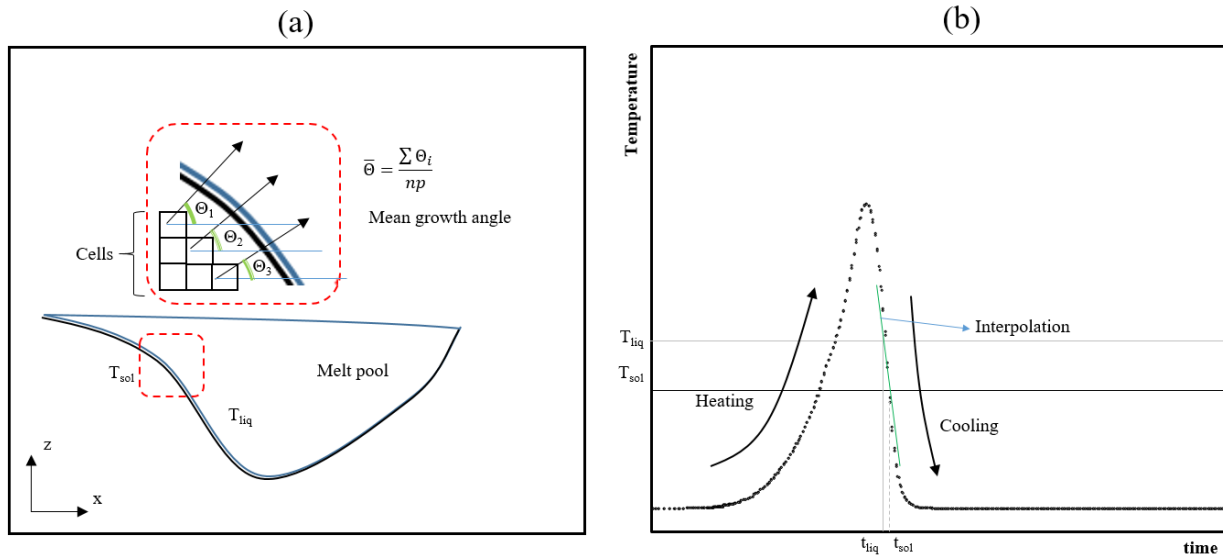
$$F_{\psi,x} = -\frac{\gamma}{\psi} \frac{\partial T}{\partial x}, \quad \Omega_{\psi} \quad (30)$$

$$F_{\psi,y} = -\frac{\gamma}{\psi} \frac{\partial T}{\partial y}, \quad \Omega_{\psi} \quad (31)$$

235  $F_{\psi,x}$  and  $F_{\psi,y}$  ( $\text{N.m}^{-3}$ ) are volumetric forces related to the thermo-capillary effect that are imposed  
 236 inside the top subdomain  $\Omega_{\psi}$ . This a standard procedure for implementing the Marangoni effect in  
 237 CFD model [17].

#### 238 2.4. Metallurgical sub-model

239 In order to study the metallurgical phenomena of the process, a metallurgical sub-model has been  
 240 developed and subsequently linked to the CFD model described earlier in sections 2.1. and 2.3..  
 241 The methodology used to derive the important metallurgical indicators, are schematically shown  
 242 in Figure 6.



243  
 244 Figure 6. Schematics of the methodologies used to find (a) the growth direction and (b) metallurgical conditions.  $np$  in (a) stands  
 245 for the number of computational cells.

246 A schematic view of a cross-section of the melt pool profile, parallel to the laser track, is shown  
 247 in Figure 6 (a). The black arrows show the temperature gradient vectors in the  $x$ - $z$  plane and as  
 248 expected, they are perpendicular to the melt pool borders. The growth direction for each  
 249 computational element is found via the following expression [40], [56]

$$\theta = \arctan\left(\frac{G_z}{G_x}\right), \quad (32)$$

250 where  $G_z$  and  $G_x$  in equation (32) are the temperature gradients in z and x directions, respectively.  
 251 According to Figure 6 (a), the mean growth direction is the average value of the cells' growth  
 252 angles. In this way, first, the growth direction for every individual cell is calculated with equation  
 253 (32), then the mean growth direction is found via an averaging on all studied cells.

254 The cooling rate can be found via [32], [33]

$$Cr = \frac{T_{liq} - T_{sol}}{t_{liq} - t_{sol}}, \quad (33)$$

255 And  $t_{liq}$  and  $t_{sol}$ , are the times for start and end of solidification. Due to very large cooling rates, a  
 256 linear interpolation is performed to find the times at which the temperature intersects the liquidus  
 257 and solidus lines, as shown in Figure 6 (b).

258 The solidification temperature gradient is found at the start of the solidification (at  $t = t_{liq}$ ) [33].  
 259 The growth velocity  $R$  ( $m.s^{-1}$ ) is, however, a dependent variable and is defined as [15], [16]

$$R = \frac{Cr}{G}. \quad (34)$$

260 There is another metallurgical condition, denoted the morphology factor which serves as an  
 261 indicator of the morphology of the grains, which is also a dependent variable e.g.

$$F = \frac{G}{R}. \quad (35)$$

262 All of the four aforementioned metallurgical conditions along with the growth direction are  
 263 calculated based on the CFD model data and have been reported in the subsequent sections.

### 264 3. Mesh sensitivity analysis and validation

265 The current model has been numerically verified by means of mesh independency analysis and  
 266 experimentally validated by comparing the predicted melt pool with those found in experiments.

#### 267 Mesh independency

268 The process parameters chosen for the mesh independency analysis are given in Table 2. Five  
 269 different cases for the size and number of elements are selected. Based on Table 2, by increasing  
 270 the number of elements to about 2 million from 800,000, the average temperature of the domain  
 271 increased 0.04 K and the melt pool size became about 1% larger. Hence, in the current study we  
 272 use the case 3 configuration for the calculations henceforth in the paper, due to its sufficient  
 273 accuracy and lower required CPU time, compared to the cases 4 and 5.

274  
275

Table 2. Process parameters for all three different cases for mesh independency analysis along with the calculated average domain temperature.

Case	Process parameter				Elements count	Mean temperature (K)	Melt pool volume ( $\times 10^{-14} \text{ m}^3$ )
	Input power	Scan speed	Beam radius	Layer thickness			
	$P_w$ (W)	$V_w$ (mm.s <sup>-1</sup> )	$R_w$ ( $\mu\text{m}$ )	$\psi$ ( $\mu\text{m}$ )			
1					385398	402.95	43.6
2					584736	403.16	50.1
3	200	800	50	20	803966	403.17	50.8
4					1151773	403.20	51.2
5					1906215	403.21	51.4

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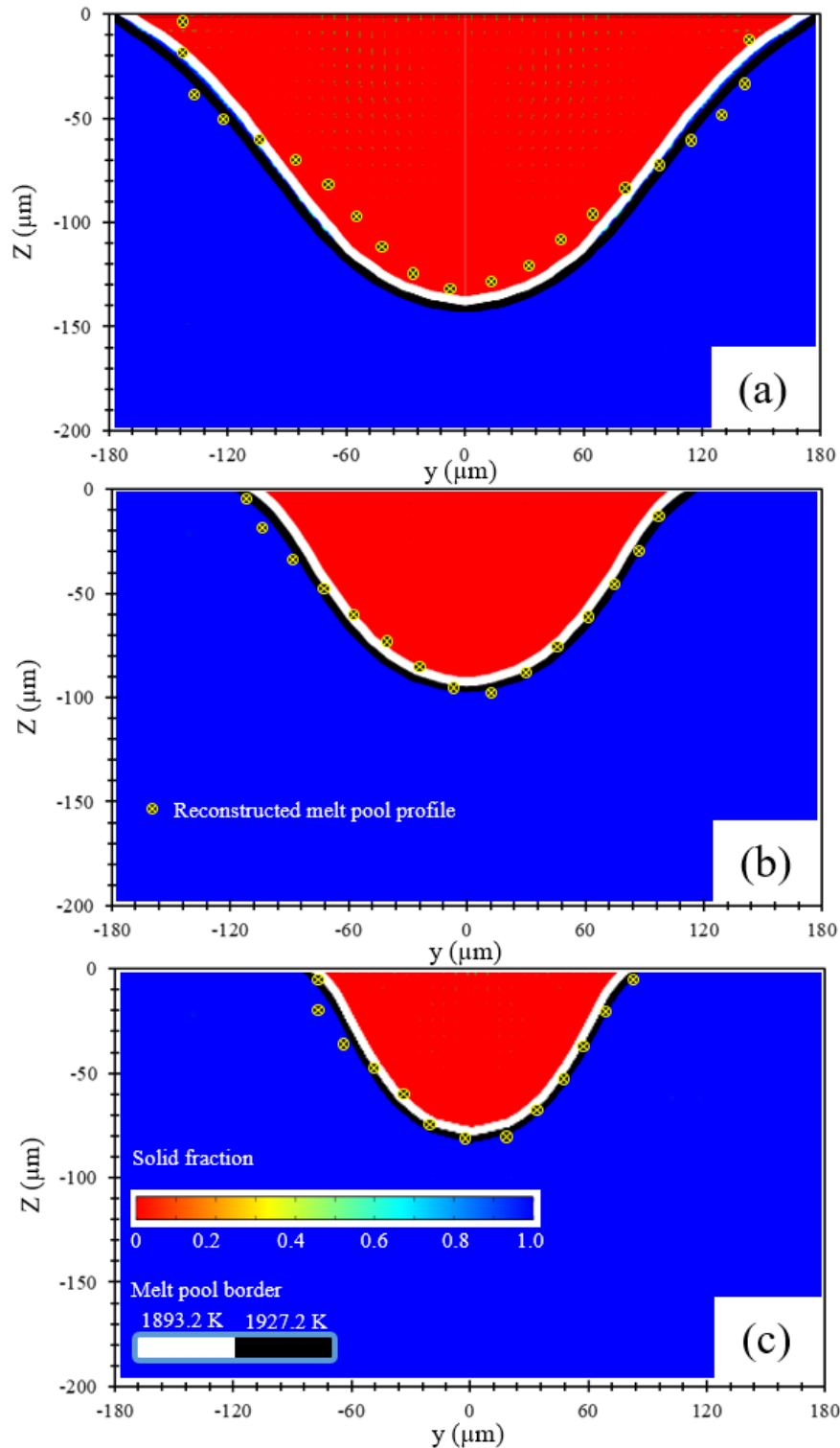
Moreover, the current model has been calibrated by varying the scanning speed and comparing the size of the predicted melt pool with the ones measured experimentally [57]. The process parameters used for this validation are given in Table 3.

