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Performance of `interFoam` on the simulation of progressive waves

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ABSTRACT

The performance of `interFoam` (a widely-used solver within the popular open source CFD package `OpenFOAM`®) in simulating the propagation of a nonlinear (stream function solution) regular wave is investigated in this work, with the aim of systematically documenting its accuracy. It is demonstrated that over time there is a tendency for surface elevations to increase, wiggles to appear in the free surface, and crest velocities to become (severely) over estimated. It is shown that increasing the temporal and spatial resolution can mitigate these undesirable effects, but that a relatively small Courant number is required and fine discretization is needed, indicating that many past simulations have not converged. It is further demonstrated that the choice of discretization schemes and solver settings (often treated as a "black box" by users) can have a major impact on the results. This impact is documented, and it is shown that obtaining a "diffusive balance" is crucial to accurately propagate a surface wave over long distances without requiring exceedingly high temporal and spatial resolutions. Finally, the new code `isoAdvector` is compared to `interFoam`, which is demonstrated to produce comparably accurate results, while maintaining a sharper surface. It is hoped that the systematic documentation of the performance of the `interFoam` solver will enable its more accurate and optimal use, as well as increase awareness of potential shortcomings, by CFD researchers interested in the general CFD simulation of free surface waves.

KEYWORDS

`interFoam`, waves, discretization practises, `isoAdvector`

1. Introduction

As a tool to simulate waves `interFoam`, in the widely-used CFD package `OpenFOAM`® (or other solvers build on `interFoam`, e.g. `waves2Foam` developed by Jacobsen et al. (2012)) are becoming increasingly popular. As examples, `interFoam` has been utilized to simulate breaking waves by e.g. Jacobsen et al. (2012); Brown et al. (2016); Jacobsen et al. (2014); Lupieri and Contento (2015); Higuera et al. (2013). It has also been used to simulate wave-structure interaction by e.g. Higuera et al. (2013); Chen et al. (2014); Paulsen et al. (2014); Hu et al. (2016); Jacobsen et al. (2015); Schmitt and Elsaesser (2015).

Wave breaking and wave-structure interaction are both very complex phenomena, but `interFoam` has also been utilized to simulate more simple cases, such as the pro-

gression of a solitary wave by Wroniszewski et al. (2014), which was suggested as a benchmark to compare to other CFD codes. The study by Wroniszewski et al. (2014) highlighted a problem, that to our knowledge, has gone largely unnoticed in the formal journal literature, namely that the velocity at the crest of the wave is over-predicted relative to the analytical solutions. This was also highlighted in conference paper Roenby et al. (2017), the MSc thesis of Afshar (2010) and the PhD thesis Tomaselli (2016). A second problem was highlighted in the study by Paulsen et al. (2014), where it was shown that `interFoam` is not capable of maintaining a constant wave height for long propagation distances. They also mentioned, though not going into great detail, that the choice of convection scheme affected this behaviour. The choice of convection scheme was also briefly touched upon by Wroniszewski et al. (2014), who, like Paulsen et al. (2014), utilized an upwind scheme, chosen for stability reasons. These two studies thus hinted at the importance of discretization practises when using `interFoam` to simulate waves, but no further discussion of this was made. A third (again not well described in the literature) problem is the appearance of wiggles in the air-water interface, as documented by Afshar (2010). A fourth problem, which has received considerable attention (though not in the context of waves), is the growth of spurious velocities in low density fluid near the interface; see e.g. Francois et al. (2006); Meier et al. (2002); Rudman (1997); Popinet and Zaleski (1999); Shirani et al. (2005); Menard et al. (2007); Tanguy et al. (2007); Galusinski and Vigneaux (2008); Hysing (2006). The previous mentioned studies all related the growth of spurious velocities to the surface tension. More recently, however, it should be noted that Vukcevic (2016); Vukcevic et al. (2016, 2017); Wemmenhove et al. (2015) demonstrated development of spurious velocities in situations without surface tension.

While a benchmark case as presented in Wroniszewski et al. (2014) is, in principal, a good idea many relevant details of the `interFoam` setup were not presented, and this is typically the case in many of the previous mentioned studies. Such details are quite important, at least from the perspective of benchmarking, as it turns out that the performance of `interFoam` is quite sensitive to the setup (briefly touched upon in Paulsen et al. (2014) and Wroniszewski et al. (2014) in the choice of convection scheme). Hence, prior to benchmarking `interFoam` or other CFD solvers, it is imperative that an "optimal" (or at least reasonably so) settings be known and utilized.

As the intended audience of the present paper is `OpenFOAM`® users, a working knowledge of this software is assumed throughout. To shed light on the general CFD simulation of surface gravity waves, the present study will systematically investigate the performance of `interFoam` on a canonical case involving a simple, intermediately deep, progressive regular wave train. It will demonstrate that taking `interFoam` "out of the box," i.e. utilizing the standard setup from one of the popular tutorials, will yield quite poor results (This could be expected since the `OpenFOAM`® tutorials are designed to run first and foremost stably rather than accurately). After showing the default performance of `interFoam` the sensitivity of `interFoam` to different settings will be investigated. First, a standard sensitive analysis is conducted with respect to the Courant number and mesh resolution. This is done specifically to highlight that commonly-used Courant numbers may not be sufficiently small to accurately simulate gravity waves, indicating that many past results might not have converged. Then, utilizing a lower Courant number, different `interFoam` settings will be systematically tested to demonstrate the importance of discretization considerations when simulating waves and finally the settings will be combined to form a reasonably optimal set up. The recently developed code `isoAdvector` will finally be coupled with `interFoam`, and the performance of `interFoam` (utilizing `isoAdvector` instead of `MULES`) will be

62 compared to the performance of the standard `interFoam` solver.

63 2. Model description

64 2.1. Hydrodynamics

65 The flow is simulated by solving the continuity equation coupled with momentum
66 equations, respectively given in (1) and (2):

$$\frac{\partial u_i}{\partial x_i} = 0, \quad (1)$$

67

$$\frac{\partial \rho u_i}{\partial t} + u_j \frac{\partial \rho u_i}{\partial x_j} = -\frac{\partial p^*}{\partial x_i} - g_j x_j \frac{\partial \rho}{\partial x_i} + \frac{\partial}{\partial x_j} (2\mu S_{ij}) + \sigma_T \kappa \frac{\partial \alpha}{\partial x_i}, \quad (2)$$

68 Here u_i are the mean components of the velocities, x_i are the Cartesian coordinates,
69 ρ is the fluid density (which takes the constant value ρ_{water} in the water and jumps at
70 the interface to the constant value ρ_{air} in the air phase), p^* is the pressure minus the
71 hydrostatic potential $\rho g_j x_j$, g_j is the gravitational acceleration, $\mu = \rho \nu$ is the dynamic
72 molecular viscosity (ν being the kinematic viscosity), and S_{ij} is the mean strain rate
73 tensor given by

$$S_{ij} = \frac{1}{2} \left(\frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i} \right). \quad (3)$$

74 The last term in equation (2) accounts for the effect of surface tension, σ_T , where
75 κ is the local surface curvature and α is the so-called indicator field introduced for
76 convenience, which takes value 0 in air and 1 in water. It can be defined in terms of
77 the density as

$$\alpha = \frac{\rho - \rho_{\text{air}}}{\rho_{\text{water}} - \rho_{\text{air}}}. \quad (4)$$

78 We assume that any intrinsic fluid property, Φ , can be expressed in terms of α as

$$\Phi = \alpha \Phi_{\text{water}} + (1 - \alpha) \Phi_{\text{air}}. \quad (5)$$

79 The evolution of α is determined by the continuity equation, which in terms of α reads

$$\frac{\partial \alpha}{\partial t} + \frac{\partial \alpha u_j}{\partial x_j} = 0. \quad (6)$$

80 In `interFoam` the numerical challenge of keeping the interface sharp is addressed using
81 a numerical interface compression method and limiting the phase fluxes based on the
82 "Multidimensional universal limiter with explicit solution" (`MULES`) limiter. Numerical
83 interface compression is obtained by adding a purely heuristic term to equation (6),
84 such that it attains the form

$$\frac{\partial \alpha}{\partial t} + \frac{\partial \alpha u_j}{\partial x_j} + \frac{\partial}{\partial x_j} (\alpha(1 - \alpha)u_j^r) = 0. \quad (7)$$

85 Here u_j^r is a modelled relative velocity used to compress the interface and is given by

$$u_j^r = \min \left(\frac{c_\alpha |F_f|}{|s_f|}, \max \left[\frac{|F_f|}{|s_f|} \right] \right) n_{f,j} \quad (8)$$

86 where c_α is a user defined value that determines the strength of the compression, F_f is
87 the face flux, $n_{f,j}$ is the j 'th component of the interface normal and s_f is the face area
88 vector normal to the face pointing out of the cell. For more details on the numerical
89 implementation, the reader is referred to Deshpande et al. (2012).

90 All simulations are performed utilizing OpenFOAM® version `foam-extend 3.2`. The
91 authors are aware of a "new" MULES algorithm (not present in the extend versions)
92 in newer versions from OpenFOAM-2.3.0, and also of the new commit support for
93 Crank-Nicolson on the time integration of α . Therefore the base case to be presented
94 later, was also simulated utilizing a newer version of the standard OpenFOAM®, namely
95 OpenFOAM-3.0.1. We were unable to produce significantly different results with these
96 newer versions as compared to our simulations with `foam-extend 3.2`, hence the base
97 performance demonstrated in what follows is likewise expected to be representative of
98 newer versions.

99 2.2. *Boundary and initial conditions*

100 For this study a simple base case of a regular propagating wave will be simulated
101 with various numerical settings. The quality of the simulated wave will be assessed
102 through comparison with the analytical solution in terms of surface elevations and
103 velocity profiles. We use a so-called stream function wave from Rienecker and Fenton
104 (1981), initialized with `waves2Foam` developed by Jacobsen et al. (2012), with a period
105 $T = 2$ s and wave height $H = 0.125$ m at a water depth of $h = 0.4$ m. This gives
106 $kh = 0.66$ and $H/h = 0.31$, which indicates that the simulated wave is non-linear and
107 at intermediate depth, with k being the wave number. The stream function solution can
108 be considered as a numerically exact wave solution based on nonlinear potential flow
109 equations. The properties have been selected to correspond to the incoming wave in the
110 well-known spilling breaker experiment of Ting and Kirby (1994). For all simulations
111 the wave will be propagated through a domain which is exactly one wave length long
112 and two water depths high with cyclic periodic boundary conditions on the sides.
113 Unless stated otherwise the domain is discretised into cells having an aspect ratio of
114 1 with the number of cells per wave height $N = H/\Delta y = 12.5$, resulting in cells with
115 $\Delta x = \Delta y = 0.01$ m. This results in a two dimensional domain with 379×80 cells. At
116 the bed a slip condition is utilized for the velocities in accordance with potential flow
117 theory. At the top the `pressureInletOutletVelocity` is used. This means that there
118 is a zero gradient condition except on the tangential component which has a value of
119 zero. For p^* zero-gradient conditions are used for the bed and the periodic boundaries
120 whereas the top used a `totalPressure` condition with `p0=0`. Note that this setup
121 was also used in the study by Larsen and Fuhrman (2018) in the testing of their new
122 turbulence model.

123 3. interFoam settings

124 In this section the default numerical settings for our simulations, as well as a general
125 description of OpenFOAM®’s discretization practices, are presented. Our base numerical
126 settings will be those found in the popular `damBreak` tutorial shipped with
127 `foam-extend-3.2`. With this starting point we will change various settings to investigate
128 their effect on the quality of the numerical solution. More specifically, we copy the
129 `controlDict`, `fvSchemes` and `fvSolution` files directly from the `damBreak` tutorial. In
130 the `constant` directory the mesh and the physical parameters of the case are specified:
131 $\rho_{\text{water}}=1000 \text{ kg/m}^3$, $\rho_{\text{air}} = 1.2 \text{ kg/m}^3$, $\nu_{\text{water}} = 1 \cdot 10^{-6} \text{ m}^2/\text{s}$, $\nu_{\text{air}} = 1.45 \cdot 10^{-5} \text{ m}^2/\text{s}$,
132 and $\sigma_T = 0.0 \text{ N/m}$ (i.e. no surface tension). We note that the analytic stream function
133 solution does not take into account the presence of air, nor the effect of viscosity or
134 surface tension. With the chosen wave parameters and boundary conditions (e.g. no
135 slip at the bed) the physics are dominated by inertia and gravity. With a density ratio
136 of $\rho_{\text{water}}/\rho_{\text{air}} \sim 833$, the air will behave like a “slave fluid” moving passively out of the
137 way for the water close to the surface. To confirm the insignificance of the physical
138 viscosity in our setup, we have compared simulations with these set to their physical
139 values and to $\nu = 1 \cdot 10^{-16} \text{ m}^2/\text{s}$, and confirmed that this had no effect on our results.
140 We have also performed simulations with $\rho_{\text{air}} = 0.1 \text{ kg/m}^3$ and $\rho_{\text{air}} = 10 \text{ kg/m}^3$. This
141 had almost no effect in the short term, but had some effect for long propagation distances.
142 Increasing the density made the air behave less like a “slave fluid” and slowed
143 the propagation of the wave. Decreasing the density created larger air velocities, but
144 did not alter the wave kinematics significantly. We have confirmed that switching the
145 surface tension between zero and its physical value ($\sigma_T = 0.07 \text{ N/m}$) had next to
146 no effect on our simulation results, as expected in the gravity wave regime. Finally,
147 the simulations are performed without turbulence, as the results are intended to be
148 compared with the idealized stream function (potential flow) solution.