Table 3. Process parameters for the validation [57]–[59].

Case	Process parameter			
	Input power	Scan speed	Beam radius	Layer thickness
	$P_w$ (W)	$V_w$ (mm.s <sup>-1</sup> )	$R_w$ ( $\mu\text{m}$ )	$\psi$ ( $\mu\text{m}$ )
A		200		
B	200	300	50	20
C		400		

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The comparison between the numerically predicted and experimentally measured molten zones for all three different cases gathered in Table 3, is shown in Figure 7.



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Figure 7. Comparison between predicted and experimentally measured shapes of the melt pool [57] for (a)  $V = 200 \text{ mm.s}^{-1}$ , (b)  $V = 300 \text{ mm.s}^{-1}$  and (c)  $V = 400 \text{ mm.s}^{-1}$ . The red color shows the molten region while the blue stands for the solid zone. White and black lines shown in the numerical contours respectively show the liquidus and solidus lines.

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The reconstructed melt pool profiles which were measured experimentally are shown with markers in Figure 7. According to Figure 7 (a), the predicted melt pool profile is slightly wider than the

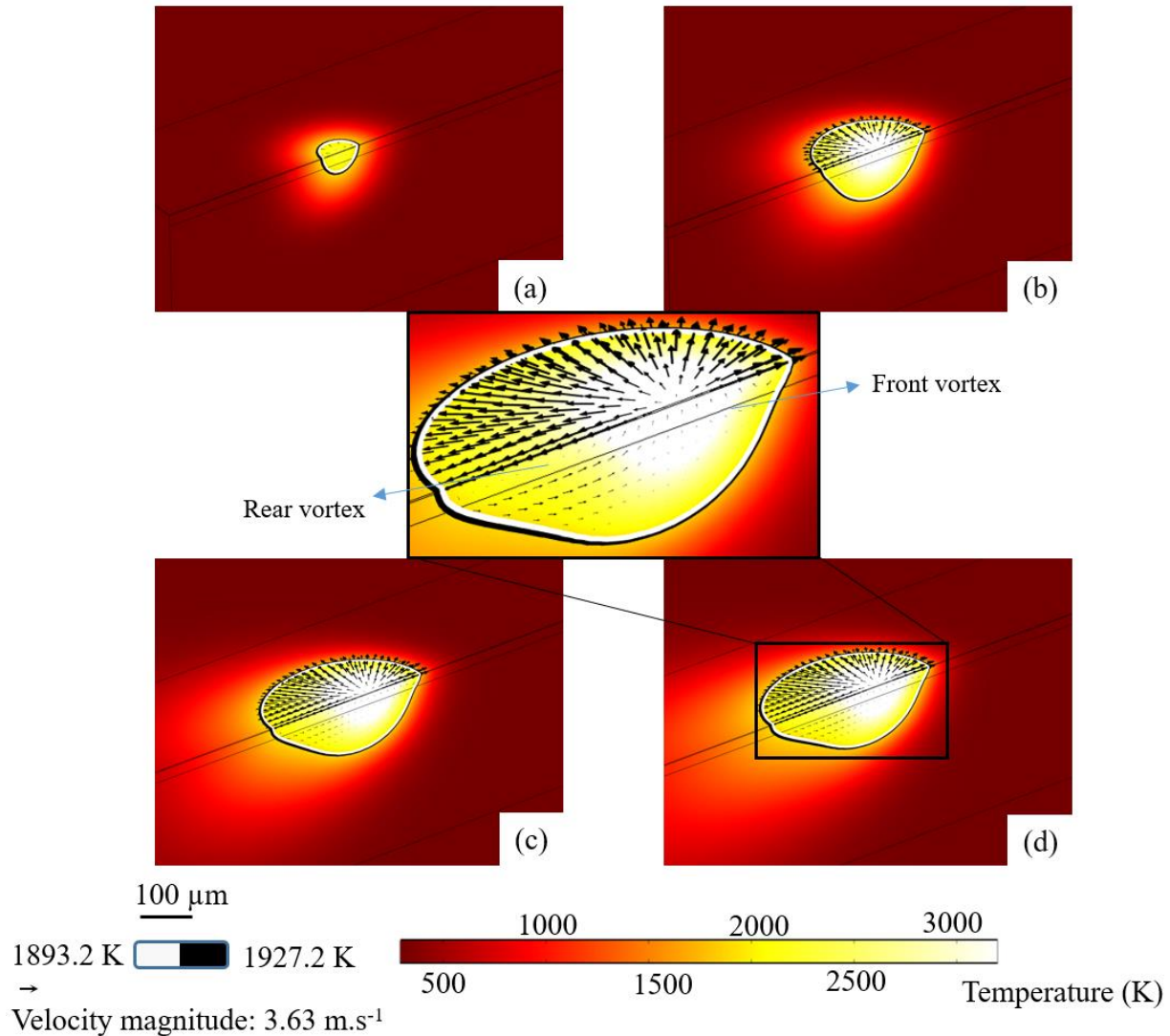


288 experimental one. The predicted depth of the melt pool is in an even better agreement with the  
289 experiments, see Figure 7 (a). When the scanning speed is increased to  $300 \text{ mm.s}^{-1}$ , the melt pool  
290 becomes smaller in both depth and width, as expected. This was also experimentally shown by  
291 Wang et al. [60] for stainless steel. For  $300 \text{ mm.s}^{-1}$ , shown in Figure 7 (b), the predicted melt pool  
292 profile is in better agreement with the experimental one. According to Figure 7 (b), the predicted  
293 width and depth of the melt pool in case B are  $240 \mu\text{m}$  and  $100 \mu\text{m}$  respectively, which matches  
294 well with the reconstructed experimental micrographs. Moreover, it is observed that further  
295 increase in travel speed leads to even smaller width and depth of the melt pool, which is due to  
296 shorter contact time between laser and the material (see Figure 7 (c)). It is also visually seen in  
297 Figure 7 that when the laser speed is increased from  $200 \text{ mm.s}^{-1}$  to  $400 \text{ mm.s}^{-1}$ , the melt pool depth  
298 is reduced from  $140 \mu\text{m}$  to less than  $90 \mu\text{m}$ . The resulting melt pool shape based on the  
299 conventional cylindrical OPD-based heat source (originally devised for pure conduction models  
300 [47]) will have an unrealistically high width to depth ratio [48]–[50], unlike the proposed conico-  
301 Gaussian heat source which can capture the shape of the melt pool more correctly.

## 302 4. Results and discussions

### 303 Thermal behavior

304 The temperature profile along with the calculated velocity field are shown for four different times  
305 in Figure 8. The corresponding laser input power, scanning speed and beam radius are respectively  
306  $200 \text{ W}$ ,  $300 \text{ mm.s}^{-1}$  and  $90 \mu\text{m}$ .



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Figure 8. Temperature contour for four different instants in time (calculated from when the laser starts moving): (a) 0.1 ms, (b) 0.4 ms, (c) 1.2 ms and (d) 2.0 ms. The melting lines are shown in black and white colors. Note the progressive transition of the shape of the melt pool from symmetrical to elongated in the x-direction.

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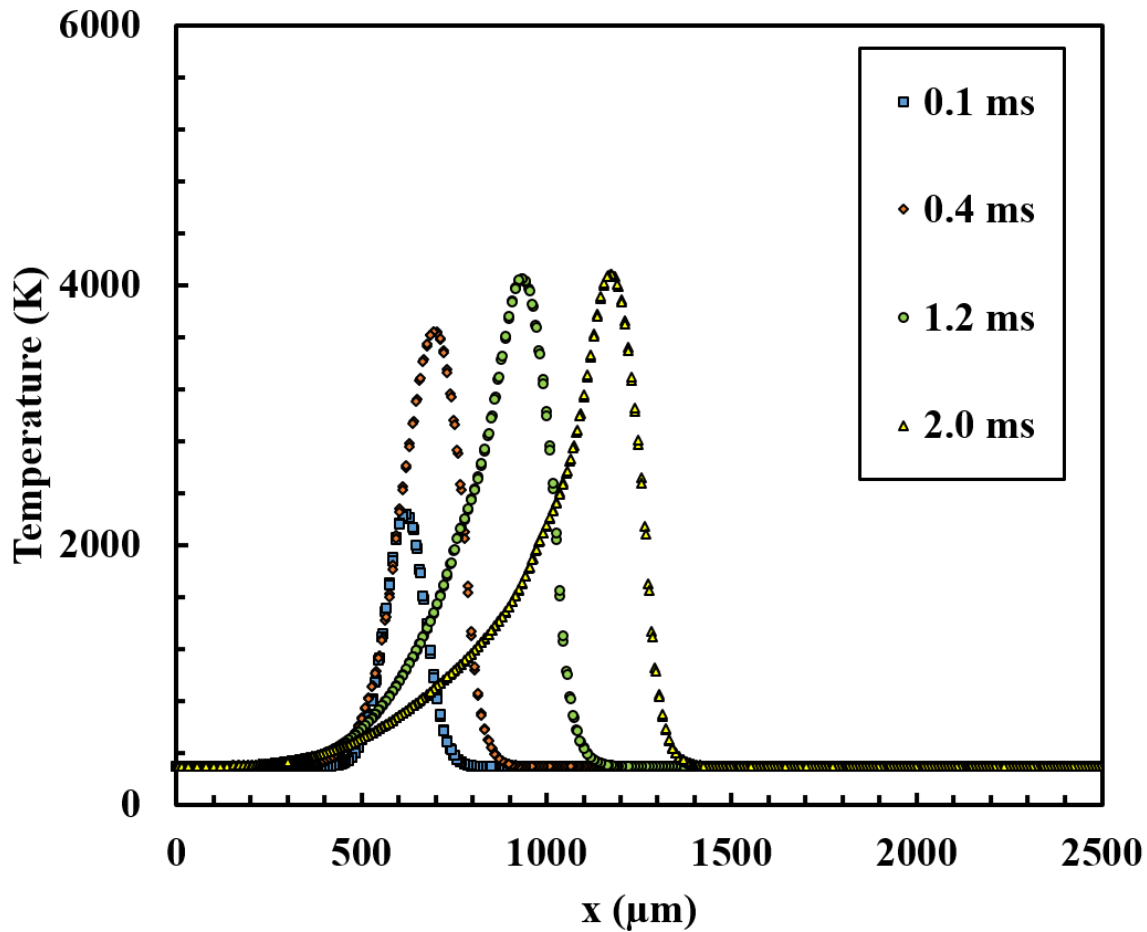
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Based on Figure 8 (a) and (b), the melt pool at the onset of the process grows equally to the sides, while becoming also deeper towards the bulk material. As time passes further, the melt pool obtains its final egg-shaped morphology, according to Figure 8 (c) and will keep it for the rest of the process. The velocity field on the top surface shown in Figure 8 is radially outward and due to incompressibility of the liquid zone, two vortices are spotted on front and back of the melt pool. The relative size of these two vortices are highly dependent on the process parameters and will be discussed later.

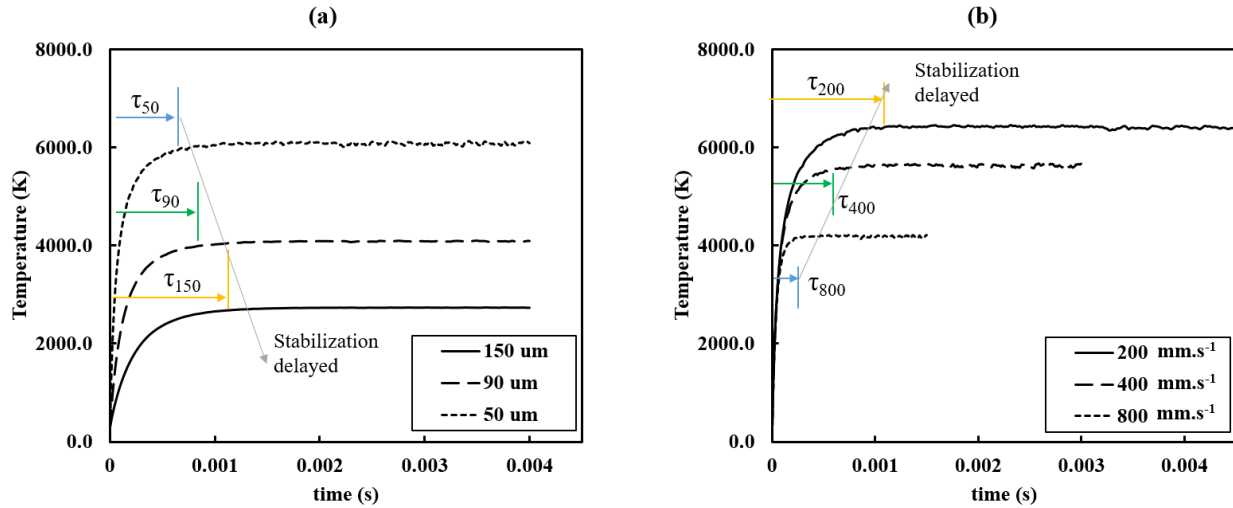


318

319 Figure 9. T-x profile measured at the scanning line at four different times. Note that the maximum temperature reaches a stable  
 320 condition from which it remains constant throughout the rest of the process.

321 What is also interesting is that, not only the shape of the melt pool will not change after 1 ms, but  
 322 also the peak temperature will remain the same after 1 ms, according to Figure 9. In other words,  
 323 the process becomes stable from this point on and this relaxation or stabilization time might vary  
 324 depending on the imposed boundary conditions or input parameters.

325 To further investigate the effect of laser speed and beam size on this stabilization time, the peak  
 326 temperature is plotted against time for different processing conditions in Figure 10.



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328

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Figure 10. Plot of peak temperature against time for (a) varying beam size ( $v = 300 \text{ mm.s}^{-1}$ ) and (b) varying scanning speeds ( $R = 50 \text{ }\mu\text{m}$ ). The laser power is set to 200 W.

330

According to Figure 10 (a) when the beam size is increased, lower peak temperatures are obtained,

331

which is because of a lower and more uniform distribution of laser heat flux over the beam area.

332

It is also noticed that choosing bigger beam sizes lead to a delay in the peak temperature profiles

333

shown in Figure 10 (a). The same trend is observed for varying scanning speeds where lower laser

334

travel speeds will cause a delay in stabilization of the process, since a bigger melt pool is formed

335

due to longer laser-material interaction, hence more time is required to reach the stable condition,

336

see Figure 10 (b),.

337

### Melt pool evolution and dimensionless analysis

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The volume of the melt pool versus time has been plotted in Figure 11 (a) where it is well observed

339

that like the peak temperature which reaches a stable condition after some time, shown in Figure