149 The OpenFOAM® case setup is contained in a file called `controlDict` which, among
150 others things, controls the time stepping method. The schemes used to discretize the
151 different terms in the governing equations are specified in the `fvSchemes` file, and the
152 file `fvSolution` contains various settings for the linear solvers and for the solution
153 algorithm. In Table 1 the essential parameters for the base set up from these three
154 files are indicated. The most important details of the scheme and solver choices presented
155 in Table 1 will be described in the following. For descriptions of the remaining
156 settings, the reader is referred to the OpenFOAM® user guide and programmers guides
157 in Greenshields (2015, 2016).

158 3.1. *controlDict*

159 In this subsection the most important `controlDict` settings are presented. The time
160 step can be specified either as `fixed`, such that the user defines the size of the time step,
161 or as `adjustable`. In the latter case the time step is adjusted such that a maximum
162 Courant number $Co = u_i \Delta t / \Delta x_i$ or a maximum `AlphaCo` (The Courant number in
163 interface cells) is maintained at all times. Since these two for the remainder of this
164 study are kept equal it is Co that controls the time step. In the `damBreak` tutorial an
165 adjustable time step is used with $Co = 0.5$, hence this value will be utilized initially.

Table 1. Base setup from the damBreak tutorial

controlDict	Scheme/Value
adjustTimeStep	true
maxCo	0.5
maxAlphaCo	0.5
fvSchemes	
ddt	Euler
grad	Gauss Linear
div(rho*phi,U)	Gauss LimitedLinearV 1
div(phi,alpha1)	Gauss VanLeer01
div(phirb,alpha1)	Gauss interfaceCompression
laplacian	Gauss linear corrected
interpolation	linear
snGrad	corrected
fvSolution	
pcorr(solver,prec,tol,relTol)	PCG, DIC, 1e-10, 0
pd(solver,prec,tol,relTol)	PCG, DIC, 1e-07, 0.05
pdFinal(solver,prec,tol,relTol)	PCG, DIC, 1e-07, 0
U(solver,prec,tol,relTol)	PBiCG, DILU, 1e-06, 0
cAlpha	1
momentumPredictor	yes
nOuterCorrectors	1
nCorrectors	4
nNonOrthogonalCorrectors	0
nAlphaCorr	1
nAlphaSubCycles	2

166 3.2. *fvSchemes*

167 In this subsection some of the discretisation schemes are presented to aid in the
168 understanding of the forthcoming analysis. The `ddt` scheme specifies how the time
169 derivative $\partial/\partial t$ is handled in the momentum equations. Available in `OpenFOAM` are:
170 `steadyState`, `Euler`, `Backwards` and `CrankNicolson`. In this study, `steadyState` is
171 naturally disregarded as the simulations are unsteady. The `Euler` scheme corresponds
172 to the first-order backward implicit Euler scheme, whereas `Backward` corresponds to
173 a second-order, `OpenFOAM` implemented time discretization scheme, which utilizes the
174 current and two previous time steps. The `CrankNicolson` (CN) scheme includes a
175 blending factor ψ , where $\psi = 1$ corresponds to pure (second-order accurate) CN and
176 $\psi = 0$ corresponds to pure `Euler`. This blending factor is introduced to give increased
177 stability and robustness to the CN scheme.

178 In the finite volume approach used in `OpenFOAM`, the convective terms in the mass
179 (7) and momentum (2) equations are integrated over a control volume, and afterwards
180 the Gauss theorem is applied to convert the integral into a surface integral:

$$\int_V \nabla \cdot (\phi u) dV = \oint_S \phi (n \cdot u) dS \approx \sum_f \phi_f F_f, \quad (9)$$

181 where $\phi(x, t)$ is the field variable, ϕ_f is an approximation of the face averaged field
182 value. ϕ_f can be determined by interpolation, e.g. using central or upwind differencing.
183 Central differencing schemes are second order accurate, but can cause oscillations in

184 the solution. Upwind differencing schemes are first order accurate, cause no oscillations,
 185 but can be very diffusive.

186 `OpenFOAM` includes a variety of total variation diminishing (TVD) and normalized
 187 variable diagram (NVD) schemes aimed at achieving good accuracy while maintaining
 188 boundedness. TVD schemes calculate the face value ϕ_f by utilizing combined upwind
 189 and central differencing schemes according to

$$\phi_f = (1 - \Gamma)\phi_{f,UD} + \Gamma\phi_{f,CD0} \quad (10)$$

190 where $\phi_{f,UD}$ is the upwind estimate of ϕ_f , $\phi_{f,CD}$ is the central differencing estimate of
 191 ϕ_f . Γ is a blending factor, which is a function of the variable r representing the ratio
 192 of successive gradients,

$$r = 2 \frac{d \cdot (\nabla \phi)_P}{\phi_N - \phi_P}. \quad (11)$$

193 Here d is the vector connecting the cell centre P and the neighbour cell centre N .
 194 In NVD-type schemes the limiter is formulated in a slightly different way. In the
 195 `damBreak` tutorial base setup the TVD scheme is utilized by specifying the key-
 196 word `limitedLinearV 1` for the momentum flux, `div(rho*phi,U)`, and `vanLeer01`
 197 for the mass flux, `div(phi,alpha1)`, where the keyword `phi` means face flux. With
 198 the `limitedLinear` scheme $\Gamma = \max[\min(2r/k, 1), 0]$, where k is an input given by
 199 the user, in this case $k = 1$. When using the scheme for vector fields a "V" can be
 200 added to the TVD schemes, which changes the calculation of r to take into account
 201 the direction of the steepest gradients. The `vanLeer` scheme calculates the blending
 202 factor as $\Gamma = (r + |r|)/(1 + |r|)$. The `01` added after the TVD scheme name means
 203 that Γ is set to zero if it goes out of the bounds 0 and 1, thus going to a pure upwind
 204 scheme to stabilize the solution. The other available TVD/NVD schemes differ in their
 205 definition of Γ and resulting degree of diffusivity. Since r depends on the numerically
 206 calculated gradient of ϕ , the choice of gradient scheme can also play an important role.
 207 In general the gradients are calculated utilizing a Gauss linear scheme, but this might
 208 lead to unbounded face values, and therefore gradient limiting can be applied. As an
 209 example the gradient scheme can be specified as `Gauss faceMDLimited`. The keyword
 210 `face` or `cell` specifies whether the gradient should be limited base on cell values or
 211 face values and the keyword `MD` specifies that it should be the gradient normal to the
 212 faces. In addition to the linear choice of gradient schemes there also exists a least
 213 square scheme as well as a fourth order scheme.

214 The `laplacian` scheme specifies how the Laplacian in the pressure correction equa-
 215 tion within the PISO algorithm, as well the third term on the right hand side of
 216 equation (2), should be discretized. It requires both an interpolation scheme for the
 217 dynamic viscosity, μ , and a surface normal gradient scheme `snGrad` for ∇u . Often
 218 a `linear` scheme is used for the interpolation of μ and the proper choice of surface
 219 normal gradient scheme depends on the orthogonality of the mesh. Besides being used
 220 in the Laplacian, the `snGrad` is also used to evaluate the second and fourth term on
 221 the right hand side of equation (2). Often a linear scheme will be used, with or with-
 222 out orthogonality correction. Another option is to use a fourth order surface normal
 223 gradient approximation. Finally, the interpolation scheme determines how values are
 224 interpolated from cell centres to face centres.

225 3.3. *fvSolution*

226 In the `fvSolutions` file the iterative solvers, solution tolerances and algorithm settings
227 are specified. The available iterative solvers are preconditioned (bi-)conjugate gradient
228 solvers denoted `PCG/PBiCG`, a `smoothSolver`, generalised geometric-algebraic multi-
229 grid, denoted `GAMG`, and a `diagonal` solver. Each solver can be applied with different
230 preconditioners and the smooth solver also has several smoothing options. The `GAMG`
231 solver works by generating a quick solution on a coarse mesh consisting of agglomerated
232 cells, and then mapping this solution as the initial guess on finer meshes to finally
233 obtain an accurate solution on the simulation mesh. The different preconditioners and
234 smoothers will not be discussed here, but Greenshields (2015, 2016) can be consulted
235 for additional details.

236 In addition to the solver choices the `PISO`, `PIMPLE` and `SIMPLE` controls are also given
237 in the `fvSolution` file. The `cAlpha` keyword controls the magnitude of the numerical
238 interface compression term in equation (7). `cAlpha` is usually set to 1 corresponding
239 to a “compression velocity” of the same size as the flow velocity at the interface. The
240 `momentumPredictor` is a switch specifying enabling activation/deactivation of the pre-
241 dictor step in the `PISO` algorithm. The parameter, `nOuterCorrectors` is the number of
242 outer correctors used by the `PIMPLE` algorithm and specifies how many times the entire
243 system of equations should be solved within one time step. To run the solver in “`PISO`
244 mode” we set `nOuterCorrectors` to 1. The parameter `nCorrectors` is the number
245 of pressure corrector iterations in the `PISO` loop. The parameter `nAlphaSubCycles`
246 enables splitting of the time step into `nAlphaSubCycles` in the solution of the α equa-
247 tion (7). Finally, the parameter `nAlphaCorr`, specifies how many times the `alpha` field
248 should be solved within a time step, meaning that first the `alpha` field is solved for,
249 and this new solution is then used in solving for the `alpha` field again.

250 4. Results and discussion

251 In this section the simulated results involving the propagation of the regular stream
252 function wave will be presented and discussed for various settings.

253 4.1. *Performance of interFoam utilizing the damBreak settings*

254 First, the “default” performance of `interFoam` in the progression of the stream function
255 wave is presented, utilizing the settings from the `damBreak` tutorial. The setup utilized
256 here will be considered as the base setup, and the remainder of the simulations in this
257 study will utilize this base setup with minor adjustments.

258 Starting from the analytical stream function solution imposed as an initial condi-
259 tion (utilizing the `waves2Foam` toolbox of Jacobsen et al. (2012)), the simulation is
260 performed for 200 s (corresponding to 100 periods). This is sufficiently long to high-
261 light certain strengths and problems of `interFoam`. Results are sampled at the cyclic
262 boundary 20 times per period. In Figure 1 the surface elevation time series is shown.
263 Quite noticeably, even though the depth is constant, the wave height immediately
264 starts to increase, and this continues until the wave at some point (approximately
265 at $t = 20T$) breaks. This rather surprising result demonstrates the potentially poor
266 performance of `interFoam`, as the wave does not come close to maintaining a con-
267 stant form. A similar result has been shown in Afshar (2010). A feature that seems
268 to contribute, though is not solely responsible for, the un-physical steepening of the

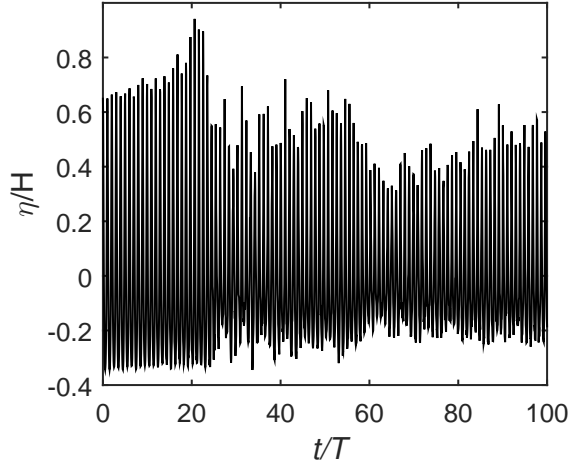


Figure 1. Surface elevation for the propagating wave utilizing the damBreak setup

269 wave, is small "wiggles" on the interface. These are illustrated in Figure 2 where a
 270 snapshot of the wave is seen after approximately five and 16 periods. The vertical
 271 axes are exaggerated to highlight the presence of the wiggles. As the wave propagates
 272 these wiggles emerge, continue to grow and sometimes merge, hence contributing to
 273 the steepening of the wave, which ultimately breaks. The cause of the wiggle feature
 274 will be discussed in Section 4.4.

275 While propagating, in addition to steepening, the celerity is also increasing com-
 276 pared to the analytical stream function solution, resulting in a phase error. To demon-
 277 strate this the surface elevation for the first 20 periods is compared with the stream
 278 function solution in Figure 3. Here it is quite evident that significant phase errors occur
 279 after approximately propagating for 10 periods, where the simulated results start to
 280 lead the analytical solution. This corresponds approximately to the time where over-
 281 steepening is apparent, hence the phase error may be attributed to the un-physical
 282 increase in the nonlinearity of the wave.

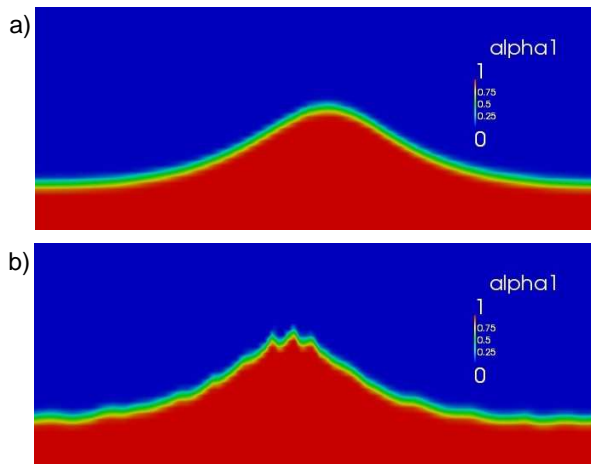


Figure 2. Snapshot at a) $t = 5.5T$ and b) $t = 16.25T$, illustrating the appearance of small wiggles in the crest after sufficiently long propagation

283 Also of great interest is the velocity profile beneath the propagating wave, as velocity
 284 kinematics often form the basis for force calculations on coastal or offshore structures,

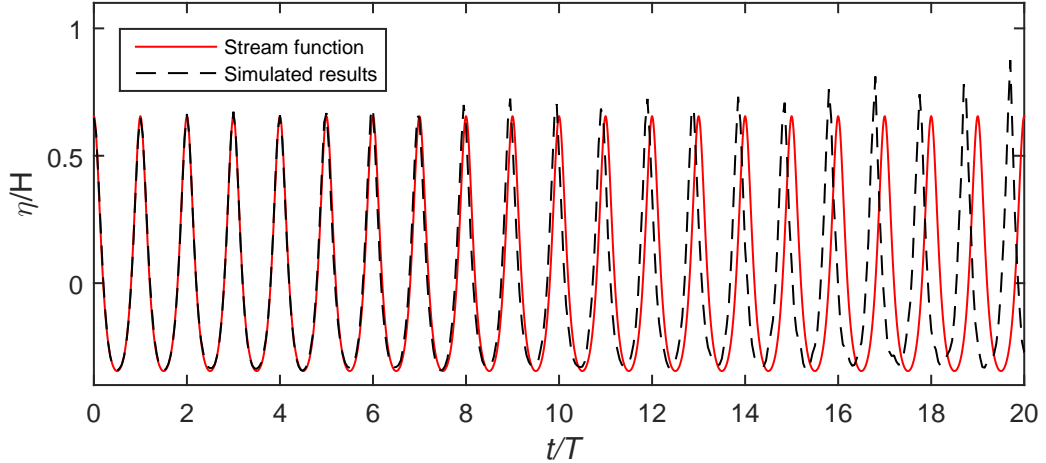


Figure 3. Surface elevation for the propagating wave utilizing the damBreak setup

285 while also influencing e.g. bed shear stresses and hence sediment transport predictions
 286 (in simulations where the boundary layer is also resolved). In Figure 4 the velocity
 287 profile directly beneath the crest of the wave after five periods is shown together with
 288 the analytical stream function solution. It should be noted that the velocity here, and
 289 in future results, is taken as $U = u_1\alpha$, and it is only shown from the bed until the
 290 height where it reaches its maximum value. This is done to capture the velocity all
 291 the way to the crest of the wave and not merely to a predefined height (as just shown,
 292 the wave height increases). Furthermore, this formulation also includes the velocities
 293 at cells containing a mixture of air and water, which is desirable, as some diffusion of
 294 the interface is seen.

295 As seen in Figure 4, the velocity beneath the crest is underestimated close to the
 296 bed and, especially near the free surface, is severely overestimated. This is despite the
 297 fact that the wave has still reasonably maintained its shape up to this time, see Figure
 298 2a and 3. This over-predicted crest velocity, in addition to the steepening of the wave,
 299 also likely contributes to the wave breaking. The overestimation of crest velocities in
 300 regular waves by *interFoam* has, to our knowledge, gone almost un-recognized in the
 301 journal literature. It is recorded in Wroniszewski et al. (2014) in the propagation of a
 302 solitary wave and in Roenby et al. (2017) as well as in the MSc thesis of Afshar (2010)
 303 and the PhD thesis of Tomaselli (2016). The overestimation of the crest velocity is
 304 believed to arise from an imbalance in the discretized momentum equation near the
 305 interface. As the wave propagates the increase in crest velocity becomes continually
 306 worse, and in addition to the imbalance in the momentum equation near the free
 307 surface, the steepening of the wave also contributes to this increase.

308 Finally, though not shown herein for brevity, we note that regions of high air ve-
 309 locities were seen to develop just above the free surface and in the mixture cells. Such
 310 spurious velocities have elsewhere been attributed to surface tension effects, see e.g.
 311 Deshpande et al. (2012), but the spurious velocities found in these simulation are
 312 clearly of a different nature as the surface tension is turned off. The main challenge
 313 leading to this behavior is that when the water/air density ratio is high, even small
 314 erroneous transfers of momentum across the interface from the heavy to the light fluid
 315 will cause a large acceleration of the light fluid, as also discussed by Vukcevic (2016);
 316 Vukcevic et al. (2016); Wemmenhove et al. (2015). The resulting large air velocities
 317 may then be subsequently diffused back across the interface into the water, the degree

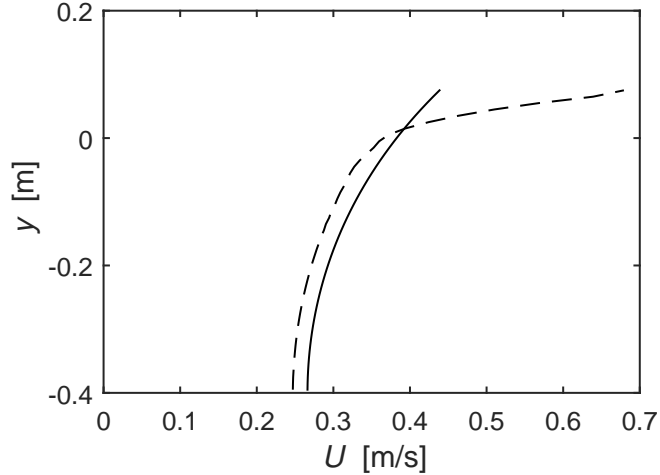


Figure 4. Simulated velocity distribution beneath the crest (- -) and stream function solution, (-) at $t = 5T$.

318 to which will be discussed in Section 4.4.

319 4.2. *Effect of the Courant number, Co*

320 With the poor performance previously shown using the default `damBreak` settings,
 321 two natural places to attempt improvement in the solution would be in the temporal
 322 and spatial resolutions. In this section the effect of the temporal resolution will be
 323 investigated by varying Co .

324 Figure 5 shows the surface elevation as a function of time for six different values of
 325 Co . From this it is evident that lowering Co has a significant impact on `interFoam`'s
 326 performance. However, even with $Co = 0.02$ `interFoam` is not capable of keeping the
 327 wave shape for 100 periods as the wave heights are still seen to increase. Up until 20
 328 wave periods the wave height is close to constant when using $Co \leq 0.15$. The wave is
 329 still leading the analytical stream function solution and in general lowering Co reduces
 330 the overestimation of the wave celerity as can be seen in table 2 where the phase-shift
 331 at $t = 25T$ is shown for the six different values of Co . The phase shift is calculated
 332 as $\phi_{shift} = (t_{peak} - t_{analytical})/T \cdot 360^\circ$, where t_{peak} is the time where the crest of the
 333 wave passes the sampling position, and $t_{analytical}$ is the time where the stream function
 solution should have passed the sampling position.

Table 2. Phase-shift at $t = 25T$.

Co	0.02	0.05	0.10	0.15	0.25	0.50
$\phi_{shift} [^\circ]$	0.0	0.0	-18	-36	-72	-198

334

335 Figure 6 shows the velocity profiles beneath the crest at $t = 5T$ for the six different
 336 values of Co together with the stream function solution, similar to Figure 4. It can
 337 be seen that as Co is lowered the simulated velocity profiles become closer to the
 338 analytical solution. The reason for this is probably two-fold. First, lowering Co delays
 339 the presence and growth of the interface wiggles and thus also the steepening of the
 340 wave. Second, any inconsistent treatment of the force balance near the free surface
 341 is substantially limited by the small time step as it reduces e.g. the error committed
 342 in linearising the convective term $u_j(\partial\rho u_i/\partial x_j)$. The importance of keeping a low
 343 time step in `interFoam` when doing two-phase simulations has also been highlighted

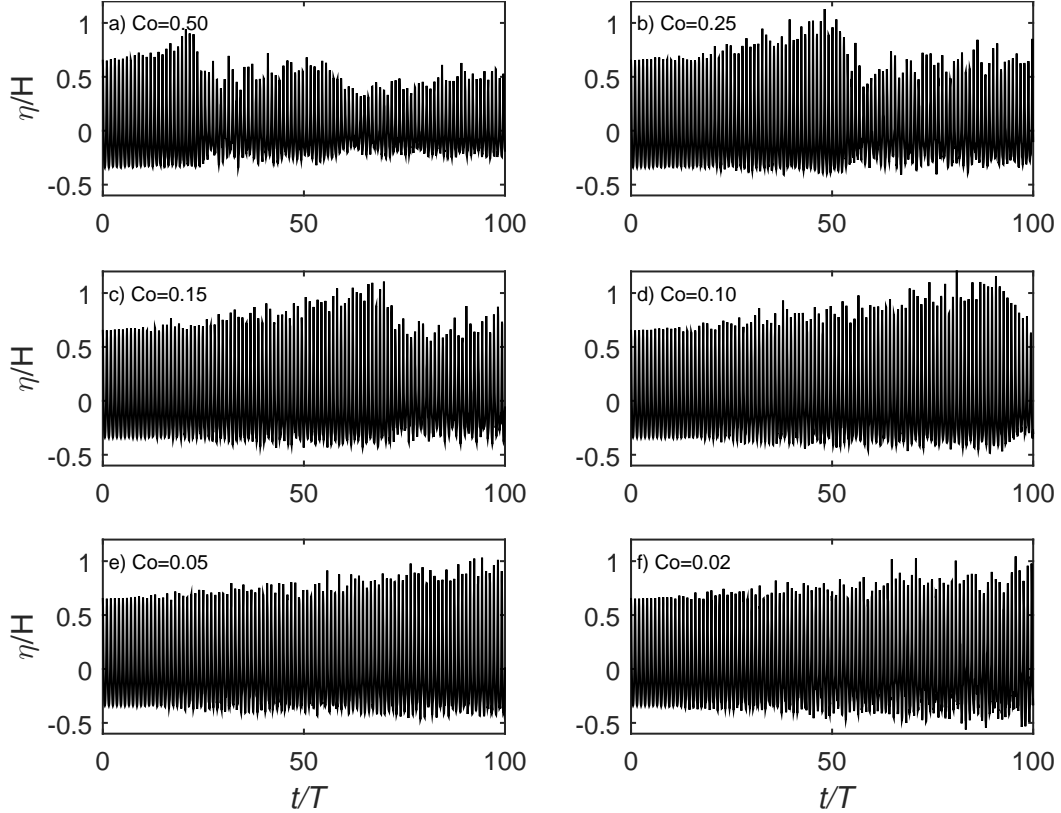


Figure 5. Simulated surface elevation as a function of time for six different Courant numbers (Main fixed parameters: $N = 12.5$, ddt-Euler , grad-Gauss Linear , $\text{div}(\rho*\phi,U)$ - Gauss LimitedLinearV 1, $\text{laplacian-Gauss linear corrected}$, $c_\alpha = 1$).

344 by Deshpande et al. (2012) in the context of surface tension dominated flows, where
 345 it was shown that a small time step is crucial for limiting the growth of spurious
 346 velocities. Even though the present inertia dominated situation is different from the
 347 analysis of Deshpande et al. (2012), the solution to minimize the interface imbalance
 348 by limiting the time step still seems to hold.