340

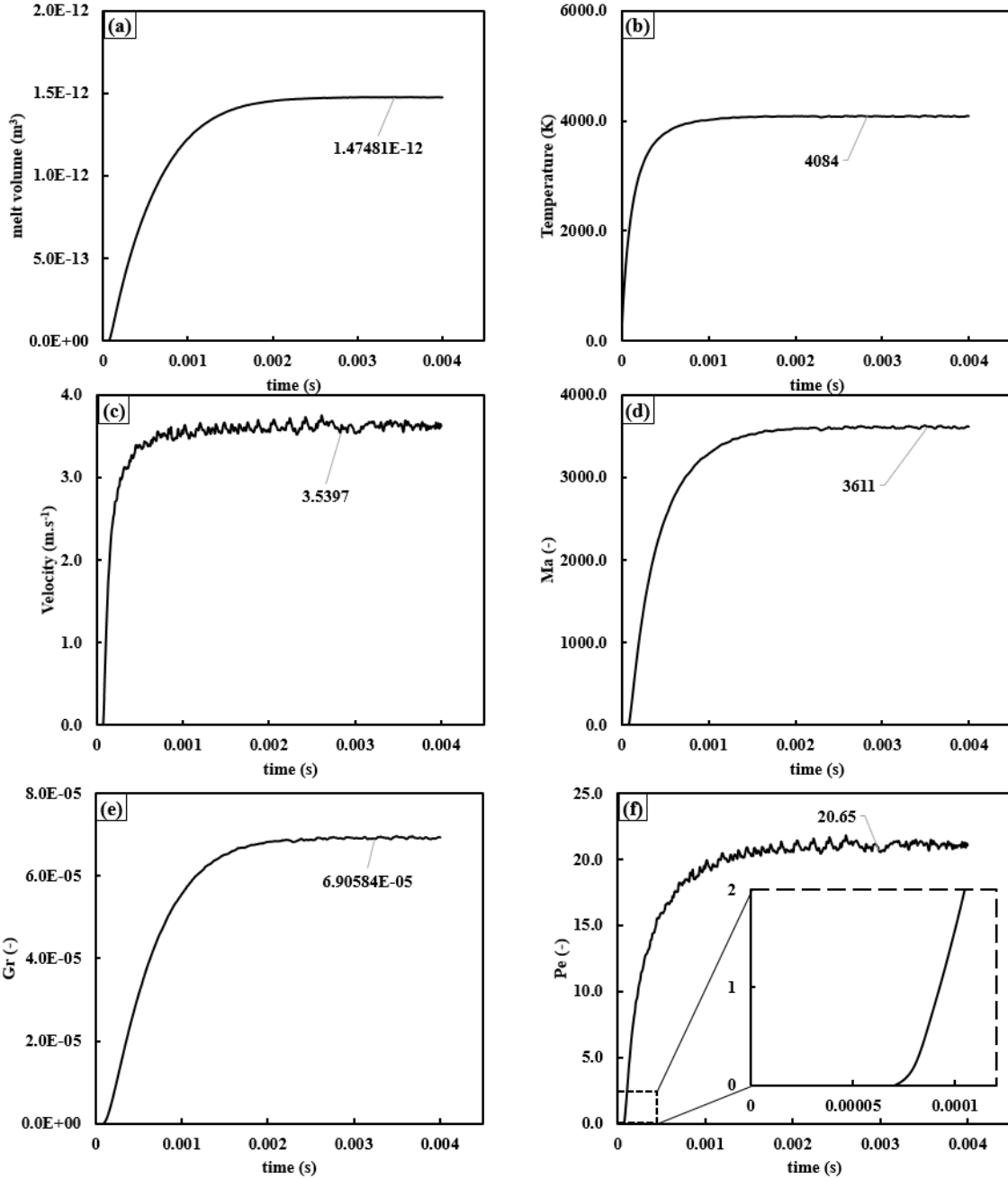
11 (b), the melt pool size will also converge to a specific value as well. The same trend is moreover

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seen in Figure 11 (c) where the maximum velocity magnitude also approaches a stable level after

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some time from the onset of the process.



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Figure 11. (a) Melt pool volume, (b) peak temperature, (c) maximum velocity magnitude, (d) Marangoni number, (e) Grashof number and (f) Peclet number versus time. The input laser power and travel speed are 200 W and 300 mm.s<sup>-1</sup>. The beam radius is set to be 90 μm.

347

348

To understand the effect of the thermally-induced shear stresses on the fluid flow, the Marangoni number is used

$$Ma = \frac{\rho L_M \Delta T_{max} |\gamma|}{\mu^2}, \quad (36)$$

349 which is a dimensionless number expressing the relative strength of thermally-induced shear  
350 stresses to the viscous stresses in a liquid.  $L_M$  (m) is the characteristic length of the melt pool which  
351 is assumed to be one-eighth of the apparent melt pool radius [20]. Apparent melt pool radius is in  
352 turn the radius of a hemisphere with the same volume as the melt pool.  $\Delta T_{max}$  (K) is the difference  
353 between the solidus and peak temperature.

354 Moreover, Grashof's number is used to study the effect of the buoyancy force on the fluid flow

$$Gr = \frac{\rho^2 \beta g L_B^3 \Delta T_{max}}{\mu^2}. \quad (37)$$

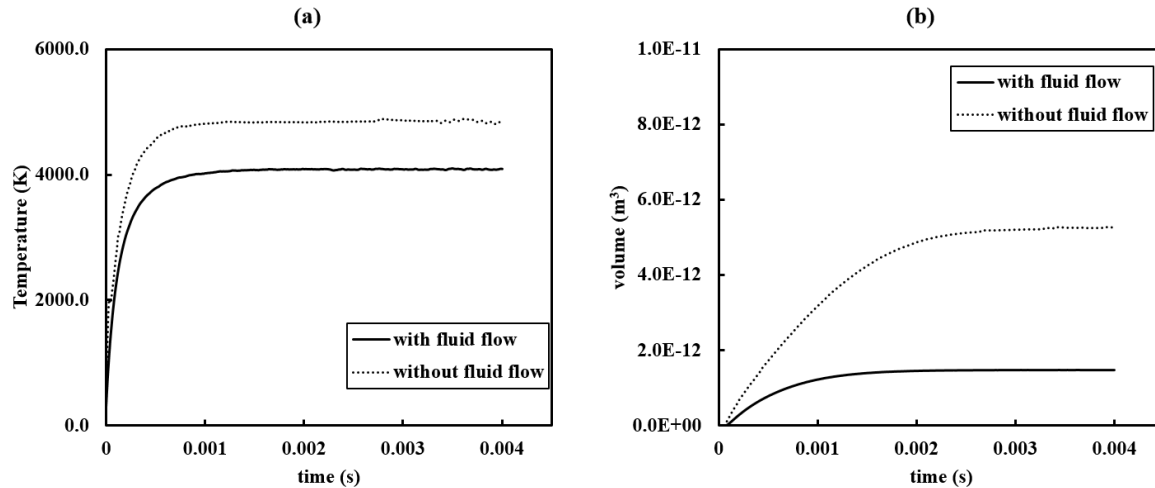
355 Grashof's number scales the relative strength of the buoyancy forces to the viscous forces. A low  
356 order of magnitude of Grashof's number ( $O(Gr) < 1$ ) means that the buoyancy forces have an  
357 insignificant effect on the fluid flow motion, while a higher order of magnitude implies the  
358 opposite.

359 According to Figure 11 (d) and (e), while the Marangoni number reaches a significant value of  
360 3611, the Grashof number converges to a negligible number of  $6.9e-5$ . This comparison reveals  
361 that the buoyancy effect has a minimal impact on the fluid dynamics inside the melt pool,  
362 compared to the viscous forces. On the other hand, a much bigger order of magnitude of the  
363 Marangoni number means that the thermally-induced shear forces outweigh the viscous forces in  
364 the melt region, meaning that the melt flow is mostly driven by the thermo-capillary effect. Finally,  
365 in order to study the dominant mode of heat transfer, the Peclet number is applied [16]

$$Pe = \frac{\rho C_p U L_M}{k}. \quad (38)$$

366 A Peclet number smaller than one indicates conduction as the dominant mode of heat transfer  
367 whereas a value greater than one, indicates convection being more dominant. According to Figure  
368 11 (f), the Peclet number, except for the very beginning of the process, will be considerably greater  
369 than one and within 1 millisecond it reaches the value of 20.65 which underlines the dominant role  
370 of fluid flow in the transfer of heat in the melt pool. As will be shown in the next section, the  
371 dominance of convection in heat transfer will highly govern melt pool size and its peak  
372 temperature, which will consequently affect the predicted metallurgical conditions as well.

373 To further understand the importance of the convective heat transfer, or in other words, the impact  
374 of neglecting the fluid flow inside the melt pool, a simplified model without the fluid flow and  
375 hence only considering conductive heat transfer was developed.



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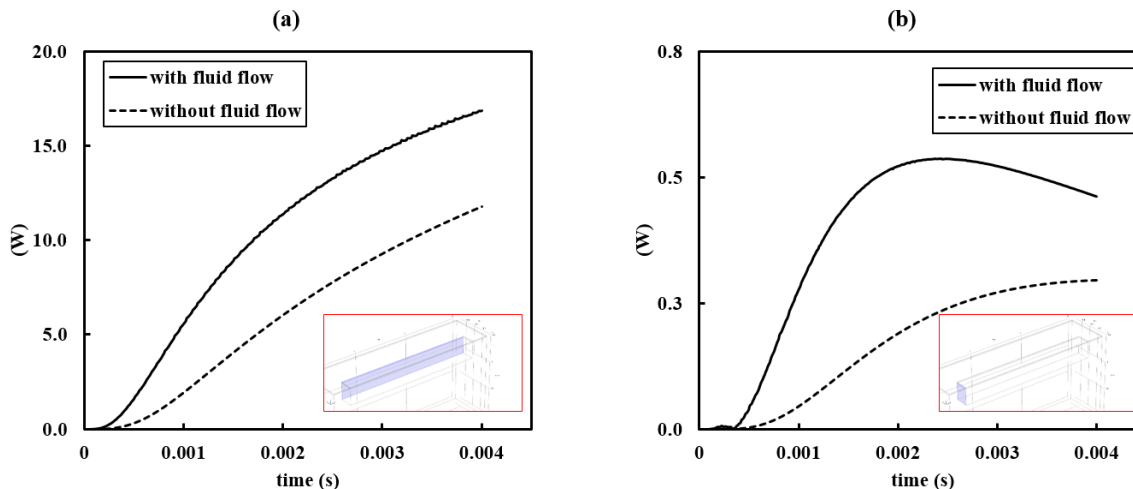
Figure 12. Plot of (a) peak temperature and (b) melt pool volume versus time. The dashed lines are the results of the pure conduction model without the fluid flow. Beam radius  $90\ \mu\text{m}$ , laser speed  $300\ \text{mm}\cdot\text{s}^{-1}$  and power set to  $200\ \text{W}$ .