349 In addition to the velocity profiles depicted in Figure 6, it is also of interest to see
 350 how the overestimation of the crest velocity evolves in time. Therefore, in Figure 7 the
 351 error in the crest velocity calculated as

$$\Delta E = \frac{\max(U) - U_{\text{analytical}}}{U_{\text{analytical}}} \quad (12)$$

352 is shown for each of the six values of Co considered. Regardless of Co , the overes-
 353 timation of the crest velocity is apparent and grows in time. From Figure 7 it can
 354 be seen that even with a relatively small Co , e.g. $Co = 0.15$, after only propagating
 355 five periods, the crest velocity is approximately 17% larger than the analytical. It thus
 356 seems that, what is generally viewed as a rather "low" Co , is still not sufficiently small
 357 to accurately simulate surface waves. In contrast, the error in the crest velocity for
 358 the case with $Co = 0.05$ is only 0.1% after five periods, thus this value seems like a
 359 proper Co for the accurate simulation of this wave. These results indicate that many
 360 previous simulations of free-surface waves have not achieved time step convergence.

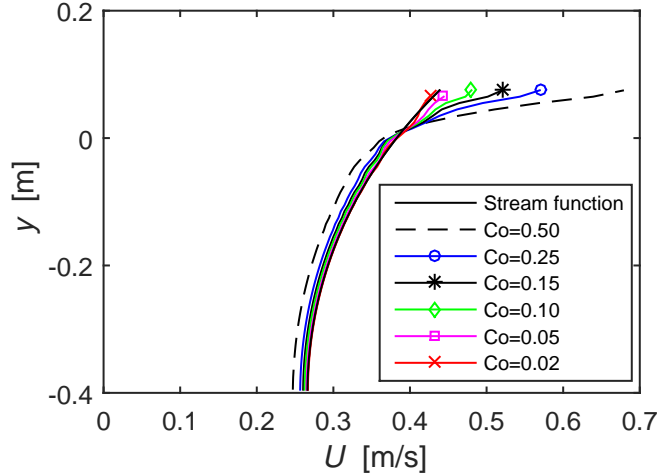


Figure 6. Velocity distribution beneath the crest at $t = 5T$ for various Courant numbers (Main fixed parameters: $N = 12.5$, `ddt-Euler`, `grad-Gauss Linear`, `div(rho*phi,U)-Gauss LimitedLinearV 1`, `laplacian-Gauss linear corrected` $c_\alpha = 1$).

361 4.3. The effect of mesh resolution

362 Having checked the effect of the temporal resolution, it now seems natural to check the
 363 effect of varying the spatial resolution. However, as the solution with $Co = 0.5$ from the
 364 `damBreak` tutorial was poor, the rest of the forthcoming analysis will be continued with
 365 $Co = 0.15$, with the hope of further improving the previous results. In Jacobsen et al.
 366 (2012) it was noted that `interFoam` performed best with cell aspect ratios, defined
 367 as $\Delta x/\Delta y$, of 1, and this ratio will be maintained throughout the analysis. In the
 368 previous cases $N = 12.5$, and now three additional simulations will be performed with
 369 $N = 50$, $N = 25$ and $N = 6.25$ respectively. Figure 8 shows the surface elevations as
 370 a function of time for the four different resolutions. Similar to increasing the temporal
 371 resolution (i.e. lowering Co) it can be seen that increasing the number of cells per
 372 wave height greatly improves the solution when considering the ability to propagate the
 373 wave while maintaining constant form.

374 Before continuing, it is also worth commenting on the shape of the air–water inter-
 375 face in the different resolutions, which is illustrated in Figure 9 for $N = 6.25$ and
 376 $N = 25$. As expected with $N = 6.25$ the interface looks smeared and is not well cap-
 377 tured. With $N = 12.5$ (not shown here for brevity) the interface looks similar to Figure
 378 2a, but the wave gradually steepens in time as previously explained. With $N = 25$
 379 and also $N = 50$ the interface is even sharper and with $N = 25$ the wave heights were
 380 also seen to increase, but somewhat slower. This is probably related to the size of the
 381 wiggles being much smaller with the finer mesh. In these cases the wiggles were not
 382 only present in the top of the crest, but also along the whole wave surface. They also
 383 appeared at an earlier time, as seen in Figure 9b.

384 In Figure 10 the velocity profiles beneath the crest at $t = 5T$ are shown for the four
 385 different spatial resolutions together with the analytical stream function solution. In
 386 general, it can be seen that, improving the spatial resolution improves the solution.
 387 However, for the case with $N = 25$ the crest velocity is as high as in the coarser resolved
 388 cases. This can be explained by the afore mentioned wiggles. At the crest of such a
 389 surface wiggle, the velocity is much higher compared to the rest of the wave. This is
 390 not seen to the same degree with $N = 50$ where the surface wiggles are much smaller.
 391 When propagating the wave longer than the five periods, it was experienced that the

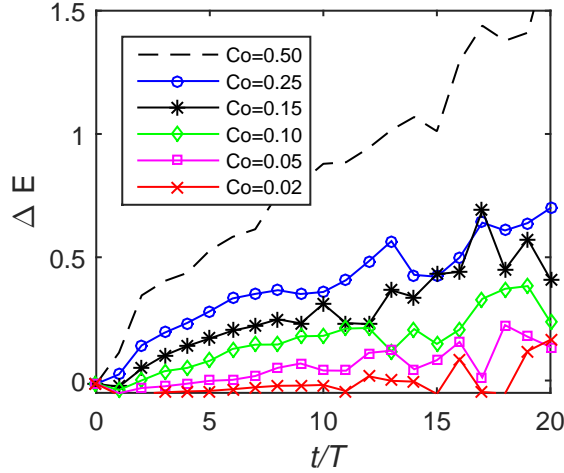


Figure 7. Error in the maximum crest velocity as a function of periods (Main fixed parameters: $N = 12.5$, ddt-Euler, grad-Gauss Linear, div(rho*phi,U)-Gauss LimitedLinearV 1, laplacian-Gauss linear corrected $c_\alpha = 1$).

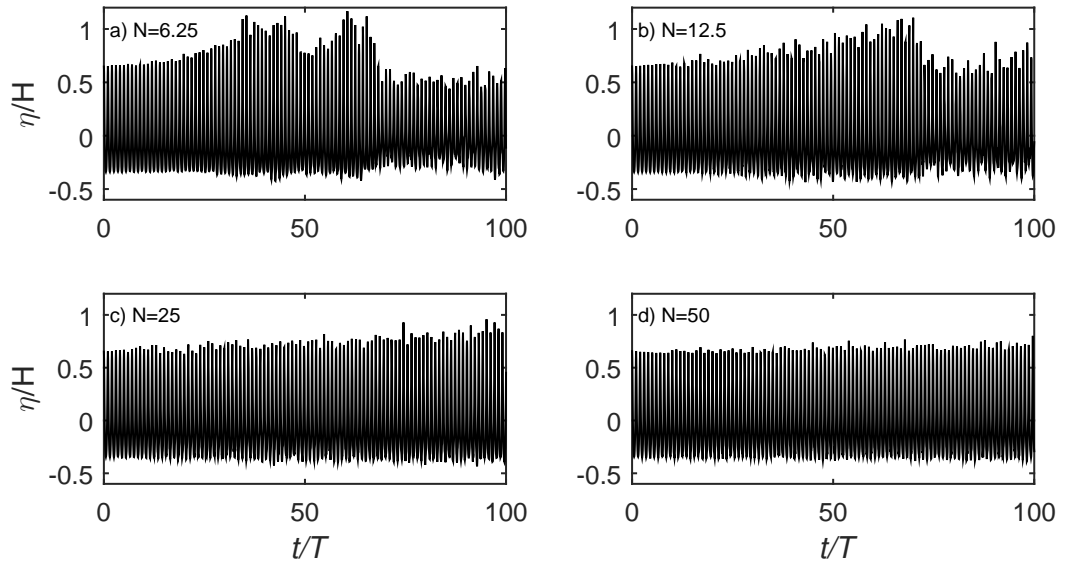


Figure 8. Simulated surface elevation as a function of time for four different mesh resolutions (Main fixed parameters: $Co = 0.15$, ddt-Euler, grad-Gauss Linear, div(rho*phi,U)-Gauss LimitedLinearV 1, laplacian-Gauss linear corrected $c_\alpha = 1$).

392 case with $N = 25$ had crest velocities closer to the analytical solution than the two
 393 coarser resolved cases. From the above results it is worth noting that increasing the
 394 spatial resolution was not able to produce as good results for the velocity profiles as
 395 increasing the temporal resolution, see Figures 6 and 10. From a computational point
 396 of view decreasing Co seem to be a more efficient alternative to increase accuracy, than
 397 increasing the mesh resolution. This is especially true considering that increasing the
 398 mesh resolution, will also make the time step decrease to maintain a given Co . However,
 399 in terms of keeping the wave height constant for the entire simulation, increasing the
 400 spatial resolution does seem to yield better results compared to simply increasing the
 401 temporal resolution.

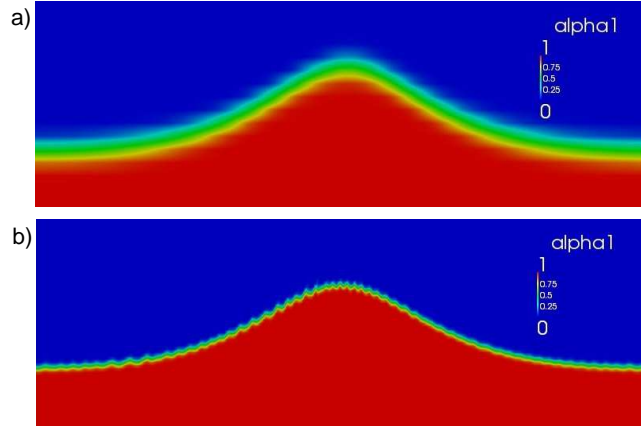


Figure 9. Snapshot at $t = 5.5T$ for a) $N = 6.25$ and b) $N = 25$ (Main fixed parameters: $Co = 0.15$, ddt-Euler, grad-Gauss Linear, div(rho*phi,U)-Gauss LimitedLinearV 1, laplacian-Gauss linear corrected $c_\alpha = 1$).

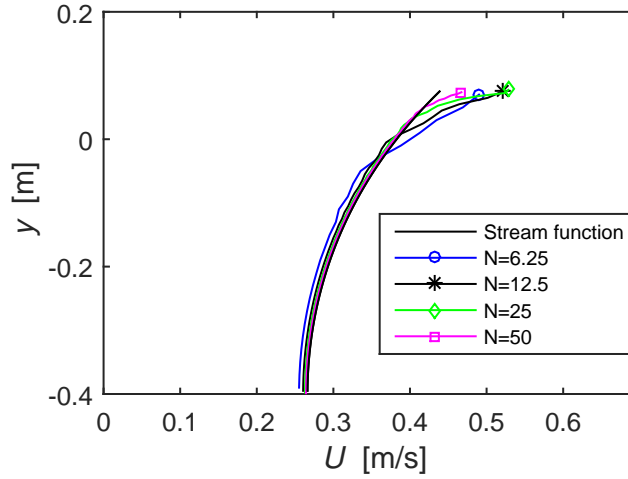


Figure 10. Velocity distribution beneath the crest at $t = 5T$ for various mesh resolutions (Main fixed parameters: $Co = 0.15$, ddt-Euler, grad-Gauss Linear, div(rho*phi,U)-Gauss LimitedLinearV 1, laplacian-Gauss linear corrected $c_\alpha = 1$).

402 4.4. *fvSchemes and fvSolution settings*

403 Thus far increasing the temporal and spatial resolution have been attempted, and
 404 unsurprisingly, these improved the solution. For the rest of this study $Co = 0.15$
 405 and $N = 12.5$ will be maintained for the sake of balancing computational costs and
 406 accuracy, and the additional effects of changing schemes and solution settings will be
 407 investigated. As quite a few schemes are available, not all results of our investigations
 408 will be shown. Our findings will be summarized and figures will be included when
 409 found to be most relevant. Later, we will combine some of the investigated schemes to
 410 improve the overall solution quality.

411 It has been shown that the interface between air and water in time develop wiggles,
 412 which in time grow and sometimes lead to breaking. First, the additional effects of
 413 modifying `cAlpha` (with default value $c_\alpha = 1$), which controls the size of the compression
 414 velocity, will be investigated. It was experienced that increasing c_α causes the
 415 wiggles to appear earlier and grow faster. Reducing c_α reduces the wiggles and at the

416 same time causes the interface to smear out over more cells. This strongly indicates
 417 that the wiggles are caused by the numerical interface compression method.

418 To illustrate the effect of c_α , the surface elevations are shown for four different values
 419 in Figure 11. In this figure, to demonstrate the effect of c_α on the interface, we also
 420 plot the $\alpha = 0.99$ and $\alpha = 0.01$ contours for the crest and the trough for each period.
 The reduction in wave height seen in the case with $c_\alpha=0$ (Figure 11a), is the effect

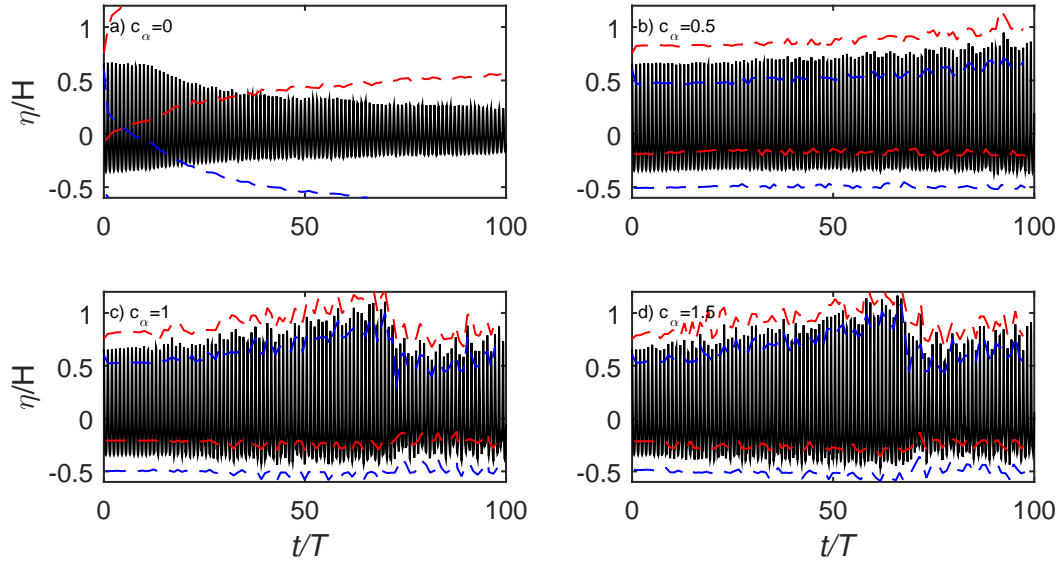


Figure 11. Simulated surface elevations (-) as a function of time for different values of c_α together with the $\alpha = 0.99$ and $\alpha = 0.01$ contours (- -) (Main fixed parameters: $Co = 0.15$, $N = 12.5$, ddt-Euler, grad-Gauss Linear, div(rho*phi,U)-Gauss LimitedLinearV 1, laplacian-Gauss linear corrected).

421 of a very heavy diffusion of the interface. This can be seen even more clearly when
 422 looking at the $\alpha = 0.99$ and $\alpha = 0.01$ contours. It can be seen that after 20 periods
 423 the 0.99 contour at the crest is actually positioned lower than the trough level and
 424 the 0.01 contour at the trough is almost at the crest level. The distance between the
 425 0.01 contour and 0.99 contour is approximately four cells with $c_\alpha = 0.5$ (Figure 11b),
 426 whereas it only spans approximately three cells for $c_\alpha = 1$ (Figure 11c) and $c_\alpha = 1.5$
 427 (Figure 11d). This shows that increasing c_α does compress the interface, but that the
 428 interface will span more than one cell, even with a high value of c_α .
 429

430 In addition to the c_α value, various other settings affect the size and behaviour of
 431 the wiggles, and in the following $c_\alpha = 1$ will be maintained, for the sake of comparison.
 432 The effect of the time discretization scheme on the surface elevations is shown in Figure
 433 12. Changing the time discretization scheme from **Euler** (first order) to **CN** (second
 434 order) exacerbates the wiggle feature, causing them to develop earlier and extend
 435 throughout the surface. Contrary to results utilizing the **Euler** scheme, the wiggles
 436 do not cause the wave to steepen to the same extent. The wiggles grow in size, but
 437 they often break on top of the wave before merging, and therefore the wave does not
 438 steepen as much as with the **Euler** scheme. It is believed that the wiggle feature is
 439 more pronounced with the **CN** scheme simply because the scheme is less diffusive than
 440 the **Euler** scheme. The artificial compression term, as just shown, adds some erratic
 441 behaviour to the interface, and this is diffused by numerical damping when using the
 442 **Euler** scheme, but less so when using **CN**.

443 The reduction or complete removal of wiggle formations is also seen utilizing other

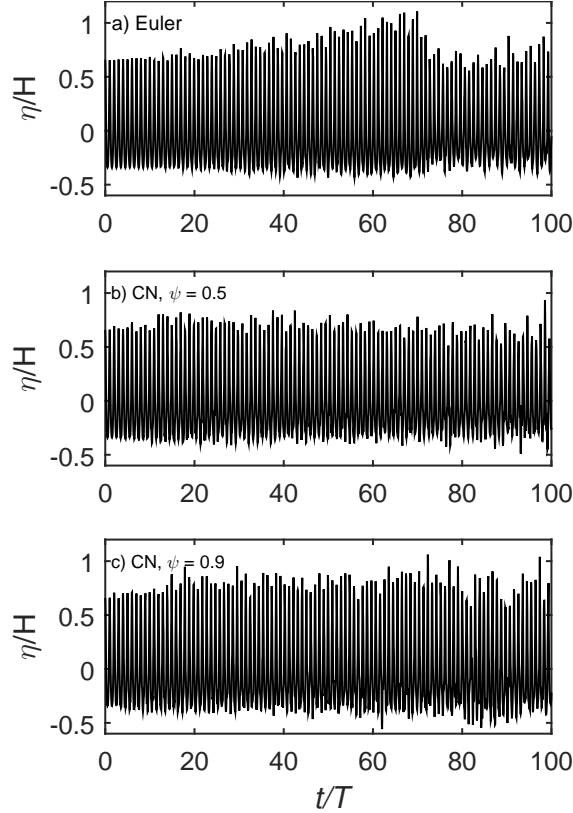


Figure 12. Simulated surface elevation as a function of time for different time discretization schemes (Main fixed parameters: $Co = 0.15$, $N = 12.5$, `grad-Gauss Linear`, `div(rho*phi,U)-Gauss LimitedLinearV 1`, `laplacian-Gauss linear corrected`, $c_\alpha = 1$).

444 more diffusive schemes, e.g. when using the upwind scheme for the convection of the
 445 α field or using the upwind scheme for the convection of momentum. In the case of
 446 utilizing the upwind scheme for the convection of the α field the solution is very similar
 447 to that seen when setting $c_\alpha = 0$ (Figure 11a), with the interface experiencing heavy
 448 diffusion and the resulting wave height decaying rapidly. Utilizing an upwind scheme
 449 for the convection of momentum also causes the wave height to decay, but at a much
 450 slower rate, and is not accompanied by the same degree of interface diffusion. However,
 451 utilizing a pure upwind scheme is generally not recommended due to excessive smearing
 452 of the solution.

453 Thus far it has been shown that c_α and the time discretization scheme have a sig-
 454 nificant impact on the surface elevation and interface. However, regarding the velocity
 455 profile beneath the crest (not shown here for brevity), the impact is very small, except
 456 for the case with $c_\alpha = 0$, which made made the velocities throughout the water column
 457 beneath the crest too low. This is probably due to heavy diffusion of the interface (see
 458 Figure 11a).

459 As mentioned, the wiggles can be limited by choosing more diffusive schemes, but
 460 it still needs to be determined how these schemes affect the general propagation of
 461 the wave and the underlying velocity profile. Figure 13 shows the surface elevation for
 462 four different convection schemes (`div(rho*phi,U)`), and the influence of the choice
 463 on convection scheme is readily apparent. The most diffusive among the four schemes,
 464 the `upwind` scheme, makes the wave decay in a quite stable fashion (Figure 13b).

465 The SFCD scheme (Figure 13c) is slightly more diffusive than the `limitedLinearV 1`
 466 scheme (Figure 13a), and is seen to limit the growth in the wave height. The wave
 467 height still increases as time progresses but the increase is delayed and the simulation
 468 is less erratic. The fourth scheme is the `SuperBee` scheme (Figure 13d). This scheme
 469 is also within the TVD family, but it is much more erratic, and almost immediately
 the wave heights start to increase.

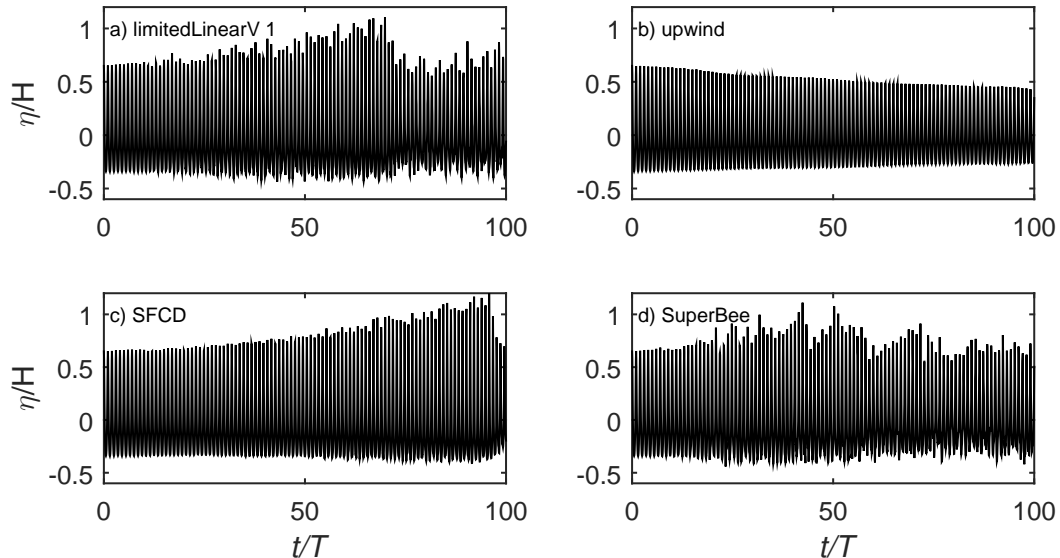


Figure 13. Simulated surface elevation as a function of time for different convection schemes (Main fixed parameters: $Co = 0.15$, $N = 12.5$, `ddt-Euler`, `grad-Gauss Linear`, `laplacian-Gauss linear corrected`, $c_\alpha = 1$).

470

471 The velocity profiles beneath the crest for the four convection schemes are like-
 472 wise shown at $t = 5T$ in Figure 14, and once again the importance of the convection
 473 scheme is quite clear. The `upwind` scheme limits the error in the velocity at the top
 474 crest whereas it underestimates the velocity closer to the bed. The SFCD scheme be-
 475 haves slightly better than the `limitedLinearV 1` scheme, and the `SuperBee` scheme
 476 performs the worst. When propagating further the `SuperBee` scheme has oscillations
 477 in the velocity profile beneath the crest, which can also be seen to a smaller degree in
 478 Figure 14.

479

A range of other convection schemes have also been attempted. None of them,
 480 however, show significantly different results than those shown here, which have been
 481 selected to demonstrate the effect of convection scheme diffusivity on the propagation
 482 of the wave and velocity profile beneath the crest. While the convection schemes have
 483 been shown to have a great effect on both the ability to maintain a constant wave
 484 height, limit the wiggle feature in the interface and predict the velocity profile, it is
 485 not directly evident which scheme performs the best overall. The `upwind` scheme limits
 486 the error in the crest velocity the most, which would be beneficial when e.g. doing loads
 487 on structures, but due to the diffusivity of the scheme might not be able to capture
 488 e.g. vortex shedding around such a structure. The SFCD scheme improves the ability to
 489 maintain a constant wave height and limits the growth in the crest velocity compared
 490 to the `limitedLinearV 1` scheme from the `damBreak` tutorial, but the crest velocity
 491 is still severely overestimated.

492

We will now turn our attention to the gradient (`grad`) schemes. These effects (rel-
 493 ative to the default `Gauss Linear` scheme in Figures 5c and 6) on the wave propa-

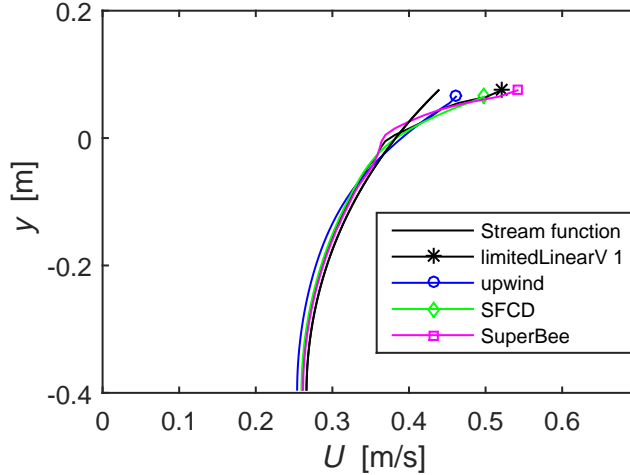


Figure 14. Velocity distribution beneath the crest at $t = 5T$ for various convection schemes (Main fixed parameters: $Co = 0.15$, $N = 12.5$, `ddt-Euler`, `grad-Gauss Linear`, `laplacian-Gauss linear corrected`, $c_\alpha = 1$).