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As seen in Figure 12 (a), neglecting the fluid flow will result in higher maximum temperatures and it will also dramatically affect the melt pool size and its geometry as well, see Figure 12 (b). More specifically, the numerical results show that neglecting the fluid flow inside the melt pool will lead to an almost +20% higher peak temperatures as well as a 3-4 times bigger melt region, based on Figure 12.

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The main reason that the peak temperature and also the melt pool size are lower in the case with inclusion of the fluid flow is not surprisingly that, the fluid flow will highly increase the rate of heat transfer between the melt pool borders and the colder bulk material.



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Figure 13. The net outwards power calculated on (a) a plane at  $y=0.225\ \text{mm}$  (at the left side of the melt pool) and (b) a plane at  $x=0.3\ \text{mm}$  (at the back of the melt pool), versus time. Dashed lines stand for pure conduction calculations and continuous lines belong to the thermo-fluid model. The planes of which the net output power is calculated, are shown with blue color in the red boxes.

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In order to analyze this phenomenon, the rate of heat transfer towards the colder bulk material through the melt pool borders has been calculated on two planes adjutant to the laser track, see Figure 13. According to Figure 13 (a), the net output power predicted by the thermo-fluid model

395 reaches 16 W within 3 milliseconds while for the pure conduction model this value would be  
 396 around 10 W, which is one-third lower. The same trend is also observed on the y-z plane at the  
 397 back of the melt pool, according to Figure 13 (b). Hence, the inclusion of fluid flow inside the melt  
 398 pool will highly increase the rate of heat transfer which will ultimately lead to lower peak  
 399 temperatures and smaller melt pool sizes.

400 **Parametric study**

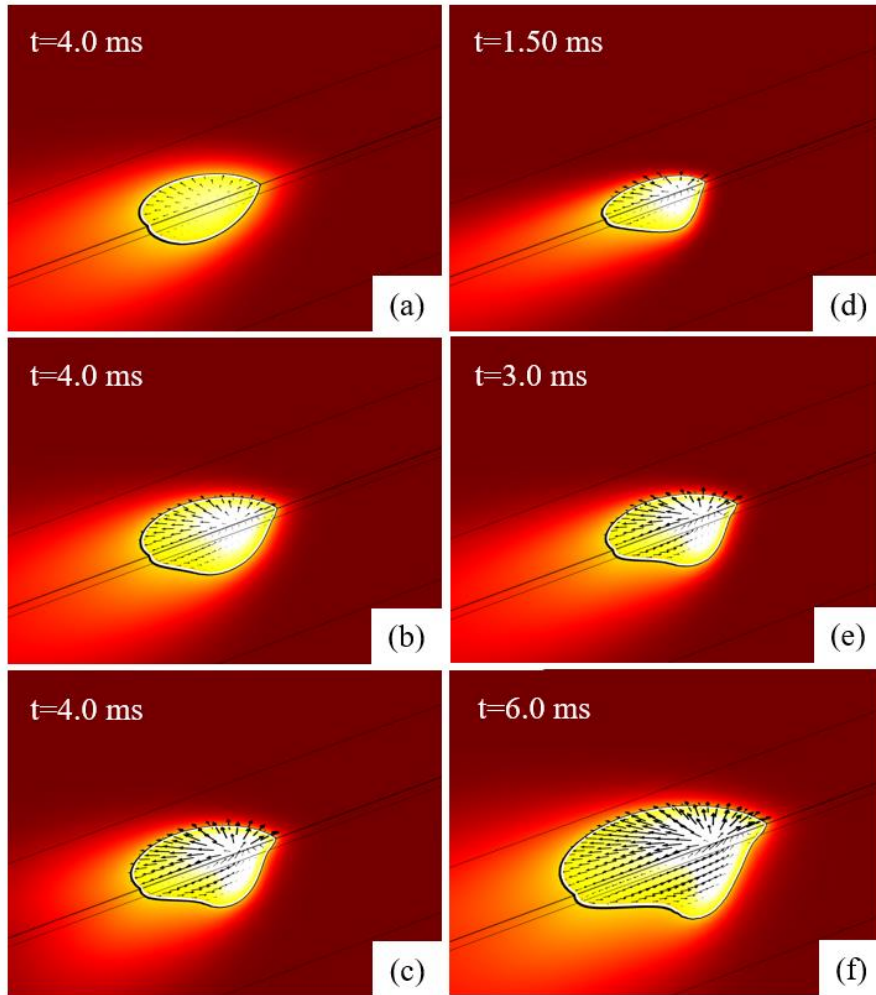
401 A parametric study has been performed to separately investigate the effect of the laser beam radius  
 402 and its travel speed on shape and size of the melt pool and its peak temperature. The process  
 403 parameters for the parametric study are given in Table 4. Two different cases have been analyzed  
 404 here, one group with varying beam radius and another group with varying scanning speeds,  
 405 denoted the R-group and V-group, respectively.

406 Table 4. Process parameters and case ids for the parametric study.

group	Case id	Process parameter			
		Power (W)	Scan speed (mm.s <sup>-1</sup> )	Beam radius (μm)	Layer thickness (μm)
V-group	V200		200		
	V400	200	400	50	20
	V800		800		
R-group	R50			50	
	R90	200	300	90	20
	R150			150	

407 The stable melt pool temperature contour and velocity fields are shown for all six different cases  
 408 in Figure 14 (a)-(f) and at the end of each process.



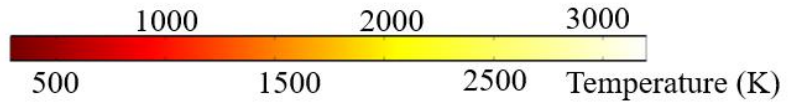


100 μm

1893.2 K 1927.2 K

→

Velocity magnitude: 5.13 m.s<sup>-1</sup>



409

410 Figure 14. Temperature contour and velocity field along with the melt pool borders shown for case: (a) R150, (b) R90, (c) R50, (d)  
 411 V800, (e) V400 and (f) V200, at the end of the process. Note that for larger beam sizes, the melt pool will have a more uniform  
 412 shape.

413 It is clearly seen in Figure 14 (a)-(c) that increasing the laser beam radius will lower the peak  
 414 temperature and the overall molten zone. Furthermore, it is revealed that by reducing the size of  
 415 the laser beam radius from 150 μm to 90 μm and finally 50 μm, the melt pool's geometry becomes  
 416 more asymmetric, while the speed is kept constant. The reason behind this transformation can be  
 417 attributed to the fact that for case R150, the heat flux has lower peak values and is more uniformly  
 418 distributed.

419 It is also seen from Figure 14 (a)-(c) that for smaller beam sizes, because of the existence of larger  
 420 temperature gradients, the maximum velocity magnitude will be higher as well, which is directly