494 gation and velocity profile will be described, but for brevity no additional figures will
 495 be included. The fourth-order scheme (`fourth`) improves the propagation and delays
 496 the increase in wave heights, similar to the behaviour seen with the `SFCD` convective
 497 scheme (Figure 13c), which is more diffusive than the standard `limitedLinearV 1`
 498 scheme. The `fourth` scheme is however not more diffusive than the `Gauss Linear`
 499 scheme, and the delayed increase in wave height is probably due to the scheme
 500 having higher-order accuracy. The velocity profile beneath the crest, on the other
 501 hand, is not improved relative to the `Gauss Linear` scheme (Figure 6, $Co=0.15$). The
 502 `faceMDLimited Gauss Linear 1` gradient scheme has also been tested, and behaves
 503 very similar to the `upwind` convection scheme (Figure 13b), in the sense that the wave
 504 heights decrease with time. The reason for this is probably that the gradient limiter,
 505 coupled with the `limitedLinearV 1` convection scheme, effectively makes the con-
 506 vection scheme an upwind scheme. With respect to the velocities the `faceMDLimited`
 507 gradient scheme produced a velocity profile very similar to that from the `upwind`
 508 scheme (Figure 14). That the limited gradient scheme can produce results similar to
 509 the `upwind` convection scheme was also observed by Liu and Hinrichsen (2014), who
 510 studied the effect of convection and gradient schemes on bubbling fluidized beds using
 511 `OpenFOAM`.

512 We will now describe how changing the Laplacian scheme effects the solution, rela-
 513 tive to the default setting (`Gauss linear corrected`). As previously mentioned the
 514 Laplacian scheme requires keywords for both `interpolation` and `snGrad`, but the in-
 515 puts for the stand alone `interpolation` and `snGrad` schemes are not changed. For the
 516 Laplacian scheme, combining the `Gauss linear` interpolation with the `fourth snGrad`
 517 scheme, resulting in the Laplacian scheme `Gauss Linear fourth`, gave improved re-
 518 sults, both in terms of the ability to maintain constant wave heights and in terms of
 519 the velocity profile beneath the crest. However switching to the fourth-order scheme
 520 (`fourth`), resulted in very high spurious velocities in the air region above the wave,
 521 and hence (due to the Co -controlled time step) leads to reductions in the time steps
 522 used during the simulation. In this way changing to a fourth-order `snGrad` schemes in
 523 the Laplacian is effectively similar to lowering Co . To check whether the fourth-order
 524 `snGrad` scheme in the Laplacian really improved the solution, or if it is merely a result

525 of a reduced time step, two additional simulations, now utilizing a fixed time step
526 $dt=0.0025$ s, have been performed, with both `corrected` and `fourth snGrad` scheme
527 in the Laplacian. The resulting velocity profiles at $t = 5T$, together with the result
528 from a simulation with $\rho_{air} = 0.1$ kg/m³ (also utilizing the same fixed time step), are
shown in Figure 15. The three simulations show similar results in the water phase,

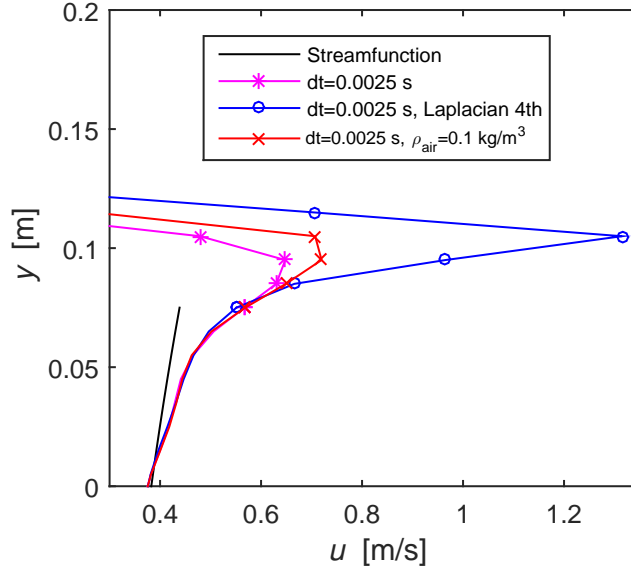


Figure 15. Velocity distribution beneath the crest at $t = 5t$ with a fixed time step utilizing the standard setup as well as 4th order Laplacian and $\rho_{air} = 0.1$ kg/m³. Full lines represent the velocities in pure water and the lines with symbols represent the velocities in the air or mixture cells (Main fixed parameters: $N = 12.5$, `ddt-Euler`, `grad-Gauss Linear`, `div(rho*phi,U)-Gauss LimitedLinearV 1`, $c_\alpha = 1$)

529 but rather different velocities in the air phase. These results indicate that, while being
530 an un-physical and undesirable phenomenon, the spurious velocities in the air do not
531 seem to effect the wave significantly. The case with a fourth-order `snGrad` scheme had
532 approximately twice as high air velocities as the standard set up, but similar (actu-
533 ally slightly lower) crest velocities. The case with lower density also has higher air
534 velocities, but very similar water velocities to the standard case. To summarize: Even
535 though the fourth-order Laplacian scheme is able to produce better wave kinematics,
536 caution must be taken as it produces large spurious velocities. These will, utilizing a
537 variable time step, lead to very low time steps. Alternatively, a fixed time step may
538 result in an unstable Courant number.

540 Before conducting the present study it was expected that the discretization schemes
541 would have an effect on the solution, but it was also expected that in particular the
542 choice of iterative solvers for the pressure would not have an effect, at least if the tol-
543 erances were sufficiently low. It turns out, however, that the iterative solver settings
544 in `fvSolution` also affect the wave propagation. For the pressure equations (`pcorr`,
545 `pd` and `pdFinal`) switching from `PCG` to `GAMG` made the simulations more erratic as
546 the wave broke much earlier (however the simulation time was much lower), whereas
547 switching to a smooth solver (`smoothSolver`) did not affect the quality of the solution,
548 but took much longer time. It was also attempted to lower the tolerance by a factor
549 1000 on both the pressure and the velocity, but hardly any difference in the solution
550 was seen. For the controls of the solution algorithm increasing the number of alpha cor-
551 rectors, `nAlphaCorr`, as well as alpha subcycles, `nAlphaSubCycles`, improved, though
552 not dramatically, the propagation of the wave in terms of it maintaining its' shape,

553 whereas increasing the number of correctors, `nCorrectors` did not change anything.
554 Increasing the number of outer correctors, `nOuterCorrectors` (`nOCorr`), effectively
555 making it into the PIMPLE algorithm, surprisingly made the wave height decrease
556 very rapidly. This behaviour was also seen in Weber (2016) and will be investigated
557 further in the forthcoming section.

558 The choice of iterative solvers could also potentially effect the velocity profile. The
559 `GAMG` solver produced much higher crest velocities (close to that seen with $Co = 0.5$ in
560 Figure 4). The `SmoothSolver`, which was a lot slower, produced an almost identical
561 velocity profile to the `PCG` solver (Figure 6, $Co = 0.15$). Lowering the tolerances by a
562 factor 1000 had almost no effect on the surface elevation, and the effect on the velocity
563 profile was also negligible. Changing the number of α subcycles (`nAlphaSubCycles`),
564 α correctors (`nAlphaCorr`) and number of correctors (`nCorrectors`) did not influence
565 the crest velocity in any significant way, and raising the number of α correctors actually
566 worsened the result closer to the bed.

567 It has now been shown that the discretization schemes and solution procedures have
568 a potentially large impact in the solution, both in terms of the wave height and velocity
569 profile, as well as the wiggles in the interface and the spurious air velocities. Using more
570 diffusive schemes than the base setup from the `damBreak` tutorial has been shown to
571 limit or remove the growth of the wiggles, limit the overestimation of the crest velocity,
572 and also limit the growth of the wave heights. However, the more diffusive schemes
573 were seen to smear the interface, and could potentially be more inaccurate for other
574 situations. The demonstration of the large importance of the discretization schemes
575 on the accuracy of the solution can be considered an important finding in its own as
576 this has not previously been documented but only hinted e.g. by Paulsen et al. (2014);
577 Wroniszewski et al. (2014).

578 4.5. *Combined schemes*

579 It would be ideal to achieve a setup capable of propagating a wave for 100 periods,
580 while keeping a relatively large time step and at the same time maintaining both its
581 shape and the correct velocities. Changing one single scheme has not achieved that. It
582 was however shown that adding some diffusion in some of the schemes could mitigate
583 both the increase in wave height as well as the increased near-crest velocities.

584 To test whether a combination of schemes can improve the solution further, the
585 `upwind` scheme on the convection of momentum, which was actually seen to cause
586 the wave to decay (Figure 13b), will be combined with the slightly less diffusive
587 blended CN scheme (Figure 12c). It is also attempted to increase the artificial compression,
588 by increasing c_α while picking a more diffusive scheme for the gradient, namely
589 `faceMDLimited` which also caused the wave height to decrease. Finally, the outer correctors
590 are increased to two and combined with the blended CN scheme, together with
591 the `SFCD` scheme for the momentum flux.

592 The surface elevations for three such combinations are seen in Figure 16b–d. Here
593 it can be seen that by combining the diffusive `upwind` scheme for the convection of
594 momentum and shifting from the more diffusive `Euler` scheme to a less diffusive CN
595 scheme (Figure 16b) can maintain the wave height for the entire 100 periods. The
596 same can be done by increasing the compression factor c_α while maintaining a more
597 diffusive gradient scheme (Figure 16c, although in this case the wave heights actually
598 decayed a bit), and also by increasing the number of outer correctors together with
599 the CN scheme (Figure 16d). The latter results in slightly more variations in the wave

600 height, but also utilized a much higher blending value in the CN scheme, which can
 601 cause oscillations in the solution and, as previously shown, excite wiggles in the free
 602 surface. All three cases show a great improvement compared to the original default
 case, repeated as Figure 16a to ease comparison. It should also be stated that the

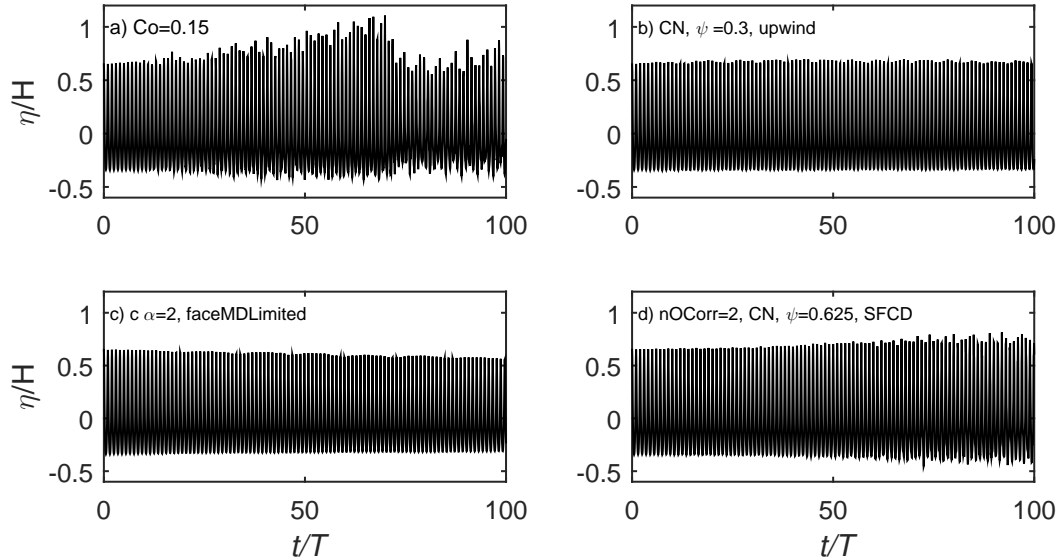


Figure 16. Simulated surface elevation as a function of time for different schemes (Main fixed parameters: $Co = 0.15$, $N = 12.5$, `laplacian-Gauss linear corrected`).

603
 604 balance obtained for the case with the outer correctors is particularly delicate. First
 605 it was attempted to run with two outer correctors and a blended CN scheme, while
 606 maintaining the `limitedLinearV 1` scheme on the momentum flux. This however
 607 caused wiggles in the interface, as also previously described, and therefore the `SFCD`
 608 scheme was chosen to counteract the wiggles. The wiggles were not removed altogether
 609 with the `SFCD` scheme, but their presence was significantly delayed. Further, the best
 610 result was obtained with CN, $\psi = 0.625$, but lowering the blending factor to $\psi = 0.6$
 611 made the wave height decrease slightly over the 100 periods, and raising it to $\psi = 0.65$
 612 made it increase slightly and caused more wiggles.

613 The resulting velocity profiles beneath the crest at $t = 5T$ for the three cases shown
 614 in Figure 16b-d are shown in Figure 17, together with the velocity profile obtained
 615 utilizing the base settings. Here it is evident that all three combinations give lower
 616 velocities in the crest than the standard setting. However the standard setup shows
 617 a slightly better comparison with the analytical result closer to the bottom than the
 618 case utilizing `upwind` for the momentum flux together with CN as well as the case
 619 utilizing $c_\alpha = 2$ together with the `faceMDLimited` gradient scheme. The final combi-
 620 nation, utilizing two outer correctors together with a blended CN scheme and a `SFCD`
 621 scheme shows a significantly better result, and is very similar to the analytical profile.
 622 It can be seen that there are small odd oscillations in the profile of this case, and
 623 these oscillations actually become larger as the wave propagates. Nevertheless, this
 624 significant improvement is achieved with minimal increase in computational expense,
 625 especially compared to the results obtained utilizing the settings from the `damBreak`
 626 tutorial. The improvement in the velocity profile with the outer correctors is inter-
 627 preted as the outer correctors ensuring a better coupling between velocity, pressure
 628 and the free-surface.

629 It has now been shown that it is possible to achieve a "diffusive balance" in the

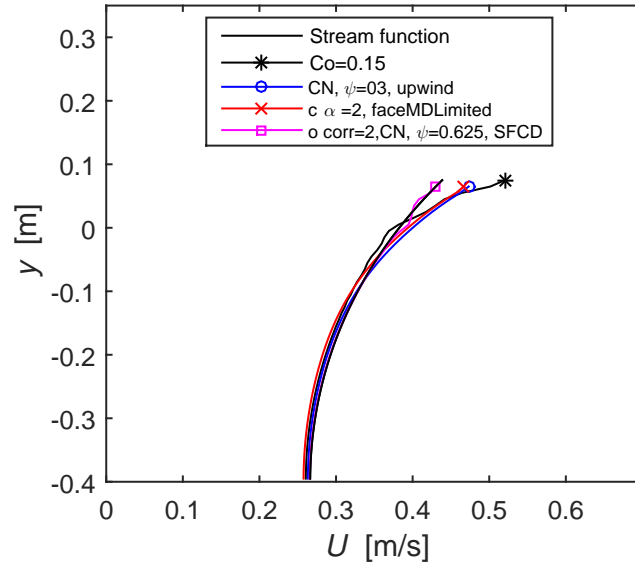


Figure 17. Velocity distribution beneath the crest at $t = 5T$ for various combined schemes (Main fixed parameters: $Co = 0.15$, $N = 12.5$, `laplacian-Gauss linear corrected`).

630 schemes, that enables `interFoam` to progress the wave while maintaining its shape.
631 The same diffusive balance is also shown to limit, but (except for the case utilizing
632 outer correctors) not eliminate, the overestimation of the velocity in the crest. This
633 diffusive balance is, however, not universal. What seems a proper amount of diffusion
634 in the case of $Co = 0.15$ is not so with a lower Co where the error in velocity of the crest
635 is much smaller, and more diffusive schemes would actually worsen the solution. Also,
636 what gives the best balance for this wave, might not give the best balance for a wave
637 with another shape, but the present study reveals a generic strategy that can be fine
638 tuned for individual cases. Interestingly, this implies that for variable depth problems,
639 where waves would not maintain a constant form, there may not be a globally optimal
640 combination. Nevertheless, it is still hoped that better-than-default accuracy can be
641 achieved with the combinations suggested herein.

642 4.6. Summary of experience using `interFoam`

643 To summarize our experience using `interFoam` from this section: The safest way to
644 get a good and stable solution is by using a small Courant number. If the time step
645 is low enough, `interFoam` is capable of producing quite good results. However, due
646 to limited time or computational resources, this solution may often not be realistic in
647 practice.

648 If wishing to use larger time steps, alternatively, it is advised to try to obtain a
649 diffusive balance. The best choice can then be determined on a case by case basis,
650 though it is hoped that the examples utilized above may be a good starting point for
651 more general situations. If looking to simulate e.g. wave breaking, the incoming waves
652 could first be simulated in a cyclic domain, as done herein, prior to doing the actual
653 larger-scale simulation. In this smaller simulation, the proper balance between, diffu-
654 sivity, time step, computational expense and solution accuracy could be determined,
655 before doing more advanced simulations. This should help ensure that reasonable ac-
656 curacy in the initial propagation is maintained, which is important as this will affect

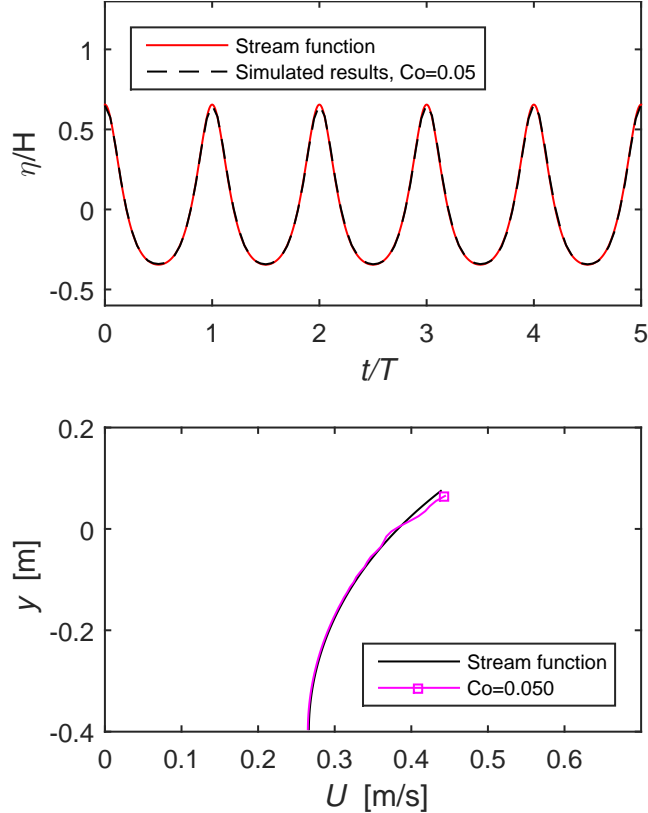


Figure 18. Surface elevations and velocity distribution beneath the crest at $t = 5T$ for $Co = 0.05$ (Main fixed parameters: $N = 12.5$, `ddt-Euler`, `grad-Gauss Linear`, `div(rho*phi,U)-Gauss LimitedLinearV 1`, `laplacian-Gauss linear corrected`, $c_\alpha = 1$).

657 the initial breaking point and hence the subsequent surf zone processes.

658 The present results have focused on a rather demanding task of simulating long-
659 time CFD wave propagation over 100 periods, though the problem with the overes-
660 timation of crest velocities show up much earlier (see again Figure 4). To underline
661 that `interFoam` is capable of producing a good result for most practical applications
662 involving shorter propagation horizons, without having to resort to a diffusive balance
663 strategy, Figure 18 shows the surface elevations for the first five periods, as well as
664 the velocity profile beneath the crest at $t = 5T$ using a small $Co = 0.05$. Here a good
665 match with the analytical stream function solution is achieved. A similar improve-
666 ment in the prediction of the crest velocities, with reduction of Courant number, were
667 shown in Roenby et al. (2017), and this thus seems to be a robust and generally viable
668 strategy.

669 5. `interFoam` coupled with `isoAdvector`: `interFlow`

670 One of the problems with `interFoam` is that the surface gets smeared over several
671 cells, as demonstrated in Section 4.4. This is mitigated by the artificial compression
672 term, which makes the surface sharper, but (as shown herein, Figure 11) also produces
673 some undesired effects. In this section we will finally test the results using `interFoam`
674 coupled with the `isoAdvector` algorithm, recently developed by Roenby et al. (2016),
675 which is also available in the newest version of `OpenFOAM` (`OpenFOAM-v1706`). The

676 `isoAdvect` version in `OpenFOAM-v1706` has a slightly different implementation of
677 the outer correctors than the version used in the present study, see Roenby et al.
678 (2017) for details. With `isoAdvect` the equation for α (6) is not solved directly.
679 Instead the surface is identified by an iso-line, similar to those shown for $\alpha = 0.99$
680 and $\alpha = 0.01$ in Figure 11. After identifying the exact position of the surface, it
681 is then advected in a geometric manner. For more details on the implementation of
682 `isoAdvect` the reader is referred to Roenby et al. (2016).

683 The new `isoAdvect` algorithm, coupled with `interFoam` will for the remainder
684 of this study be named `interFlow`. As a first case, `interFlow` and `interFoam` will be
685 compared for the previously well-tested case with the `damBreak` settings and $Co =$
686 0.15 . It should be stated however, that `interFlow` was not able to propagate the wave
687 with the settings used in `interFoam`. The tolerances on p^* (`pd`) needed to be reduced
688 by a factor 100 and the tolerances on U (`U`) by a factor 10. Comparing the performance
689 of the two is, however, still justified as `interFlow` actually, even with the decreased
690 tolerances, performed the simulation slightly faster than `interFoam`. Moreover, the
691 simulations with `interFoam` did not improve when lowering the tolerances with a
692 factor 1000 as shown in Section 4.4. The speed-up in computational time was not due
693 to larger time steps, but rather to the algorithm moving the free surface faster.

694 Figure 19 shows the surface elevations obtained utilizing the two different solvers. It
695 is quite noticeable that, while with `interFoam` the wave heights start to increase, with
696 `interFlow` the wave heights decrease mildly. Also shown are the contours for $\alpha = 0.99$
697 and $\alpha = 0.01$ for the crest and trough for each period. Here it can be seen that the
698 two contours are substantially closer with `interFlow`. They are constantly separated
699 by less than two cell heights meaning that there is actually only one interface cell in
700 the vertical direction. This is a substantial improvement of the surface representation
701 compared to `interFoam`. Since equation (7) is not solved, there is no artificial com-
702 pression term, and the interface wiggles previously observed are gone altogether. This
703 is likewise a desirable improvement. The artificial compression term has been shown
704 to have undesired effects, as it cause wiggles in the interface, in the simple propagation
705 of a stream-function wave over sufficiently long propagation times. How these wiggles
706 might behave in more complex situation like e.g. wave breaking is an open question,
but one can imagine a greater effect in such a more chaotic situation.

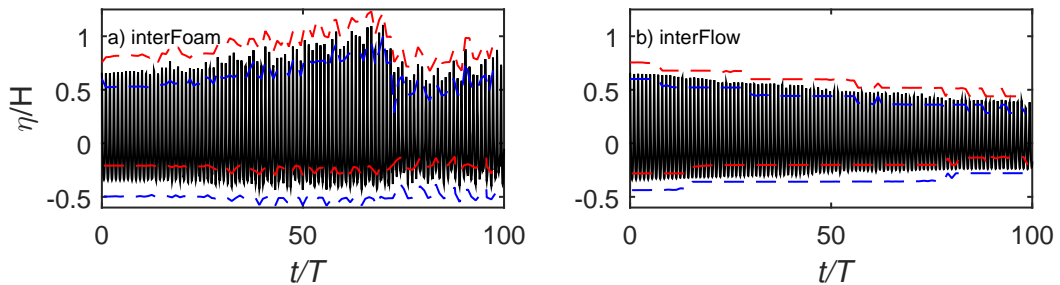


Figure 19. Simulated surface elevations (-) as a function of time utilizing `interFoam` and `interFlow` together with the $\alpha = 0.99$ and $\alpha = 0.01$ (- -) contours (Main fixed parameters: $Co = 0.15$, $N = 12.5$, `ddt-Euler`, `grad-Gauss Linear`, `div(rho*phi,U)-Gauss LimitedLinearV 1`, `laplacian-Gauss linear corrected`).

707
708 In Figure 20 the velocity profile beneath the crest at $t = 5T$ is shown utilizing
709 both `interFoam` and `interFlow`. Here it is quite clear that `interFlow`, with the cur-
710 rent settings is not improving the velocity profile. The crest velocity is slightly larger
711 than the `interFoam` solution, and closer to the bed, the velocity is underestimated.