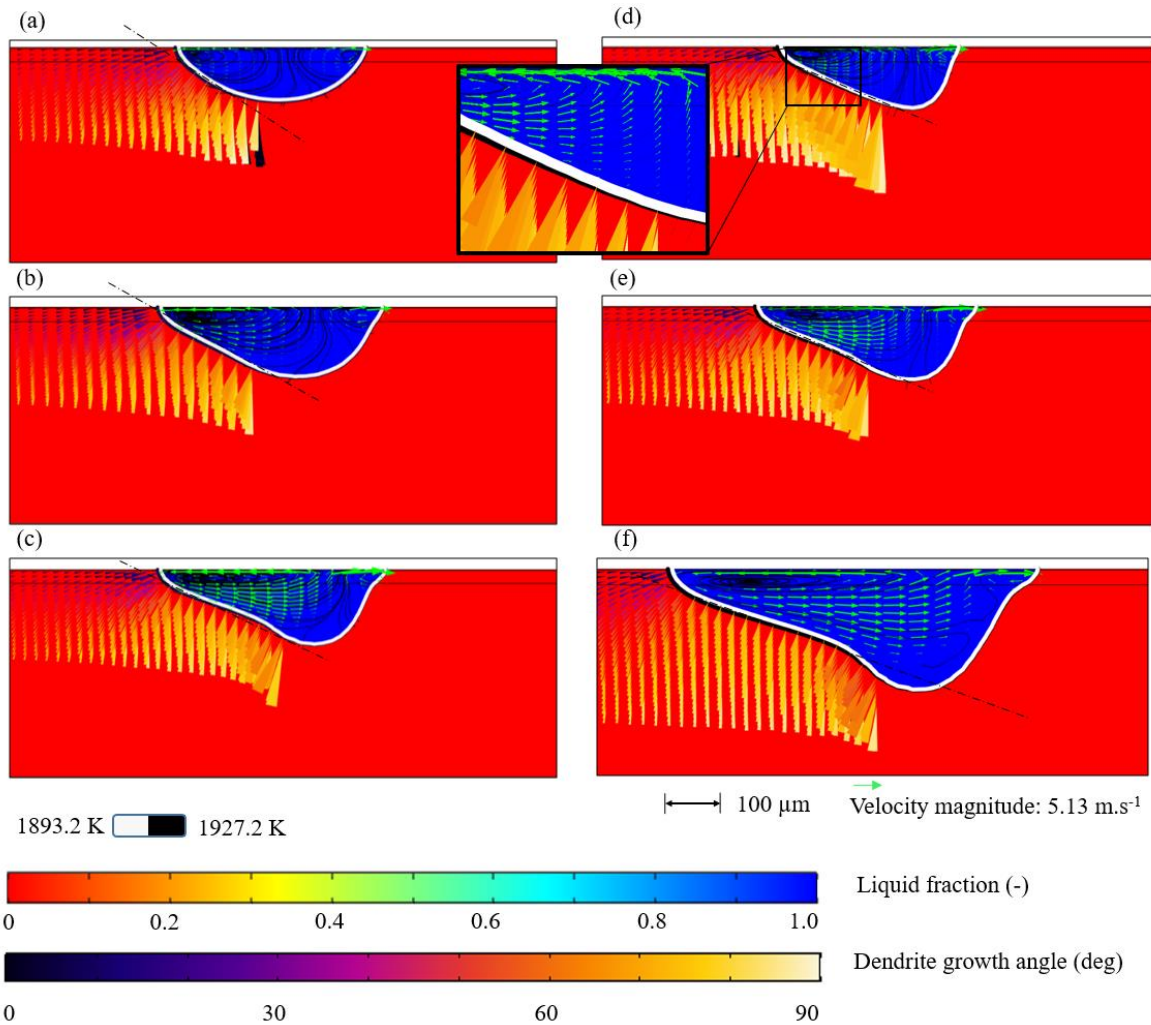
421 linked to formation of elevated thermally-induced shear stresses in the fluid. On the other hand,  
 422 increasing the laser travel speed will result in a more asymmetrical melt pool, with shorter tails,  
 423 according to Figure 14 (d)-(f). For further analysis regarding the melt pool size, peak temperature  
 424 and opposing forces, relevant quantities have been calculated and presented in Table 5.

425 Table 5. Data regarding melt pool information and the corresponding dimensionless numbers for the parametric study.

Case id	Melt pool data				Dimensionless numbers				
	Volume (m <sup>3</sup> )	Peak Temp. (K)	Max. Velocity (m.s <sup>-1</sup> )	Vorticity (s <sup>-1</sup> )	Gr (-)	Ma (-)	Pe (-)	Ar (-)	Ec (-)
V200	2.2E-12	6388.9	5.1	2174.0	2.1e-04	8414.7	33.9	2.4e-09	7.7e-06
V400	1.1E-12	5672.1	4.9	768.2	9.3e-05	5710.0	26.3	1.7e-09	8.5e-06
V800	5.1E-13	4220.5	4.1	308.2	2.5e-05	2676.5	16.7	1.2e-09	9.6e-06
R50	1.5E-12	6091.9	5.3	1033.6	1.4e-04	6966.5	31.1	1.8e-09	8.8e-06
R90	1.5E-12	4091.7	3.6	661.8	6.9e-05	3610.3	21.1	2.0e-09	7.9e-06
R150	9.8E-13	2730.5	2.0	282.3	1.8e-05	1200.0	10.0	2.3e-09	6.1e-06

426 It can be seen that lower travel speeds will lead to higher peak temperatures and bigger melt pools  
 427 and one can say that there is a linear relation between the inverse travel speed and the melt pool  
 428 size, which goes well in hand with the classical effect of linear heat input [27]. On the other hand,  
 429 an increase in beam size will lead to a sharp decrease in the peak temperature. Moreover, from the  
 430 table, it is evident that the Marangoni number is directly linked with the input parameters and that  
 431 a reduction in beam size and travel speed will lead to bigger thermally-induced shear forces, which  
 432 will directly impact the mode of heat transfer, via changing the fluid flow patterns. Accordingly,  
 433 the associated Peclet number will also rise, as either the beam size or travel speed decreases, hence  
 434 resulting in a more pronounced convective heat transfer. On the other hand, the role of the  
 435 buoyancy forces are very negligible in both heat and fluid flows, which is seen from low levels of  
 436 the Grashof's number given in Table 5. However, it is the Archimedes number ( $Gr.Re^{-2}$ ) that  
 437 decides the relative importance of free convection (due to buoyancy) to forced convection (due to  
 438 thermally-induced shear stresses). This number, according to Table 5 is very negligible, so the  
 439 dominant mode of convection heat transfer is the forced convection. Furthermore, to study the  
 440 relative weight of kinetic energy to the internal energy, the Eckert number is calculated (defined  
 441 as  $V^2.C_P^{-1}.\Delta T^{-1}$ ) and given in Table 5. As expected, it is observed that by reducing the travel speed,  
 442 the Eckert number goes down as well. Moreover, according to the table, bigger beam sizes lead to  
 443 lower Eckert numbers, which is also expected, since lower speed levels are found. Finally, the  
 444 calculated average vorticity inside the melt pool is also provided in Table 5 and accordingly lower  
 445 beam size and travel speed will lead to stronger circulations and vortices, which is directly linked  
 446 to higher thermally-induced shear forces.

447 The melt pool profile in the x-z plane, including the liquid fraction contour, solidus and liquidus  
 448 lines, velocity vectors, stream lines and more importantly, the vectors of direction of dendrite  
 449 growth for all six mentioned cases are illustrated in Figure 15.

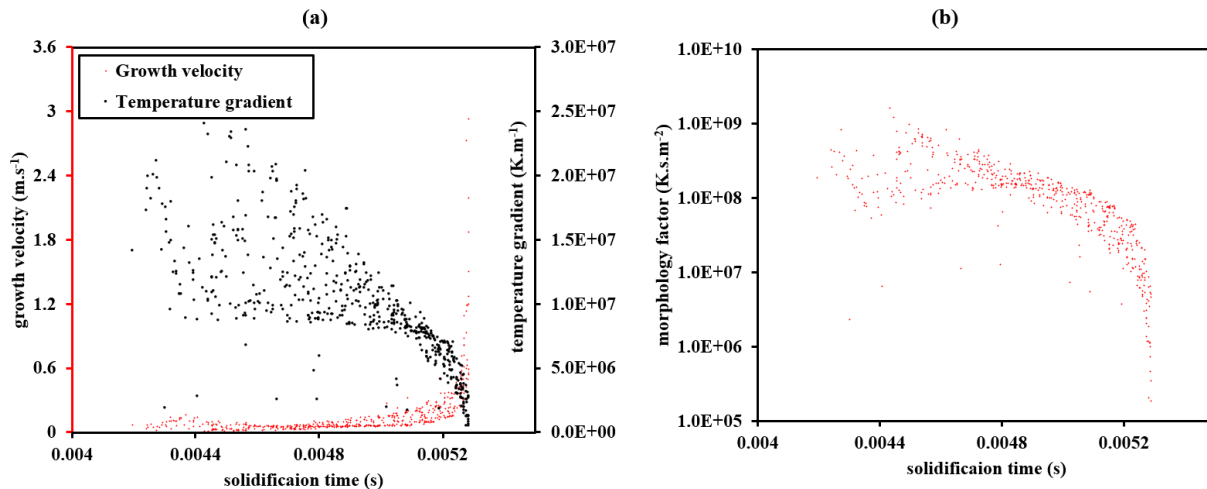


450  
 451 Figure 15. Contour of liquid fraction and vectors of plane velocity field along with the vectors of direction of dendritic growth  
 452 (shown with cones) for case: (a) R150, (b) R90, (c) R50, (d) V800, (e) V400 and (f) V200 after reaching their corresponding  
 453 stabilization times. Note the formation of the two vortices on the front and back of the melt pool.

454 Several interesting informations regarding the fluid dynamics and microstructure can be deduced  
 455 from Figure 15. It is clearly seen in Figure 15 (a)-(c) that by increasing the size of the laser beam  
 456 radius, the melt pool becomes more symmetric. According to Figure 15 (a), for a relatively large  
 457 beam radius, the melt pool profile becomes elliptical in the x-z plane and two almost equally-sized  
 458 vortices are formed on the front and back of the melt pool. However, by reducing the beam size,  
 459 hence imposing higher heat fluxes on the domain, the front vortex gets much smaller with higher  
 460 x-direction velocities while the rear circulation expands dramatically. Furthermore, based on  
 461 Figure 15 (d)-(f), by reducing the scanning speed of the laser, the size of the melt pool increases  
 462 in all directions.

463 The effect of process parameters on the dendritic growth directions is interesting as well. It is very  
 464 well established in the literature that the dendrites grow parallel to the solidification temperature  
 465 gradient [56]. The colored cones shown in Figure 15 represent the direction of the dendritic growth,  
 466 which is parallel to the heat flow direction [61], and it is seen that these lines are all perpendicular  
 467 to the tail of the melt pool (see blowup for clarification). It is moreover observed from Figure 15  
 468 (a)-(c) that reducing the size of the laser beam, the mean dendritic growth direction angle with  
 469 respect to the horizontal plane decreases. Moreover, based on Figure 15 (d)-(f), the reduction in  
 470 scanning speed results in a more horizontal tail of the melt pool (is quantified later on in Table 6)  
 471 which will consequently lead to lower angles of dendritic growth as well.

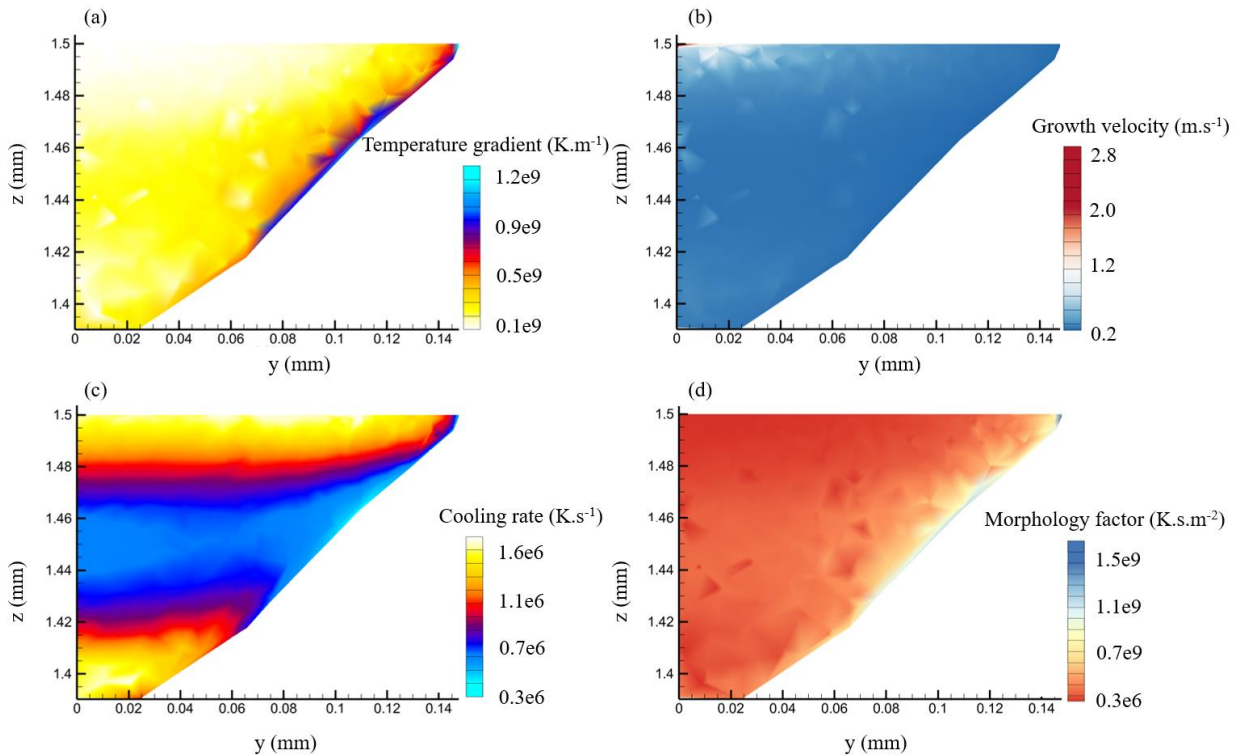
472 The solidification parameters, including cooling rate, morphology factor, temperature gradient and  
 473 solidification growth speed are calculated by means of a microstructural sub-model described in  
 474 section 2.4. The solidification temperature gradient and solidification growth velocity are plotted  
 475 against time for a cross-section of the melt pool in a y-z plane, at  $x=1.4$  mm, in Figure 16 (a). Each  
 476 point in Figure 16 corresponds to a solidified finite element node in the model in the noted cross-  
 477 section.



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 479 Figure 16. (a) Solidification growth velocity and temperature gradient versus time and (b) the plot of calculated morphology factor  
 480 against time. The calculation is carried out on a y-z plane perpendicular to the laser path at  $x=1.4$  mm for V200 case.

481 According to Figure 16 (a), the solidification growth velocity increases steadily with time and it is  
 482 observed that at the end of solidification, the growth velocity rises dramatically to a significant  
 483 amount of  $3 \text{ m.s}^{-1}$ . The underlying reason for this sudden increase can be explained by the fact that  
 484 at the final phase of the solidification, the ratio of the melt pool total area to its remaining volume  
 485 will be very big. The area-volume ratio is also an indicator of the cooling capability to the  
 486 remaining energy inside the melt pool due to the latent heat of fusion. In this regard, at the end of  
 487 the solidification, the solidification speed grows dramatically. Moreover, based on Figure 16 (a),  
 488 the temperature gradient experiences a different trend compared to that of the growth velocity and  
 489 decreases during the course of the solidification process. It is noteworthy to mention that the same  
 490 trend is observed for both solidification thermal gradient and growth speed, for the EBM process  
 491 of metals as well [33]. The morphology factor which is defined as the ratio of the solidification  
 492 thermal gradient to the solidification growth speed is determined for the mentioned cross-section

493 and has been plotted against time in Figure 16 (b) and it is well observed that the morphology  
 494 factor decreases during the solidification process continuously. A lower morphology factor is an  
 495 indicator of dominant equiaxed microstructure while higher levels stand for columnar morphology  
 496 [17]. Typically the columnar microstructures lead to an unwanted anisotropy in mechanical  
 497 properties of the parts which require an additional post-process (heat treatment) to remove it [62].  
 498 The contours of the mentioned solidification temperature gradient and growth speed are shown in  
 499 Figure 17 (a) and (b), respectively.



500  
 501 Figure 17. Contours of: (a) solidification temperature gradient, (b) growth speed, (c) cooling rate and (d) morphology factor at a y-  
 502 z cross-section at  $x=1.4$  mm at  $t=6$  ms for V200 case. Due to the symmetry only half of the data are shown.

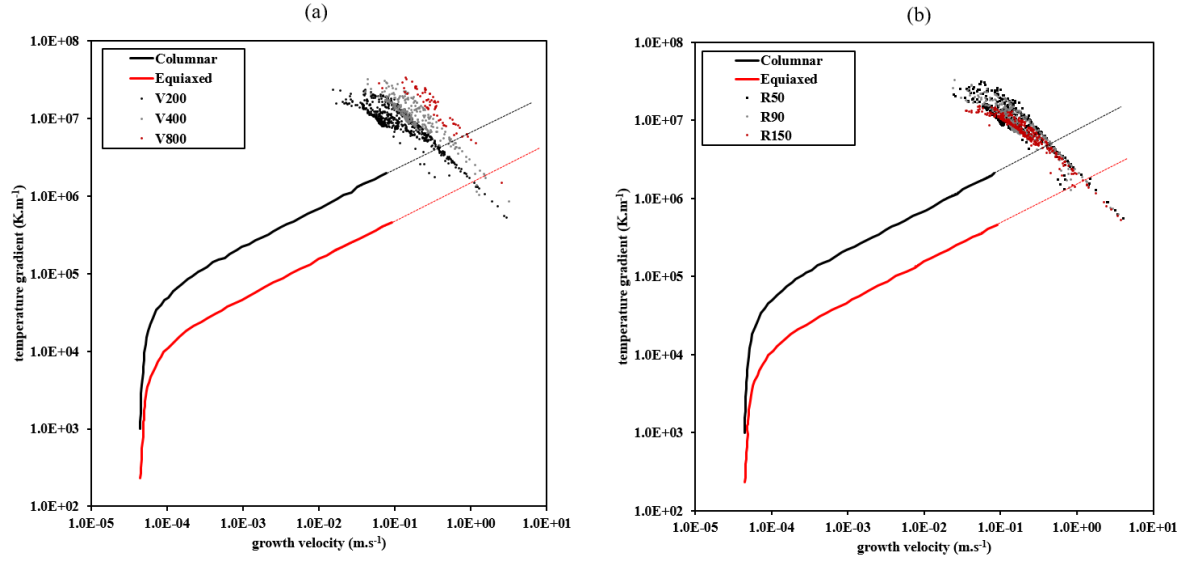
503 Based on Figure 17 (a), the maximum value of the solidification temperature gradient is found on  
 504 the initial melt pool borders and it decreases steadily as the melt pool front moves upwards. The  
 505 solidification growth speed is observed to be highest on the top of the melt pool and in the center  
 506 line, based on Figure 17 (b), where both findings are consistent with numerical results of the laser  
 507 welding process as well [17]. The contour of the cooling rate is shown in Figure 17 (c), where the  
 508 cooling rate is highest at the top and bottom of the initial melt pool's borders and has its lowest  
 509 values in the middle of the centerline. The cooling rate is a good indicator for the size of the grains  
 510 or dendrites forming during solidification where lower cooling rates lead to bigger grain or  
 511 dendrite sizes [15], [16]. Thijs et al. [63] reported that the microstructure at the bottom of the melt  
 512 pool is much finer than the one found in the internal region, which is in accordance with the cooling  
 513 rate contour shown in Figure 17 (c). On the other hand, the morphology factor is a tool to study  
 514 the morphology of the grains formed. In general, a decrease in the morphology factor results in  
 515 formation of columnar and even equiaxed grains. In Figure 17 (d), the contour of the morphology  
 516 factor is shown at the cross-section. The maximum value of the morphology factor occurs at the