This underestimation of velocity is probably due to the decrease in wave height. That

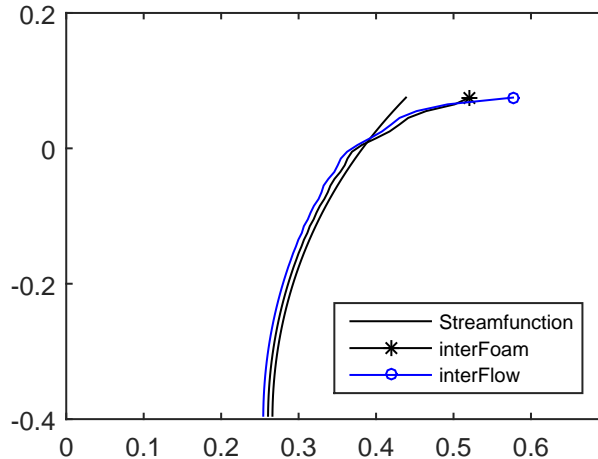


Figure 20. Velocity distribution beneath the crest at $t = 5T$ utilizing `interFoam` and `interFlow` (Main fixed parameters: $Co = 0.15$, $N = 12.5$, `ddt-Euler`, `grad-Gauss Linear`, `div(rho*phi,U)-Gauss LimitedLinearV 1`, `laplacian-Gauss linear corrected`).

712

713 `interFlow` gets an even larger error in the velocity in the top of the crest is proba-
 714 bly due to the sharper interface, creating larger gradients, and any imbalance in the
 715 momentum equation near the interface may then be increased.

716

As shown with `interFoam`, `interFlow` is also sensitive to the setup, and the same
 717 diffusive balance that could be achieved with `interFoam` can also be achieved with
 718 `interFlow`. In Figure 21 the simulated surface elevations utilizing `interFoam` and
 719 `interFlow` respectively are once again compared, this time utilizing schemes to achieve
 720 a diffusive balance. It can be seen that `interFlow`, like `interFoam`, is capable of
 721 propagating the stream function wave for 100 periods, and that `interFlow` throughout
 722 the simulation keeps a sharper interface as the $\alpha = 0.01$ and $\alpha = 0.99$ contours are
 723 much closer. It can also be seen that `interFlow` does not have the same erratic surface
 724 elevation when utilizing two outer correctors together with a blended CN scheme,
 725 which can be explained by `interFlow` not having an artificial compression term, and
 726 therefore the CN scheme does not excite any erratic behaviour near the free surface.
 727 However like `interFoam`, `interFlow` is also very sensitive to the exact value of the
 728 blended CN scheme, and lowering the blending factor, i.e. going more towards the
 729 `Euler` scheme made the wave heights decay, and raising it towards more pure CN
 730 made the wave heights increase.

731

The resulting velocity profiles are shown in Figure 22. Here it can be seen that
 732 the two solvers perform quite similarly when utilizing an `upwind` scheme together
 733 with a blended CN scheme, and that the overestimation of the velocity near the crest
 734 is reduced. Furthermore, it can be seen that `interFlow` also shows a significantly
 735 improved velocity profile when switching to two outer correctors, together with a
 736 blended CN scheme and that `interFlow` does not suffer, to the same degree, from
 737 oscillations in the velocity profile as did `interFoam`.

738

To further underline the impressive performance of `interFlow` when utilizing a
 739 balanced setup, Figure 23 shows the surface elevation from the 95th to the 100th
 740 period together with the velocity profile beneath the crest at $t = 100T$. Here it can be
 741 seen that even after propagating the nonlinear wave for 100 periods `interFlow` still
 742 follows the analytical stream function solution. The surface elevations are of the right
 743 magnitude, and there are no significant phase differences. Furthermore, it can be seen

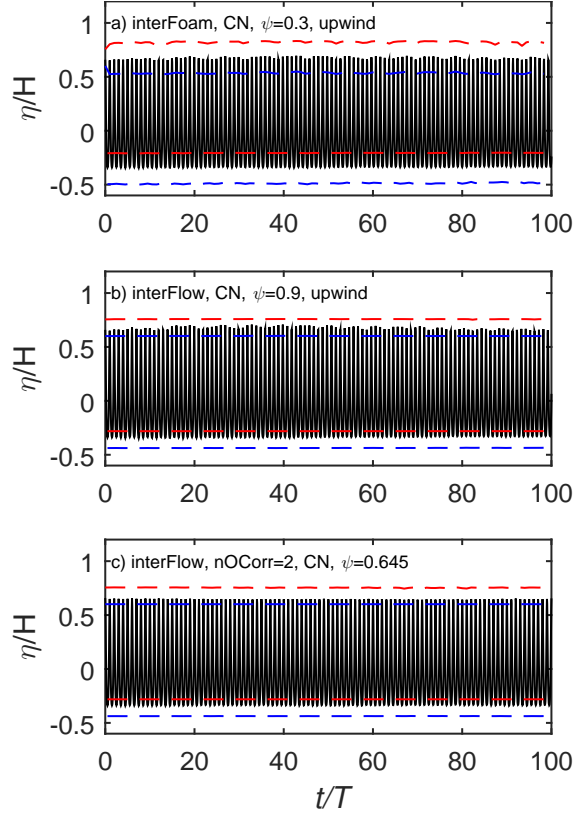


Figure 21. Simulated surface elevations (-) as a function of time utilizing `interFoam` and `interFlow` together with the $\alpha = 0.99$ and $\alpha = 0.01$ (- -) contours (Main fixed parameters: $Co = 0.15$, $N = 12.5$, `grad-Gauss Linear`, `laplacian-Gauss linear corrected`).

744 that the velocity profile is likewise quite close to the analytical result, though it suffers
 745 from minor oscillations.

746 6. Conclusions

747 In this study the performance of `interFoam` (a widely used solver in `OpenFOAM`) in
 748 the simulation of progressive regular gravity waves (having intermediate depth and
 749 moderate nonlinearity) has been systematically documented. It has been shown that
 750 utilizing the basic settings of the popular `interFoam` tutorial `damBreak` will yield quite
 751 poor results, resulting in increasing wave heights, a wiggled interface, spurious air
 752 velocities, and severely overestimated velocities near the crest. These four problems
 753 can be reduced substantially by lowering the time step and increasing the spatial
 754 resolution. It has been shown that a rather small time step, corresponding to a Courant
 755 number $Co \approx 0.05$ is needed to give a good solution when propagating a wave even
 756 short distances of around five wave wave lengths.

757 To test whether an improved solution could be achieved without (drastically) low-
 758 ering the time step and increasing the spatial resolution, a set of simulation have been
 759 performed, where the discretization schemes and iterative solution procedures were
 760 changed one at a time. By gradually increasing and lowering the artificial compression
 761 term (c_α), it was identified as root of the interface wiggles, which was exacerbated
 762 when increasing the c_α and damped or completely removed when lowering c_α . It was

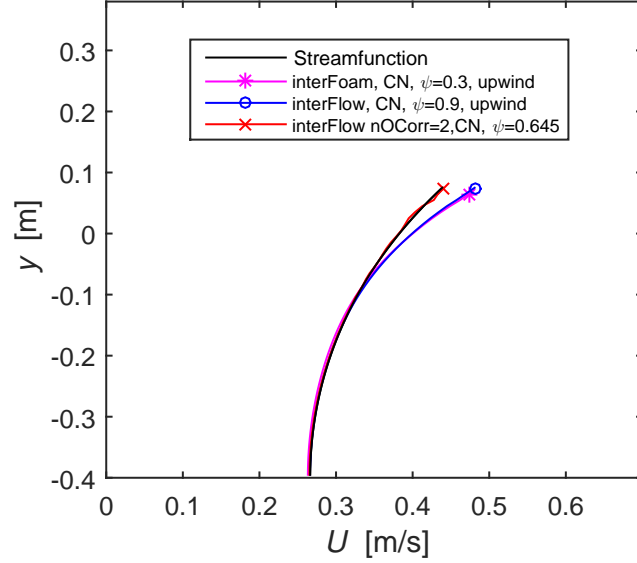


Figure 22. Velocity distribution beneath the crest at $t = 5T$ utilizing an upwind scheme for the convection for `interFoam` with $CN(\psi=0.3)$ as well as `interFlow` with $CN(\psi=0.9)$ and two outer correctors with $CN(\psi=0.645)$ (Main fixed parameters: $Co = 0.15$, $N = 12.5$, `grad-Gauss Linear`, `laplacian-Gauss linear corrected`).

763 also shown how changing from first-order backward Euler time discretization scheme
764 to the (almost) second order, and less diffusive, blended Crank-Nicolson scheme caused
765 the wiggles to appear earlier and cover a larger part of the interface. The convection
766 schemes was shown to affect not only the interface wiggles, but also the development of
767 the wave heights as well as the velocities beneath the crest. More diffusive convection
768 schemes removed the interface wiggles and delayed the increase in wave heights or in
769 fact, when using an **upwind** scheme, caused the wave heights to decrease. Furthermore,
770 the more diffusive schemes also reduced the overestimation of the crest velocities. In
771 general the effect of the gradient schemes was not as large as the convection schemes,
772 but the **fourth** scheme improved the solution, and the `faceMDLimited` scheme behaved
773 very similar to the **upwind** convection scheme. Finally changing the `snGrad` scheme
774 in the Laplacian created large spurious velocities in the air phase directly above the
775 wave. These high velocities however did not seem to influence the wave kinematics.
776 This was further backed by simulations done with a fixed time step, which clearly
777 indicated that the spurious air velocities, while being an unwanted and un-physical
778 phenomenon, do not have a large impact on the wave kinematics. By combining more
779 or less diffusive schemes it was shown that a "diffusive balance" could be reached,
780 where it was possible to propagate the wave a full 100 wave lengths while maintaining
781 its shape. One of these balanced settings also showed a significant improvement in the
782 velocity profile beneath the crest.

783 The new open source solver `interFlow` was subsequently applied, and it was shown
784 that `interFlow` was capable of propagating the wave for 100 periods. The wave de-
785 creased slightly in time, but the interface was a lot sharper, and the wiggles in surface
786 disappeared. Regarding the velocity profile `interFlow` performed slightly worse than
787 `interFoam` with the base settings. Finally it was shown that `interFlow` could achieve
788 the same kind of diffusive balance which enabled the solver to propagate the wave for
789 100 periods while maintaining it shape and also maintaining a good match with the
790 analytical velocity profile.

791 Given its rapidly growing popularity among scientists and engineers, it is hoped that

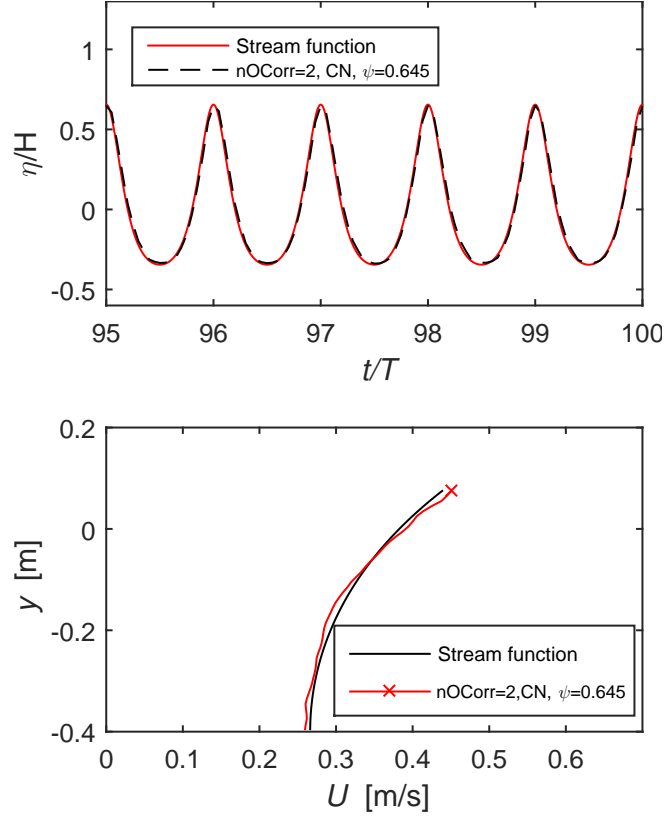


Figure 23. Surface elevations and velocity distribution beneath the crest after 100 periods utilizing interFlow.

792 the present systematic study will raise awareness and enable users to more properly
 793 simulate a wide variety of problems involving the general propagation of surface waves
 794 within the open-source CFD package `OpenFOAM`. While the present study has focused
 795 on the canonical situation involving progressive non-breaking waves, the experience
 796 presented herein is expected to be widely relevant to other, more general, problems
 797 e.g. involving wave-structure interactions, propagation to breaking and resulting surf
 798 zone dynamics, as well as boundary layer and sediment transport processes that result
 799 beneath surface waves, all of which fundamentally rely on an accurate description of
 800 surface waves and their underlying velocity kinematics.

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