**517** borders of the melt pool and lowest on the center line, leading to higher probability of formation  
**518** of equiaxed grains on this line and at the final phase of the solidification, which is also seen for  
**519** laser welding [17], [64]. The calculated mean solidification parameters for all the six different  
**520** cases are given in Table 6. The values for the solidification thermal gradient and growth velocity  
**521** are in the same order as the ones reported for another L-PBF process in the literature [40].  
**522** According to this table, the cooling rate and solidification growth velocity, both increase with  
**523** increasing laser travel speed, leading to smaller grains. The same observation was also reported  
**524** for laser welding and EBM [17], [58]. On the other hand, the morphology factor drops as the  
**525** scanning speed goes up, hence increasing the probability of formation of more equiaxed grains.  
**526** Also, the calculated dendrite direction angle, with respect to the horizon, will increase by  
**527** increasing the beam size. In this way, one can manipulate the direction, size and morphology of  
**528** the dendrites formed during the L-PBF process, by changing the process parameters such as travel  
**529** speed and beam size.

**530** Table 6. Average value of the calculated solidification parameters.

Case id	Solidification parameters				
	C (K.s <sup>-1</sup> )	G (K.m <sup>-1</sup> )	F (K.s.m <sup>-2</sup> )	R (m.s <sup>-1</sup> )	Θ (deg)
	Cooling rate	Temperature gradient	Morphology factor	Growth velocity	Dendritic direction angle
V200	1.05E+06	1.00E+07	1.52E+08	0.174	19.97
V400	2.02E+06	1.23E+07	1.00E+08	0.249	23.06
V800	4.59E+06	1.60E+07	7.69E+07	0.399	22.75
R50	1.53E+06	1.14E+07	1.25E+08	0.237	25.51
R90	1.56E+06	1.02E+07	9.97E+07	0.238	29.25
R150	1.46E+06	7.84E+06	6.34E+07	0.302	32.87

**531** Furthermore, to investigate the morphology of the grains formed during the mentioned process, a  
**532** plot of temperature gradient versus solidification growth speed has been made for varying scan  
**533** speeds in Figure 18 (a) and varying beam sizes in Figure 18 (b).



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Figure 18. G-R map for varying (a) scan speed and (b) beam size. The red and black lines respectively show the approximate borders of the equiaxed and columnar regions for Ti6Al4V. Above the black line it is purely columnar, whereas below the red line it is purely equiaxed. The dashed lines stand for extrapolated data. The map is based on the data found in [32].

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According to Figure 18 (a) and (b), most of the solidified region lies in the columnar zone, which is in agreement with experimental findings for L-PBF [62]. It is also observed that the morphology of the solidified zones moves towards the equiaxed zone at the end of the solidification where low thermal gradients along with high growth speeds are predicted, according to Figure 18 (a). Interestingly, a similar trend is observed for the EBM process, where at the end of the process, lower thermal gradients and higher growth speeds are obtained [32], [33]. This is also in agreement with the decrease in the morphology factor during the course of the solidification noted earlier. Overall, one can conclude from Figure 18 (a) that higher scanning speeds lead to higher thermal gradients and growth speeds, while the effect of changing beam size on the final solidification morphology is much lower as compared to varying the travel speed and in this way the morphology of the samples are much more sensitive to scanning speed than to beam size.

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On the other hand, based on Table 6 and Figure 18 (a), higher laser beam speeds will cause higher solidification cooling rates. In this context, it should be mentioned that Zhang et al. introduced a simple expression that relates the cooling rate to the laser beam speed for the L-PBF process [65]

$$Cr(K.s^{-1}, mm.s^{-1}) = (2.07 \times 10^4) V_{beam}^{1.2}. \quad (39)$$

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Where  $V_{beam}$  ( $mm.s^{-1}$ ) in the above expression is the laser beam speed and according to this equation, higher cooling rates are obtained for higher scanning speeds, which shows the same trend as the results given in Table 6. Also, another simple equation which relates the grain sizes to the solidification cooling rate, based on a rapid solidification assumption, has been suggested by Broderick et al. [66]

$$d(\mu m) = (3.1 \times 10^6) Cr(K.s^{-1})^{-0.93 \pm 0.1}. \quad (40)$$

557 Based on this, higher cooling rates will cause smaller grain sizes and based on what was mentioned  
558 earlier, one can obtain smaller grain sizes by simply choosing higher laser speeds. Furthermore,  
559 the tensile strength of the material is highly dependent on the grain sizes of the domain. As  
560 indicated by the following Hall-Petch-like empirical correlation [67]

$$Y \text{ (MPa)} = 802.66 + \frac{1236.5}{d(\mu\text{m})^{0.5}}, \quad (41)$$

561 which relates Ti6Al4V's tensile strength, denoted  $Y$  (MPa), to the average beta grain size  $d$ .  
562 According to equation (41) finer grain sizes will lead to higher tensile strength. Thus, one can  
563 improve a part's mechanical strength by simply increasing the laser beam speed, which causes the  
564 formation of finer grains during the solidification process. However, according to both Figure 18  
565 and Table 6, changing the laser beam will not have a similar significant effect on the cooling rate,  
566 hence it influences the mechanical strength and grain morphology to a lesser extent. Moreover, it  
567 should be noted that the dominant grain morphology is still columnar for all cases studied and  
568 accordingly mechanical anisotropy is inevitable, which necessitates a post-process such as heat  
569 treatment to remove it.

## 570 Conclusion

571 In this work, a numerical model based on the FEM framework has been developed in COMSOL  
572 Multiphysics to study the heat and fluid flow along with metallurgical conditions during the L-  
573 PBF process of the Ti6Al4V alloy. A systematic investigation regarding the impact of neglecting  
574 fluid flow inside the melt pool on the heat flow and melt pool dimensions is presented, alongside  
575 with a thorough analysis in terms of relevant dimensionless numbers. Also, a novel conico-  
576 Gaussian heat source is developed to model the thermal interaction between the part and the laser,  
577 which relies on the concept of optical penetration depth (OPD). An analytical expression is derived  
578 which can be used to adjust the shape and geometry of the melt pool for validation. Furthermore,  
579 the model is both numerically verified through mesh independency analysis and validated with  
580 experimental results. The results show that neglecting the fluid flow will result in overestimated  
581 temperature fields and unrealistically large melt pools. Also the results show that the dominant  
582 mode of heat transfer is convection, as the Peclet number is significantly larger than one.  
583 Moreover, it is shown that the role of the buoyancy effect on heat and fluid flow is negligible  
584 compared to the much more pronounced effect of the thermally-induced shear forces. A parametric  
585 study is carried out in the second part of the paper to study the effect of varying beam size and  
586 travel speed on melt pool shape, solidification pattern and size and morphology of the grains. To  
587 study the metallurgical conditions, a microstructural sub-model is developed and coupled to the  
588 CFD model. It is observed that at the end of the solidification process, the morphology tends to  
589 become more equiaxed, compared to the onset of the process where it is fully columnar. Also, it  
590 is found that by choosing different process parameters, one can manipulate the direction of the  
591 dendrites' growth. Specifically, it is found that the overall effect of changing beam size on grain  
592 morphology is less pronounced than changing the travelling speed.



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